



MSc Thesis

Parametric Structural Design

Integration of structural verifications in the early design phases

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Preface

This project is the final master thesis to acquire the MSc. in Architectural Engineering (120 ECTS) part of the Civil Engineering Department at the Technical University of Denmark.

The master thesis is credited with 30 ECTS points and has been written during the Spring Semester 2016, started on the 25th of January and handed in on the 25th of June.

It is focused on the integration of structural considerations in the first phases of the designing process, developed in the frame of the parametrical modeling. This research is carried out in collaboration with Withe Architects AB as part of the STED program (Nordic Built).

25th June 2016 Laura Fernandez Vila № ID: s142599

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Furthermore, I would like to thank Elise Grosse from White Architects Stockholm for her Skype meetings through this process which have helped me to understand the real value of the project and the necessary demands for the architectural projects. Additionally, I would also thank to make possible the meeting with Jonas Jonas Runberger, Director of the Dsearch digital design development department of White Architects Stockholm, who provided me the necessary feedback to know the use of this work for future projects. Within the Nordic Built program, I would like to also thank to Mikkel Kirkeskov Knudsen for provided me with all the necessary information about The Eko-Canopy project.

Lastly, I would like to thank to my master colleagues for his advices regarding the technical aspects of this thesis, and finally thanks to my friends and family for their great support during this long process.

4

Abstract

This thesis project is based on the study of an integrated dynamic method to be applied in the early design stages of the designing building process, which deals with the idea of introducing structural verifications in the models since the beginning. As part of the STED (Sustainable Transformation & Environmental Design) program inside of the Nordic Built, Nordic Innovation program, which principal goal is to achieve sustainable solutions in the Nordic region, the objective of the proposed method is to find reliable structures with the maximum utilization ratio of its members and using the minimal amount of material, which means less kilograms of mass and consequently less emissions of CO2, benefiting the LCA process.

This is done by means of parametrical modeling using the parametric tool for Rinho 3D, Grasshopper, which includes a Finite Element program, Karamba used to perform structural calculations of the created geometries. The combination of Grasshopper and Karamba allows to the user to couple designing tools with calculations tools, which means the mixture of creativity and technical knowledge. Optimization techniques have been integrated, which are based on form-findings methods by means of evolutionary computation. This consists in topological and sizing optimizations procedures of the structures which provide the landscape of optimal structures.

Table of Contents

Preface	2	3		
Acknowledgements				
Abstract				
Table of Contents				
Part I: INTRODUCTION				
1.1	Scope of the project (Motivation)	9		
1.2	Problem Statement	10		
1.3	Structure of the report	11		
Part II: CASE STUDY				
2.1	Background: Nordic Build Program & STED	12		
2.2	Presentation of EKO-Canopy project	13		
Part III: BACKGROUND				
3.1	Literature review	16		
3.2	The Building Design Process and its tools	19		
3.2	Potential of visual programming language tools in the building Industry	20		

3.2.2		2 Softwares used for this master thesis	22
Part	IV: N	METHOD	24
4.	1	Objective	24
4.	2	Applied integrated process description	25
4.	3	Optimization process description	28
4.	4	Normative	31
	4.4.1	L Wind loads calculations	31
	4.4.2	2 Snow loads calculations	33
	4.4.3	Steel Calculations	34
	Bean	ns Verifications	34
	Colu	mns Verifications	38
	Defle	ection verifications	41
Part	V: R	ESULTS - Structural Analysis	42
5.	1	Loads analysis	43
	5.1.1	L Live loads analysis in the Nordic countries: Wind and Snow	44
	5.1.2	2 Load combinations	48
5.	2	Structural analysis of the different shape configurations	49
	5.2.1	Case A: Basic Frame Configuration	50
	5.2.2	Case B: White Architects configuration – From Basic to Complex Structure	58
	Case	B – 1: Basic frame construction	59
	Case	B – 2: Plate & Frame structure	62
	Case	B - 3: White Architects proposal	66
Part	VI: [DISCUSSION	68
6.	1	Integrated method proposal: advantages and disadvantages	68
6.		Results and optimization process: benefits, limitations and its future vements	70
	•	CONCLUSION	
6.		Further studies	
		RAPHY	
		FIGURES	
LIST	OF C	GRAPHS	79
LIST	OF T	TABLES	81

APPENDIX	82
APPENDIX A: THE EKO-CANOPY PROJECT	83
APPENDIX B: STED PROGRAM TASKS	91
APPENDIX C: LOAD CALCULATIONS	94
APPENDIX D: RESULTS OPTIMIZATION PROCESSES	100
APPENDIX E: OUTPUT FROM GRASSHOPPER. SECTION FORCES AND VE	ERIFICATIONS 119
APPENDIX F: SCRIPT IN GRASSHOPPER DEFINITION	124

Part I: INTRODUCTION

1.1 Scope of the project (Motivation)

Nowadays the building industry is continually changing and new technologies are being introduced every day. The working methods should be adapted to the new times to be able to deal with the challenges presented.

The roles involved in the construction process are being rethinking in the process of looking for new methods and workflows which improve the productivity of the projects and their quality (W Young Jr. et al. 2009). In this scenario appears the concept of "Building Information Modeling – BIM", which main principle deals with the idea of establish a better collaboration and coordination between all the agents involved in the building process; based on the concept of having a central model where the data is exchange through different platforms/tools. Being in these ones include all the design tools, FEM software, Mathcad and spreadsheet data tools, etc.

Furthermore, in the last decade one of the main goals of the building processes is to establish methods for sustainable constructions, in this case in the frame of the Nordic countries.

In this context the aim of this master thesis project is to study a methodology to integrate the structural aspects within the architectural designs in the first stages of the designing process, introducing the new tools of the building industry. This concept will lead to achieve more reliable structures from the beginning, which means a reduction of cost through the entire building process, as the design and construability are the goals from the starting point.

9

Furthermore, this is also connected to the concept of sustainability. Taking into account structural considerations during the design phase helps to accomplish the maximum efficiency of the members of the structure, as the same time than an aesthetic design is reached. This way of working, at the end, means a reduction of the material consumption, benefiting the Life-cycle assessment (LCA) process.

Consequently this philosophy is based on the total collaboration between the architect and the engineer from the beginning, or in another way in the idea of creating a third figure in the process, called as the *hybrid practitioner, which has equal insight in technical analyses as well as architectural design* (Negendahl 2015a) and operates as connection between them.

However, this concept is difficult to introduce in the building industry, as traditionally the Architects are being the ones in charge of the design, which concerns all the aspects related to the space distribution, materials, textures and light; while the engineers are the ones which analyze the structural systems to ensure the integrity and safety of the proposal design. In other words, architects are concerned with spaces, while engineers are concerned with forces. (Byrne et al. 2011)

Parallel to this idea, as it has been mentioned before, new digital tools have been developed during the last decade, which makes more obvious the necessity of studying this approach. In this case, the method is based on parametrical modeling tools, as common link, between the two practitioners.

Accordingly to these two main principles, this project tries to demonstrate the importance of this third element in the building design phases through the proposal of an integrate process; which is based on parametrical modeling, combining, both design and structural calculations.

1.2 Problem Statement

Based on the premises introduced before, it is possible to define the research questions which have been formulated for the developing of this master thesis.

Firstly, in the scope of Building Information Modeling it is seen the importance of creating a good workflow between the different agents who participate in the building process, which at the same time means to set a logic data exchange through the different tools to achieve a good result.

Added to this point, it is clear the objective of seeking for building designs which are the most sustainable as possible. In this case, this is going to be based on the idea of achieve structures using the minimal amount of material, in the context of the LCA. Thanks to the parametrical modeling, as is going to be explained further on, optimization process will be carried out in this search for minimized the mass of the structures.

As last point is going to be study the scope and application of the method proposed as well as its feasibility.

These three statements lead to formulate the following questions:

- 1) What should be the logic workflow to integrate the different design and FEM tool in the early design stages?
- 2) It is possible to establish a specific method which can be applied to several projects and which decisions should be made in each step?
- 3) Who should be the agent in charge to lead this process: the architect, the engineer or the hybrid combination of both?
- 4) How feasible and beneficial is the idea of introducing structural aspects in the first stage of the designing process? And related to this, how the optimization processes can benefit to reduce the overall amount of mass based on designing parameters?

To be able to solve this questions, the processes haven been applied to a specific case study, which is going to be explained in Part II.

1.3 Structure of the report

The report is divided in seven different parts. Part I and Part II are focused on the presentation of the project, defining first its objectives and secondly introducing the specific case to study. Then in Part III is analyzed the background which has motivated to develop this work, explaining the building process and its tools.

The fourth part and the most important of this thesis, presents the integrated method proposed, as well as the procedure followed for the developing of this work. This is applied using a specific case study, the EKO-Canopy project, in Part V, where the structural analyses, as well as optimization methods are carried out.

Finally, the Part VI and Part VII discuss the findings of this master thesis and present the corresponding conclusions.

Part II: CASE STUDY

2.1 Background: Nordic Build Program & STED

This master thesis is based in a specific case study as it has been already introduced. The project is the EKO-Canopy proposal by White Architects for the Mullion Program in Upplands Väsby, Sweden. This collaboration is the result of the partnership between the Technical University of Denmark and White Arkitekter through the STED program.

STED stands for Sustainable Transformation & Environmental Design, which is the shorter version of Innovation in Design Methods for Sustainable Transformation of the Existing Nordic Building Stock: Energy, Environmental Design and Life cycle Thinking¹(http://stednetwork.org/, n.d.).

At the same time, STED is part of the Nordic Built, Nordic Innovation program, which was first initiated by Peter Andreas Sattrup in 2014.(Nordicbuilt 2012)

The Nordic Built program is a Nordic initiative to accelerate the development of sustainable building concepts. It was first initiated by the Nordic Ministers for Trade and Industry and consists of an integrated program divided in three different modules: The Charter, The Challenge and the Change.

¹¹ http://stednetwork.org/index.php?p=/article/2015/what-is-sted

STED is part of the last one, as its main objective is to establish design methods and solutions for sustainable transformation in the frame of Life-cycle Assessment, which can be generics and used for projects in the Nordic region.

The project involves four universities in the Nordic Region: Chalmers University of Technology, (Sweden), Norwegian University of Science and Technology, (Norway), The Royal Danish Academy of Fine Arts Schools of Architecture, Design and Conservation (Denmark) and the Technical University of Denmark, (Denmark); as well as the five architectural studios: White Arkitekter, (Sweden), Vandkunsten Arkitekter, (Denmark), Helen & Hard, (Norway), Studio Granda, (Iceland) and OOPEAA, (Finland). (http://stednetwork.org/, n.d.)

As a result of this collaboration this master thesis work was presented and discussed during the Nordic Build Sustainable Transformation & Environmental Design Network conference at NTNU Trondheim on the 9th of June 2016. [APPENDIX B: STED PROGRAM TASKS]

In this case, as it has been said, the project is a proposal of White Arkitekter, (Sweden), and it is going be presented in the following section.

2.2 Presentation of EKO-Canopy project

The project of study consists in an EKO-Canopy construction between two exiting apartment blocks belonging to the Swedish Million Program. (Miljonprogrammet 1965) This was a housing program developed in Sweden between 1965 and 1974 with the aim of building one million new houses to provide to the population with a place to live at a reduced cost.

Over the years, the social environment of these areas has been retrogressed. Therefore, with the aim of achieve their social transformation as the same time that do a sustainable transformation of the block apartments, the White proposal was presented for the Swedish Competition "Innovative utilization of excess heat in future cities" in September 2015.



Figure 2.2-1: Location of the project Dragonvägen, municipality of Upplands Väsby in Sweden

The area to study is located in Dragonvägen in the municipality of Upplands Väsby in Sweden. This district is located at twenty-five kilometers in the north respect the center of Stockholm, as it can be seen in Figure 2.2-1. It consists in a group of eight buildings of similar characteristics, arranged four by four (Figure 2.2-2, up). They are eight stories high with a total height of 22.63m. Each apartment block is 12.10m width by 72.56m length (Figure 2.2-2, down); and they are separated a distance of 28.25m approximately.



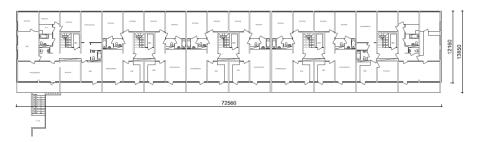


Figure 2.2-2: Space to be studied (up) and floor plan of the one block apartment (down)

The main structure consists of prefabricated concrete elements, and the materials used for the walls and the claddings are poor quality and at the present time are in bad conditions. The area around the apartments is now a bit deserted and desolated (Figure 2.2-3). All the technical documentation, floor plan drawings, materials information and measurements are included in the APPENDIX A: THE EKO-CANOPY PROJECT of this report.





Figure 2.2-3: Area of the surroundings (left) and space between two apartment blocks to be renovated (right)





Figure 2.2-4: Front façade (left) and back façade (right)

The proposal of White Architects consists on a Canopy construction between the two apartment blocks (Figure 2.2-5, right). The main concept is to create a nice and good environment, where social activities can be done, combined with playground spaces and green areas. The main structure is made with wood elements covered by a glass surface. The construction includes a serial of sustainable measures to fulfill with the goal of the competition.

The Figure 2.2-5 on the left shows the indoor environment created between two of the apartment blocks, where it can be seen the idea of the social transformation of the area.



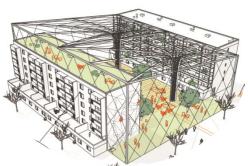


Figure 2.2-5: Render inside Canopy (left) and scheme of overall concept of the Canopy(right) (White 2015)

Consequently, it is seen that a serial of design considerations should be taken into account. The aesthetic aspects should be in equilibrium with the energy and structural premises. Therefore the following parameters were studied: surface area to cover, solar potential, snow handling, ventilation of the Canopy to avoid the green effect, wind considerations, cost, and LCA.

One of the questionable aspects from the jury after the competition was the "The major construction and the dimensioning of the structure" (Knudsen 2016) included in the (Appendix XX). Therefore this master thesis is going to focuses its investigations on the structural considerations as a particular case study for the proposed design-analytical integrated method. All the aspects related to energy and indoor comfort have been investigated by Mikkel Kirkeskov Knudsen. (Knudsen 2016)

Part III: BACKGROUND

3.1 Literature review

In the context of the proposed research, two wide areas are the main focus of study, which at the same time are correlated.

On the one hand this work is based on the idea of looking for a logic method to be applied in the early design stages of the constructions projects, integrating both architectural and structural considerations. Within this area, firstly is important to introduce the traditional building process, and its evolution until how is interpreted nowadays. Related to this, the concept of BIM plays a significant role, as it constitutes the new way of understanding the building industry, where the architecture, the engineering and the construction coexists. This is the result of the seeking of reinvention to adapt the technological advances to the new decade. (W Young Jr. et al. 2009).

Although the term BIM is not the focus of this research, it is important to define its meaning. In this way, The National Building Information Model Standard Project Committee defines BIM as "a digital representation of physical and functional characteristics of a facility. A building information model is a shared knowledge resource for information about a facility forming a reliable basis for decisions during its life-cycle; defined as existing from earliest conception to demolition"²

² Building Smart. International home for Open BIM. http://buildingsmart.org/standards/technical-vision/

Moreover, The National Institute of Building Sciences defines BIM as "a fundamentally different way of creating, using, and sharing building lifecycle data" (NIMBS Committe 2007). It divides the idea of BIM in three different meanings, firstly it is understood as a product, (BIM technology — modelling) secondly as a collaborative process and thirdly as a facility lifecycle management requirement (BIM Methodology — Management Model) (NIMBS Committe 2007) (W Young Jr. et al. 2009)

Therefore, it can be concluded that BIM involves all the processes of collaboration and exchange of data between all the agents who participate in the building process, based on the use of several software, which support a common data format (IFC format) and in last instance but being the key issue, that everything is connected by a central shared model.

This concept of collaboration of the building agents and data exchange between different platforms will be the basis of the method for this master thesis work, but express from a different approach that the BIM concept. In this frame is introduced the idea of integrated dynamic models. The IEA, International Energy Agency³, introduce this concept on its guide "Integrated Design Process, a guideline for Sustainable and Solar-Optimised Building Design" (G. Löhnert, A. Dalkowski, and W.Sutter 2003), where is clearly stated the necessity of re-think the structure the design process with the aim of combine "more knowledge and creativity in the early stages of the process" (Task 23 IEA 1997). This mentality leads to study different design options but analyzing its potential performance as the technical knowledge is included from the beginning. When the engineer point of view is introduced in the process in last stages, it is not possible that the technical knowledge has an influence on the design choice.

In this way, Integrated design process (IDP) is "based on the idea of an optimized teamwork, a qualified design process management including application of modern tools and strategies which fit to the project goals" (G. Löhnert, A. Dalkowski, and W.Sutter 2003).

This philosophy is further discussed by Negendahl in its article "Building performance simulation in the early design stage: An introduction to integrated dynamic models" (Negendahl 2015a), where is study the idea of combining analysis tools with design tools. In this way, it is possible to consider multiple alternatives while a technical feedback is provided. Therefore, in Figure 3.1-1 (Negendahl 2015a)the definition of an integrated dynamic model is illustrated.

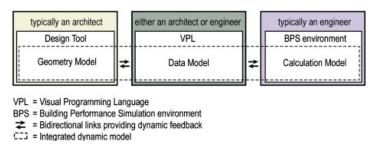


Figure 3.1-1: Definition of an integrated dynamic model (Negendahl 2015a)

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³ http://task23.iea-shc.org/description

It consists in the interaction of the design tools, often used by architects, and the calculation tools (Building Performance Simulation Tools), which normally are run by engineers. The main issue is to integrate both in a way that the use of the computer tools is maximized. Solving this, it would possible to manage a tool which gives both, aesthetic evaluation and technical feedback, allowing to identify structural problems in the first steps of the designing process. (Klitgaard, Kirkegaard, and Mullins 2006)

Both groups design and BPS tools are interconnected by means of Visual Programming Language (VPL). The model data, both geometry definition and calculations, is transferred through the different platforms in a bidirectional way. This concept is the basis to set up the method which is further explained in Part IV: Method (24).

Moreover, as it is state by Negendahl (Negendahl 2015b) the visual programing language is often based on Parametrical tools integrated in the Design softwares. These are for example, Grasshopper, the graphical algorithm editor tightly integrated with Rhino's 3-D, Autodesk® Dynamo Studio, software integrated in Autodesk Revit, as the standalone programming environment for the design, or Bentley's Generative Components. These platforms allow to the user to create multiple variables providing great flexibility to the designs.

In this scenario it should be contemplated that the idea of a common link, as is the VPL tools, contradicts the philosophy of BIM, as while "VPLs facilitate data across any content- and object relationship, schemas like IFC prescribe object relationships through attribute data" (Negendahl 2015b).

The idea of parametrical modeling leads to the second big topic of this research work. Accordingly with this, on the other hand, the second area of research is the optimization techniques to be developed with the aim of finding efficient and sustainable structures, where the minimal amount of mass for the overall construction is used.

The optimization process is a complicated practice which has been investigated for numerous researches over the years. Its complexity is mainly related that the practice of optimization is based on mathematical problems. Therefore in order to understand it numerous articles and books have been investigated, as the works from (Kirsch 1993), (Byrne et al. 2011), (Christensen and Klabring 2009), (Bendsøe 1989), (Christiansen et al. 2015), and (Sigmund 2001) etc. The aim of this thesis is incorporate the structural optimization process in early stages of the architectural designs, so it would be study from the design point of view based on the use of parametrical tools, leaving aside the mathematical formulation itself connected with programming. These aspects would be further on developed in 4.3 Optimization process description, as this section would set the basis of the optimization processes done for this thesis.

3.2 The Building Design Process and its tools

From the previous chapter, it is clearly seen the importance of the use of the different software along the long building process. They could be divided in two main different big categories: Design Tools and Building Performance Simulations Tools.

Inside the design tools group, there are found platforms such as Autodesk Revit, Rinhoceros 3D by Robert McNeel & Associates, ArchiCAD, etc.

Additionally, within the group of Building Performance Simulation Tools, softwares as Autodesk Robot Structural Analysis, FEM-Design, Staad.Pro of Bently, ANSYS, etc are found.

The features which provide each of these tools make the basis which establishes the working flow of the building process. Depending on their performance, the data exchange direction is set. However, regarding to this, the process can be unidirectional o bidirectional, as it has been introduced before. Within this context is important to also mention the agents who participate in each of the steps of this process.

In this way, the unidirectional approach is based on the traditional understanding of the designing process, which according to numerous researchers is called *linear approach* (Klitgaard, Kirkegaard, and Mullins 2006). It consists in the basic idea that is the architect in charge of the design, including both phases, sketching and creation of the 3D model; and then the engineer analyses the model once that it has been already finished, as it shown in the scheme a) Figure 3.2-1 (Klitgaard, Kirkegaard, and Mullins 2006).

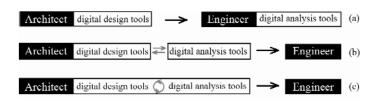


Figure 3.2-1: Three approaches for the working flow of the different agents in the building process. [Figure 2, (Klitgaard, Kirkegaard, and Mullins 2006), The three approaches – (a) the architects, (b) the hybrid practitioner and (c) the approach provided by the prototype, showing the initial sketching phase on the left side of the black arrow and the detailing phase on the right side.

This way of working is, in most of the cases, really time consuming. Probably, after the post analysis of the engineer, the model decided by the architect has to suffer changes, so it is necessary to go back in the process and do the steps again, repeating this until achieve a reliable configuration. The Figure 3.2-2 proposed by *Klitgaard (Klitgaard et al. 2006*) shows this linear approach.



Figure 3.2-2: linear approach. Working flow scheme proposed by Klitgaard (Klitgaard et al. 2006, figure 3)

Otherwise, the second case, the bidirectional approach, is then based on the idea of forward-back flow, where the data is transferred dynamically back and forward by VPL, as it was illustrated in Figure 3.1-1. This also corresponds with the scheme b) of the Figure 3.2-1 (Klitgaard, Kirkegaard, and Mullins 2006).

Therefore, it should be mentioned that exists different approaches when the idea of exchange data is studied. This is clearly explained by Negendahl 2014. The three different concepts are illustrated in the Figure 3.2-3, which could be understood in the following way: the first scheme corresponds to the traditional working process, which is usually based in the use of a simulation package. Additionally, the second scheme deals with the concept of having a central model where the data is shared by different softwares, which is clearly the idea of BIM. Lastly, the third scheme represents the lately introduced method which consists in a bidirectional process where the platforms are coupled by a middleware based on VLP. The main objective of this last approach is to use a tool which provides to the practitioner both, aesthetical and technical features (Klitgaard, Kirkegaard, and Mullins 2006), leading this again to the idea of the hybrid practitioner.

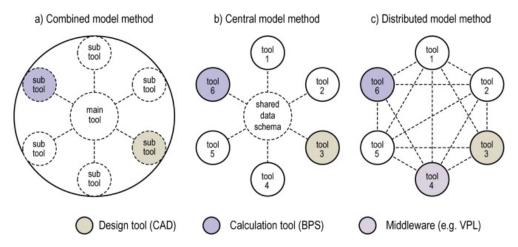


Figure 3.2-3: Differences between coupling methods [Negendahl 2014. Figure 6: a) combined model method (typically operated in a simulation package), b) central model method (using a central database/file format/schema), c) distributed model method (utilizing a middleware)]

Consequently, due to the differences between the idea of Building information modeling and Parametric design is should be found a balance where both interact, as it was stated by Boeykens (Boeykens 2012), "it is worthwhile to further elaborate the integration of these approaches in the design process, to benefit from advantages in both". This is the main objective of the method proposed in Part IV: Method, where different softwares, including both Parametric Design and BIM tools, are integrated in a flow working process.

3.2.1 Potential of visual programming language tools in the building Industry

The importance of the VPL tools is clearly stated in the previous introduction referred to the building design process. However, before enumerate its benefits; it is important to understand what *Parametric modelling* is.

The term parametric is originally related to mathematics. According to the science Oxford dictionary, parametric is all "Relating to or expressed in terms of a parameter or parameter", being defined parameter as "a numerical or other measurable factor forming one of a set that defines a system or sets the conditions of its operation" and within the mathematical field as "a quantity whose value is selected for the particular circumstances and in relation to which other variable quantities may be expressed"5. Later on, the term was related with design according to the work developed by Daniel Davis (Davis 2013), where it is stated that Moretty (1971) introduced the term "parametric architecture", which he defines as "the study of architecture systems with the goal of defining the relationships between the dimensions dependent upon the various parameters." (Davis 2013).

According to these definitions, it can be said that parametric modeling is based on the creation of several relations between certain numbers of parameters to create logic representations, which in relation with the architecture, means the definition of geometry based on links and nodes (Gallas et al., n.d.). In this way in parametric modeling the designer controls the creation of geometries from an overall logical script or scenario (Boeykens 2012).

Furthermore, it should also mention that the use of parametric modeling in the architectural field has led to a change of mentality. According to Woodbury in his book Elements of Parametric Design (Woodbury 2010), "parametric modelling introduces fundamental change: 'marks', that is, parts of the design, relate and change together in a coordinated way". In this way, while the old philosophy of the architects was "pencil, eraser and paper", they now have progressed to the ideal of "add, erase, relate and repair".

In this scenario, in relation with the work of M. Stavric and O. Marina, Parametric Modeling for Advanced Architecture, the development of digital design did not end with simple parametric modeling, it has taken a step ahead by using generative algorithms (Stavric and Marina 2011). Nowadays, these are integrated in the 3D design modeling tools, which makes easier for nonprogrammers to use, as they are Grasshopper, Dynamo or Generative Component from Bently.

At this point, we should ask ourselves, what are the real benefits of the parametrical modeling using this kind of tools?

The first benefit is the fact that they provide great flexibility to the designs. Determining a certain number of parameters, it is possible to play with the geometry until achieve the desired configuration. The parameters will vary depending on the last goal of the designer, being these parameters such as structural considerations, daylight, sun, angles, climatic data, energy limitations, etc. These parameters are often specified as number sliders, which are modified in accordance with different premises, creating different geometric models depending on the specific value.

⁵ http://www.oxforddictionaries.com/definition/english/parameter#parameter 5

⁴ http://www.oxforddictionaries.com/definition/english/parametric

This process helps to reduce the time required during the design phase, as it is possible to evaluate several configurations for a certain domain compared to the conventional 3D modeling tools.

Thanks to the new technology advances, every day appears new platforms which are fully integrated in the parametrical tools. This fact allows to the user to evaluate his designs from different perspectives, including both energy and structural aspects. For example, the plug-in Karamba for Grasshoper, allows to the user to perform Finite Element Calculations of the designed geometry in a dynamically way. This feature is the perfect example which link the labor of the architect and the engineer, as in order to use its features, it is necessary to have good structural knowledge.

Other benefit that provides these VPL tools is the possibility of performing form-finding and topological optimization processes. This is related with the idea of optimization that was earlier introduced. For example, in the case, optimization processes are carried out in grasshopper thanks to its component "Galapagos".

Furthermore, these parametrical modeling tools allow to the user to export the final desired model using the generic IFC format. This has a vital importance, as once that the parametric model design has been finished; it is possible to import it into a BIM software to continue with the corresponding analysis.

3.2.2 Softwares used for this master thesis

Once that it has been studied the different tools that coexist nowadays in the building industry, this section is meant to indicate the software used for the realization of this master thesis, including a brief explanation of their main features. Mainly, all the procedures have been done in the Grasshopper interface using the FEM plug-in Karamba, but other software have been also used in the process which is going to be explained in Part IV. (The descriptions have been taken from the corresponding webpages)

3.2.2.1 Grasshopper 3D. algorithmic modeling for Rinho⁶

It is developed by Scott Davidson and is a graphical algorithm editor tightly integrated with Rhino's 3-D modeling tools. Unlike RhinoScript, Grasshopper requires no knowledge of programming or scripting, but still allows designers to build form generators from the simple to the awe-inspiring.

It allows to install several plug-ins, in this case have been mainly used:

- GeomtryGym (Export data as IFC format)
- ToolBox (Inludes Galapagos listener)
- Karamba (FEM calculations)

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⁶ http://www.grasshopper3d.com/

3.2.2.2 Karamba Parametric engineering⁷

Karamba is developed by Clemens Preisinger in cooperation with Bollinger-Grohmann-Schneider ZT GmbH Vienna and it is a parametric structural engineering tool which provides accurate analysis of spatial trusses, frames and shells.

Karamba is fully embedded in the parametric design environment of Grasshopper, a plug-in for the 3d modeling tool Rhinoceros. This makes it easy to combine parameterized geometric models, finite element calculations and optimization algorithms like Galapagos.

3.2.2.3 Autodesk Revit⁸

Revit® BIM software powered by Atodesk Comunity is specifically built for Building Information Modeling (BIM), including features for architectural design, MEP and structural engineering, and construction. Revit is a single software application that supports a BIM workflow from concept to construction.

3.2.2.4 FEM – Design - StruSoft⁹

FEM-Design is an advanced modeling software powered by StruSoft for finite element analysis and design of load-bearing concrete, steel and timber structures according to Eurocode. The quick and easy nature of FEM-Design makes it ideal for all types of construction tasks from single element design to global stability analysis of large structures and makes it the best practical tool for structural engineers.

3.2.2.5 Autodesk Robot Structural Analysis¹⁰

Robot™ Structural Analysis Professional software powered by Autodesk Community provides engineers with advanced BIM-integrated analysis and design tools to understand the behavior of any structure type and verify code compliance

⁷ http://www.karamba3d.com/

⁸ http://www.autodesk.com/products/revit-family/overview

⁹ http://www.strusoft.com/products/fem-design

¹⁰ http://www.autodesk.com/products/robot-structural-analysis/overview

Part IV: METHOD

The objective of this part is to describe the method which has been set up and followed for this Master Thesis, with the aim of providing the necessary premises to the reader to be able to understand all the different approaches and calculations which has been done.

Accordingly, the first step has been to estipulate the main objective of the method; followed by the description of the process based on its flow illustration. This shows how the different tools which have been used form an integrated process, giving a clear understanding of how everything interacts and establishing the decisions and considerations which should be made in each step.

4.1 Objective

The main objective is to apply a highly effective workflow that integrates structural analysis in the early stages of the design process, combining design and calculation tools trying to solve the first question formulated for this master thesis work. This would allow to the designers to reduce the overall time of the design process stage, including the technical knowledge, as the structural verifications, from the beginning.

When we are talking about the method, it should be mention that two different types of methods exist. The first one is referred to the work process and how the data is transferred through the different platforms and which steps are done in each one (4.2 Applied integrated process description) the second one is more practical and is referred to the different processes

24

which have been carried out to achieve the optimal designs, based on different optimization processes (4.3 Optimization process description)

4.2 Applied integrated process description

The idea of integrated design process (IDP) was first introduced in section 3.1 Literature review based on the scheme of the Figure 3.1-1: Definition of an integrated dynamic model introduced by Negendahl (Negendahl 2015a), where the design tools, often use to create the geometries of the models, where connected to the BPS tools, which analyze the model configurations, by means of the VPL tools.

The parametrical tool Grasshopper can include certain plug-ins which allows to the practitioner to perform calculations inside the parametrical modeling frame. In this case, *Karamba* execute the FEM calculations for the structure modeled in grasshopper. This dynamic exchange of data allows to the engineer/architect to play with different designs (parametric modeling) obtaining an automatic feedback in terms of technical aspects, which speeds up the design process. In this way, the scheme of dynamic model (*Figure 3.1-1*: Definition of an integrated dynamic model has been developed to be applied to this specific method. As it can be seen the exchange of data between Rinho 3D, Grasshopper and Karamba is produce dynamically way. When the desired design is achieved, this model can be exported to another FEM software to continue with more detailed calculations.

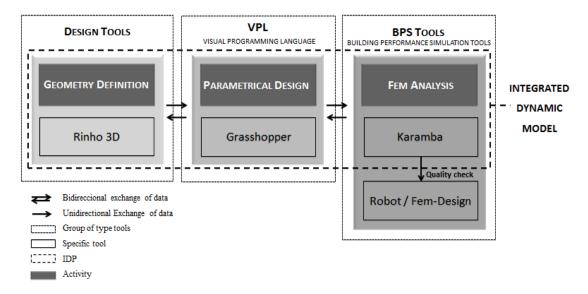


Figure 4.2-1: Developed Integrated Dynamic model

Consequently in this context, the following diagram (Figure 4.2-2) shows the working flow based on VPL (Parametrical design tools), which is Grasshopper in this case. It is divided in five different stages, which go from the definition to the geometry until reach the group of optimal configurations, going through FEM calculations and optimization processes.

In the first stage is necessary to input a series of parameters, which will vary depending on the model geometry. Furthermore, the structural definition (stage 3) will also depend on the configuration and it requires technical knowledge in order to set it correctly.

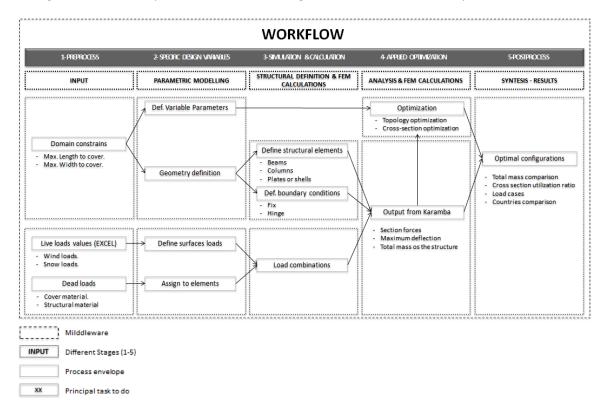


Figure 4.2-2: Workflow for the specific method. Steps carried out in each stage.

As a result, it is obtained the performance of different geometry models, knowing their maximum deflection, the total amount of mass of the structure, the utilization ratio of the cross-sections, etc. This allows to the practitioner to choose the best solution basing his decision on the parameters that he considers or the requirements that he must fulfill to accomplish the desires of the client.

In Part V: Results-Structural Analysis (5.2) different model configurations will be studied, and the workflow will be adapted to each of them, indicating the parameters considered in each case.

Furthermore based on the previous schemes (Figure 4.2-1 and Figure 4.2-2), the following diagram (Figure 4.2-3) shows the connection between the different platforms used, giving an overview of how the Design Tools, VPL tools and BPS tools are connected. Additionally, here the math and spreadsheets softwares have been also added.

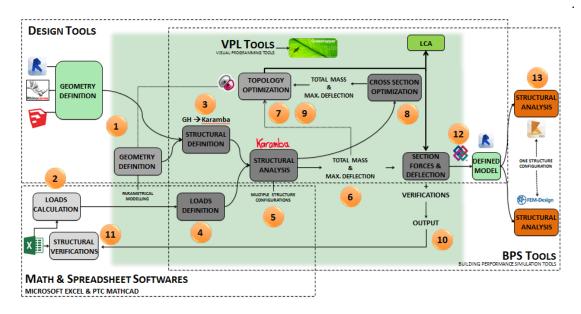


Figure 4.2-3: Applied Integrated design process

It is clearly seen that again the VPL tools serve as a common link between the other three type of tools (Design, BPS and Spreadsheets tools).

In this way the brief description of the process is as follows:

- **1.** *First step: Creation of the geometry* either in the design tools (Rinho 3D, Sketch up or Revit) or in the Grasshopper interface.
- 2. **Second step: Loads calculations** in the Excel spreadsheet (Calculations for all the Nordic countries)
- **3.** *Third step: From GH to Karamba*. Transformation of all the geometry elements into structural elements: beams, columns or plates. Definition of the boundary conditions (supports) and dimensions of the cross-sections (approximate).
- **4.** Fourth step: Definition of the loads in Karamba. Specification of the different wind and snow load areas on the model reading the values from excel. Integration of the load combinations in Grasshopper.
- 5. Fifth step: Perform the FEM analysis using Karamba.
- **6.** *Sixth step: Analysis of the results.* Check that structure doesn't collapse with the estimated cross-sections. Find ways to improve it.
- **7. Seventh step: Topology optimization 1.** Perform the 1st optimization process using Galapagos component for GH.
- **8.** *Eight step: Cross-section optimization.* Perform the 2nd optimization process using the component for Karamba, "Cross-section optimization"
- **9. Ninth step: Topology optimization 2.** Perform the 3rd optimization process using Galapagos based on the results from the 8th step cross-cross section optimization.
- **10.** *Tenth step: Export the results to excel.* Comparison for the total mass of the structure before and after the cross section optimization.
- **11.** *Eleventh step: Analysis of the configurations*. Extract the different combinations, analyse the optimal ones based on the amount of mass and the utilization ratio of the

cross sections. Check that the elements of the structure fulfil the requirements specified in the Eurocode. Repite the optimization loop from step 7 if it is necessary.

- **12.** Twelfth step: Export the chosen geometry (IFC format). Based on the analysis of the step 11th, choose one of the optimal configurations and export it to Revit using the plug-in for GH GeometryGym.
- **13.** Thirteenth step: Analysis of the Structure in FEM software. Analyses of the structure either in Robot or FEM-design. Compare the section forces values with the ones calculated from Karamba.

4.3 Optimization process description

The last diagram leads to introduce the method followed for the optimization process. As it was presented before, this procedure implies the use of mathematical programing techniques as it is stated in the paper of Y.M. Xie and G.P.Steven (Xie and Steven 1993). For example the work developed by O. Sigmund presents a Matlab script to carry out topology optimization for compliance minimization of statically loaded structures (Sigmund 2001).

However, the objective is to combine those practices coming from the engineer field within the architectural designs. This is study by means of the evolutionary structural design computation by Byrne (Byrne et al. 2011) . The concept is based in an evolutionary search of reducing the material usage and cost of the structure, but being this able to resist the stresses produced by the loads without collapse. This is done using Finite Element Methods.

In this way, the new architectural-engineers approaches tend to see the optimization processes as a way to generate aesthetic and efficient structural configurations. (Block et al. 2014). Numerous studies show that this combination adds great value to the projects.

In the work developed by Peter W. Christensen and Anders Klarbring, "An Introduction to Structural Optimization" (Christensen and Klabring 2009) it is formulated the optimization process as: "minimize f(x,y) with respect to x and y; subject to behavioral constraints on y, and design constrains on x" Where "f" is the Objective function, "x" is the design variable changed during the optimization, and "y" is the state variable which is the response of the structure.

In the case of this master thesis, the optimization procedures will be carried out thanks to the component Galapagos of Grasshopper, which is an evolutionary solver developed by David Rutten. This is a generic platform for the application of Evolutionary Algorithms to be used on a wide variety of problems by non-programmers. ¹¹ To run Galapagos is necessary to define the goal of the optimization, called as Fitness value (objective function) and the parameters to optimize, called as genes (design variables) (Rutten 2016). In this way, the state variables would be the structural verifications (stresses, buckling and deflection) that the structure should fulfill.

¹¹ http://www.grasshopper3d.com/group/galapagos

Before described the optimization method, it is important to clarify that for this particular project "optimization" is understood as the semi-automated process of finding feasible structural configurations through an integrated dynamic model (coupled to an optimization algorithm). Hence, the optimization is only applied to the design of elements that concerns structure of the selected design principle, and very important to state, do not apply to other disciples, such as energy performance, cost, functional requirements, aesthetics etc.

In this way, in order to set up the optimization process (steps 7,8 and 9), it is important to explain the types of procedures which exits. According to (Christensen and Klabring 2009) there are three types of structural optimization problems:

- Shape optimization. It is referred to the boundary of the structural domain.
- Sizing optimization. It is referred to the cross-sections of the elements of the structure.
- Topology optimization. It is the process of form-finding the optimal shape and configuration of the structure. In (Nyman and Andreas 2015) is define as the "discipline of computing the optimal shape and topology of a structure with respect to some desired effect" (Bendsøe and Kikuchi, 1988; Bendsøe and Sigmund, 2003)

The last two processes have been implemented in the Grasshopper script, as the first one (domain) is fix. Below the process of optimization is described in seven steps. These steps may be read as if they were included in a larger loop that continues until "optimality" has been found.

To be able to carry the topology optimization is necessary to pre-define a structure configuration, setting its boundary conditions and cross-sections. After this, it is run Galapagos, defining the genes and the fitness for each case (*Seventh step: Topology optimization 1.* Perform the 1st optimization process using Galapagos component for GH.), followed by the second optimization step (*Eight step: Cross-section optimization.* Perform the 2nd optimization process using the component for Karamba, "Cross-section optimization". Thanks to the component Optimized Cross Section' from *Karamba*, it is possible to run at the same time the *Ninth step: Topology optimization 2.* Perform the 3rd optimization process using Galapagos based on the results from the 8th step – cross-cross section optimization. and record the values of both processes (Galapagos listener). This second topology optimization considers the cross sections defined in step 8.

It should be mentioned that the 'Optimized Cross Section'-component from *Karamba* takes into account the buckling of each specific member depending on the buckling length (6.5.4 Buckling Modes, Preisinger 2015), however when the results are verified analytically, not always fulfill the requirements against buckling. This component considers the load bearing capacity of each member, checking that is sufficient for all the load cases of the structure. This is done following the requirements established in the Eurocode 1993-1-1. The values Cmy and Cmz are limited to a maximum of 0.9 (Section 6.5.9, Preisinger 2015)]. It is also taken into account the lateral torsional buckling of the elements according to the cross-section classification for class 1 to 4 (*Table 5.2* Eurocode3 2001). See also *Cross section class*.

29

Furthermore, as according to (Eurocode3 2001) there is a limitation of the maximum deflection of the components which form a structure; the maximum deflection allowed has been also implemented in the component in order to provide the most efficient result for the cross section.

This process is schematized in the following diagram, described in the following steps:

- Study of the space
- Optimization: from basic construction to a complex design
 - 1.Load analysis: wind, snow and self-weight
 - 2. Structural analysis single member: theory and limitations
 - 3. Pre-definition of the cross sections
 - 4.First optimization process (Step 7 topology) → Goals: minimal deflection and minimal mass of the structure.
 - 5. Second optimization process (Step 8 sizing) → optimal cross section working at 90% of their capacity
 - 6. Third optimization process (Step 9 topology) Goals: minimal deflection and minimal mass of the structure.
 - 7.Structural verifications. If the members defined do not fulfill the requirements a/EC repeat the optimization loop until a suitable solution is achieve.

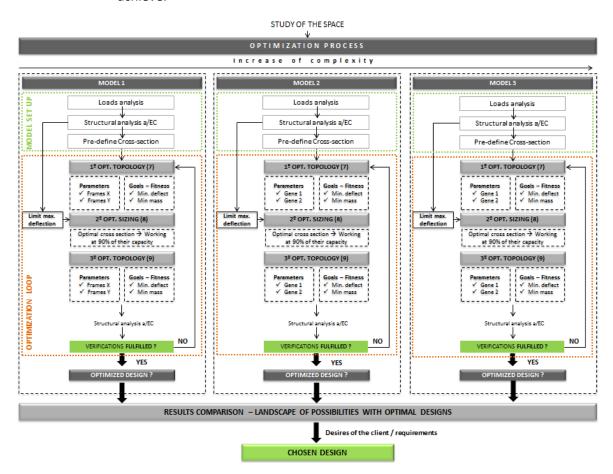


Figure 4.3-1: Optimization method scheme

4.4 Normative

The structural verifications should be implemented in the method described before; therefore this section establishes all the necessary theory regarding the structural calculations. It is based on the premises specified by the Eurocode, including all the conditions set by the corresponding National Annex.

It should be noted that this chapter formulate all the basic equation to carry out the calculations, but express in a general way. Further on, in Part V: Structural analysis, it would be mentioned the different assumptions which have been done for each specific case.

First are going to be explained the procedures to calculate the wind loads, followed by the calculation of the snow loads, and in last instance, all the necessary formulations to estimate the bearing capacities for the steel elements and their verifications.

4.4.1 Wind loads calculations

The wind loads are calculated following the specifications in the Eurocode 1, part four (DS/EN 1991-1-4 2007) and the corresponding National Annex for each country.

As it is specified in the section 3 of DS/EN 1991-1-4: 2007, the wind actions will depend on each design situation, which means that for each of the models they will vary, but all follow the same calculation procedure which will be described as follows:

4.4.1.1 Basic wind velocity (DS/EN 1991-1-4 2007, eq. no.4.1)

$$v_b = c_{dr} c_{season} v_{b,0} [m/s]$$

$$\tag{4.1}$$

Where:

- V_b is the basic wind velocity, defined as a function of wind direction and time of year at 10 m above ground of terrain category II
- V_{b,0} is the fundamental value of the basic wind velocity
- c_{dr} is the directional factor
- c_{season} is the season factor

4.4.1.2 Mean wind velocity (DS/EN 1991-1-4 2007, eq. no.4.3)

$$v_m(z) = c_r(z) c_0(z) v_b [m/s]$$
 (4.2)

Where:

- $c_r(z)$ is the roughness factor calculated according section 4.3.2 DS/EN 1991-1-4 2007
- $c_0(z)$ is the origraphy factor calculated according section 4.3.3 DS/EN 1991-1-4 2007

4.4.1.3 Wind turbulence (DS/EN 1991-1-4 2007, eq. no.4.6)

$$\sigma_v = k_r \, v_h \, k_l \tag{4.3}$$

Where:

- k_r is the terrain factor calculated according eq. 4.5 DS/EN 1991-1-4 2007
- v_b is the basic wind velocity.
- $lacktriangledown k_l$ is the turbulence factor. This value may be given in the National Annex. The recommended value is 1.0
- 4.4.1.4 Turbulence intensity (DS/EN 1991-1-4 2007, eq. no.4.7)

$$I_{v}(z) = \frac{\sigma_{v}}{v_{m}(z)} = \frac{k_{l}}{c_{0}(z)\ln(\frac{z}{z_{0}})} \text{ for } z_{min} \le z \le z_{max}$$
 (4.4)

$$I_v(z) = I_v(z_{min}) \text{ for } z \le z_{min}$$

$$\tag{4.5}$$

Where:

- $c_0(z)$ is the orography factor calculated according section 4.3.3 DS/EN 1991-1-4 2007
- z_0 is the roughness length, given in *Table 4.1 DS/EN 1991-1-4 2007*.
- $lack* k_l$ is the turbulence factor. This value may be given in the National Annex. The recommended value is 1.0

4.4.1.5 Peak velocity presure (DS/EN 1991-1-4 2007, eq. no.4.8)

$$q_p(z) = [1 + 7I_v(z)] \frac{1}{2} \rho \ v_m^2(z) = c_e(z) q_b \ [kN/m^2]$$
 (4.6)

Where:

- ρ is the air density, which depends on the altitude, temperature and barometric pressure to be expected in the region during wind storms. In all the cases is taken as 1.25 kg/m³
- $c_e(z)$ is the exposure factor given by eq. 4.9 DS/EN 1991-1-4 2007 [$c_e(z) = \frac{q_p(z)}{q_h}$]
- q_b is the basic wind velocity pressure given by eq. 4.10 DS/EN 1991-1-4 2007 $[q_b=\frac{1}{2}~\rho~v_0^2~]$
- $I_{\nu}(z)$ is the turbulence intensity specified in eq. (4.4) or (4.5)

4.4.1.6 Wind pressures on surfaces (DS/EN 1991-1-4 2007, eq. no.4.1)

$$w_e = q_n(z_e)c_{ne} (4.7)$$

Where:

- $q_p(z_e)$ is the peak velocity pressure specified in eq. (4.6)
- z_e is the reference height for the external pressure. [Section 7 DS/EN 1991-1-4 2007]
- c_{pe} is the pressure coefficient for the external pressure. [Section 7 DS/EN 1991-1-4 2007]

4.4.2 Snow loads calculations

Accordingly, the snow loads are calculated following the specifications in the Eurocode 1, part three (DS/EN 1991-1-3:2007 2007) and the corresponding National Annex for each country.

In the same way that for wind loads, the snow loads will depend on each design situation, which means that for each of the models they will vary. The characteristics causing different values are stated in section 5.1 Nature of the load (DS/EN 1991-1-3:2007) and are:

- a) The shape of the roof.
- b) Its thermal properties.
- c) The roughness of its surface.
- d) The amount of heat generated under the roof.
- e) The proximity of nearby buildings.
- f) The surrounding terrain.
- g) The local meteorological climate, in particular its windiness, temperature variations, and likelihood of precipitation.

Furthermore, the load arrangement can be considered as "undrifted snow load on roods" or as "drifted snow load on roofs". The first one is defined by section 1.6.5 DS/EN 1991-1-3:2007 as "load arrangement which describes the uniformly distributed snow load on the roof, affected only by the shape of the roof, before any redistribution of snow due to other climatic actions". Meanwhile, the second one is defined by section 1.6.6 DS/EN 1991-1-3:2007 as "load arrangement which describes the snow load resulting from snow having been moved from one location to another location on a roof, e.g. by the action of the wind".

In case of the snow load calculation, there are three different situations:

4.4.2.1 Persistent / transient design situation

$$S = \mu_l C_{\rho} C_t S_k \tag{4.8}$$

4.4.2.2 Accidental design situations where exceptional snow load is the accidental action

$$s = \mu_l C_e C_t s_{Ad} \tag{4.9}$$

4.4.2.3 Accidental design situations where exceptional snow drift is the accidental action

$$S = \mu_l \, S_k \tag{4.10}$$

Where:

- μ_l is the snow load shape coefficient, it will depends on the shape of the roof (two different cases will be calculated, explained in Part V: Structural Analysis)
- s_k is the characteristic value of snow load on the ground specified in the corresponding National Annex for each country.
- s_{Ad} is the pressure coefficient for the exernal pressure. [Section 7 DS/EN 1991-1-4 2007]
- C_e is the exposure coefficient specified in *Table 5.1 DS/EN 1991-1-4 2007*. In this case it is taken as 1,0 for Normal topography.
- C_t is the thermal coefficient. It is taken as 1, as the heat transmittance is larger than 1 W/m²K

4.4.3 Steel Calculations

The necessary steel verifications, including the checks regarding the capacities of the cross section, the buckling resistance, the lateral torsional resistance and the requirements for the deflections of the elements, are calculated according to the premises in Eurocode 3: Design of steel structures (Eurocode3 2001).

It should be mention that depending on the element which is being calculated the different coefficients to adopt will vary, so all the equations are listened in their general form, and the corresponding considerations will be made for each case during the calculation process.

The verifications are divided for the two main elements which form the structure; being on the one side the verifications regarding the beams and on the other side the verifications for columns.

Beams Verifications

The following formulations show the procedure which has been followed to check the beams cross section capacities. First must be specified the steel class and the section properties.

The beam capacity control is done for Ultimate Limit States (ULS) for members subjected to bending and shears forces. The resistance of the cross section according to the *clause 6.2* (6) of the Eurocode 3: Design of Steel structures (Eurocode3 2001) shall be verified by finding a stress distribution which equilibrates the internal forces and moment without exceeding the yield strength. As a general rule the stresses are calculated following *Navier's formula*, stated in the equeqtion (3.11)

$$\sigma = \frac{N}{A} + \frac{M_y}{I_Y} z - \frac{M_z}{I_z} y \le f_{yd} \left[\text{kN/m}^2 \right]$$
(4.11)

Furthermore, the individual checks for axial, shear and bending must be also fulfilled, being the design forces smaller than the capacities.

4.4.3.1 Cross section class

For the internal compression parts (web of the profile), case: Part subject to bending. *Table 5.2* (Eurocode3 2001)

Section class 1
$$\rightarrow \frac{c}{t} \leq 72 \ \varepsilon$$
 (4.12)
Section class 2 $\rightarrow \frac{c}{t} \leq 83 \ \varepsilon$
Section class 3 $\rightarrow \frac{c}{t} \leq 124 \ \varepsilon$
Section class 4 $\rightarrow \frac{c}{t} > 124 \ \varepsilon$

Where:

- c is the efficient length of the web, calculated as: $c = h_w 2 t_f$, being h_w the height of the profile and t_f the thickness of the flange.
- t is the thickness of the web (t_w)
- $\varepsilon = \sqrt{235/f_y}$

For the outstand flanges (flange of the profile, Table 5.2 (Eurocode3 2001)

Section class 1
$$\rightarrow \frac{c}{t} \leq 9 \, \varepsilon$$
 (4.13)
Section class 2 $\rightarrow \frac{c}{t} \leq 10 \, \varepsilon$
Section class 3 $\rightarrow \frac{c}{t} \leq 14 \, \varepsilon$
Section class 4 $\rightarrow \frac{c}{t} > 14 \, \varepsilon$

Where:

- c is the efficient length of the web, calculated as: $c = \frac{b t_w}{2} r$ being b the width of the profile, t_w the thickness of the web and r the root radius.
- t is the thickness of the flange (t_f)
- $\varepsilon = \sqrt{235/f_y}$

4.4.3.2 Bending moment Capacity control

The cross section must fulfill:

$$\frac{M_{Ed}}{M_{c,Rd}} \le 1 \tag{4.14}$$

Where the Bending resistance is calculated according to the cross-section class:

Class 1 or 2
$$\rightarrow$$
 $M_{c,Rd} = \frac{W_{pl} f_y}{\gamma_{M0}}$ [kNm]

Class 3 \rightarrow $M_{c,Rd} = \frac{W_{el,min} f_y}{\gamma_{M0}}$ [kNm]

Class 4
$$\rightarrow$$
 $M_{c,Rd} = \frac{W_{eff,min} f_y}{\gamma_{M0}}$ [kNm]

Where:

- W_{pl} is the Plastic modulus of Elasticity [cm³]
- $W_{el.min}$ is the min Elastic modulus of Elasticity [cm³]
- W_{eff} is the Effective modulus of Elasticity [cm³]
- f_v is characteristic yield strength [kN/cm²]
- γ_{M0} is the partial safety factor specified in Section 6.1 (Eurocode3 2001)

4.4.3.3 Shear Capacity control

The cross section must fulfill:

$$\frac{V_{Ed}}{V_{pl,Rd}} \le 1 \tag{4.16}$$

The shear resistance is calculated as follows:

$$V_{pl,Rd} = \frac{A_v (f_y/\sqrt{3})}{\gamma_{M0}} \text{ [kN]}$$
 (4.17)

Where:

• A_v is the shear area [cm³]. For rolled sections is calculated as:

$$A_{v} = A - 2bt_{f} + (t_{w} + 2r)t_{f} [kN]$$
(4.18)

- f_y is characteristic yield strength [kN/cm²]
- γ_{M0} is the partial safety factor specified in Section 6.1 (Eurocode3 2001)

4.4.3.4 Interaction of shear and bending moment

According to the *clause 6.2.8 (2)* (Eurocode3 2001), the effect of the plastic shear resistance of the cross-section on the moment resistance should be taken into account when the applied shear force is larger than the half of the shear resistance.

$$\frac{V_{pl,Rd}}{2} < V_d \to M_{y,V,Rd} = \frac{(W_{pl,y} - \rho A_w^2 / 4t_w) f_y}{\gamma_{M0}} \le M_{y,c,Rd}$$
(4.19)

Where:

- $V_{pl,Rd}$ is the plastic shear resistance calculated according to eq. (4.17)
- V_d is the design shear resistance
- W_{pl} is the Plastic modulus of Elasticity [cm³]
- f_v is characteristic yield strength [kN/cm²]
- γ_{M0} is the partial safety factor specified in Section 6.1 (Eurocode3 2001)

4.4.3.5 Lateral torsional buckling

Non-dimensional slenderness

$$\lambda_{Lt} = \sqrt{\frac{W_y f_y}{M_{cr}}} \tag{4.20}$$

Where:

- M_{cr} is the critical bending moment of the cross-section. Its calculation is done based on the cases specified in the Lateral-torsional buckling (LTB) loads tables from Teknisk Ståbi (Jensen 2007)
- W_{ν} is the modulus of Elasticity [cm³], which will be equal to:
 - $W_v = W_{pl,v}$ for cross-section class 1 or 2
 - $W_v = W_{el,v}$ for cross-section class 3
 - $W_v = W_{eff,v}$ for cross-section class 4

Imperfection factor

$$\phi_{Lt} = 0.5 \left[1 + \alpha_{LT} (\lambda_{LT} - \lambda_{LT,0}) + \beta \lambda_{LT}^2 \right]$$
 (4.21)

Where:

- $\lambda_{LT,0} = 0.4$ and $\beta = 0.75$ for rolled sections according to the National Annex.
- α_{LT} is the imperfection factor determined by the buckling curved according to the table 6.1 (Eurocode3 2001)

Reduction factor for lateral torsional buckling eq. (6.47) (Eurocode3 2001)

$$\chi_{Lt} = \frac{1}{\phi_{LT} + \sqrt{\phi_{LT}^2 - \beta \lambda_{LT}^2}} \le \begin{cases} 1\\ 1/\lambda_{LT}^2 \end{cases}$$
(4.22)

Where:

- ϕ_{LT} is calculated according to the equation (3.21)
- λ_{Lt} is calculated according to the equation (3.20)

Lateral Torsional Buckling capacity check eq. (6.46) (Eurocode3 2001)

$$M_{Ed} \le M_{hR} = \gamma_{LT} W f_{V} / \gamma_{M1}$$
 (4.23)

Where:

- W_y is the modulus of Elasticity [cm³], which will depend on the cross-section class as it was specified before.
- χ_{LT} is the reduction factor calculated according to eq. (4.21)

Columns Verifications

The verifications necessaries to carry out to check the cross-sections of the columns follow similar procedure than the specified for beams. First must be determined the steel class and the section properties.

In the case of the columns, the axial force will drive the final cross section, as the elements are subjected to compression forces. The capacity control is done for Ultimate Limit States (ULS).

According to the Eurocode 3: Design of Steel Structures (Eurocode3 2001), when the elements are subjected to bending and axial compression the following verifications should be carried out for cross sections such as IPE, IPN, HEB, etc.

$$\sigma = \frac{N_{Dc}}{\chi_{v}A} + k_{y,LT} \frac{M_{dy}}{\chi_{LT}W_{pl,y}} + k_{z} \frac{c_{m,z} M_{dzy}}{W_{pl,z}} \le f_{yd}$$
(4.24)

Furthermore, the individual checks for axial, shear and bending must be also fulfilled, being the design forces smaller than the capacities.

4.4.3.6 Cross section class

For the internal compression parts (web of the profile), case: Part subject to compression). *Table 5.2* (Eurocode3 2001)

Section class 1
$$\rightarrow \frac{c}{t} \le 33 \, \varepsilon$$
 (4.25)
Section class 2 $\rightarrow \frac{c}{t} \le 38 \, \varepsilon$
Section class 3 $\rightarrow \frac{c}{t} \le 42 \, \varepsilon$
Section class 4 $\rightarrow \frac{c}{t} > 42 \, \varepsilon$

Where:

- c is the efficient length of the web, calculated as: $c = h_w 2 t_f$, being h_w the height of the profile and t_f the thickness of the flange.
- t is the thickness of the web (t_w)
- $\varepsilon = \sqrt{235/f_y}$

For the outstand flanges (flange of the profile, Table 5.2 (Eurocode3 2001)

Section class 1
$$\rightarrow \frac{c}{t} \leq 9 \, \varepsilon$$
 (4.26)
Section class 2 $\rightarrow \frac{c}{t} \leq 10 \, \varepsilon$
Section class 3 $\rightarrow \frac{c}{t} \leq 14 \, \varepsilon$
Section class 4 $\rightarrow \frac{c}{t} > 14 \, \varepsilon$

Where:

- c is the efficient length of the web, calculated as: $c = \frac{b t_w}{2} r$ being b the width of the profile, t_w the thickness of the web and r the root radius.
- t is the thickness of the flange (t_f)
- $\varepsilon = \sqrt{235/f_y}$

4.4.3.7 Buckling length

The buckling length is determined as:

- $L_{xy} = 0.7 L_{column}$ for columns in connection with the foundation.
- $L_{xy} = L_{column}$ for intermediate columns.

4.4.3.8 Flexural buckling around the y-axis and z-axis

For both axes is done the same procedure. It should be taken the corresponding moment of inertia depending on the axis that is being calculated. It should be calculated the design buckling resistance which is:

$$N_{b.Rd} = \frac{\chi A f_y}{\gamma_{M1}} \le N_{Ed}$$
 for Class 1,2 and 3 cross-sections (4.27)
 $N_{b.Rd} = \frac{\chi A_{eff} f_y}{\gamma_{M1}} \le N_{Ed}$ for Class 4 cross-sections

The general procedure to calculate $N_{b.Rd}$ is as follows:

Critical axial force:

$$N_{cr,y} = \frac{\pi^2 E_d I_y}{l_{xy}^2} \tag{4.28}$$

Where:

- E_d is the design modulus of elasticity.
- I_y is the moment of inertia around the y axis (or I_z for the second case).
- L_{xy} is the buckling length.

Non-dimensional slenderness

$$\lambda = \sqrt{\frac{Af_y}{N_{cr}}}$$
 for Class 1,2 and 3 (4.29)
$$\lambda = \sqrt{\frac{A_{eff}f_y}{N_{cr}}}$$
 for Class 4

Where:

 M_{cr} is the critical bending moment of the cross-section. Its calculation is done based on the cases specified in the Lateral-torsional buckling (LTB) loads tables from Teknisk Ståbi (Jensen 2007) • W_{ν} is the modulus of Elasticity [cm³], which will be equal to:

- $W_{\nu} = W_{pl,\nu}$ for cross-section class 1 or 2

- $W_v = W_{el,v}$ for cross-section class 3

- $W_v = W_{eff,v}$ for cross-section class 4

Imperfection factor

$$\phi = 0.5 \left[1 + \alpha(\lambda - 0.2) + \lambda^2 \right] \tag{4.30}$$

Where:

- λ is the non-dimensional slenderness eq. (4.28)
- α is the imperfection factor determined by the buckling curved according to the *table* 6.1 (Eurocode3 2001)

Reduction factor for lateral torsional buckling eq. (6.47) (Eurocode3 2001)

$$\chi = \frac{1}{\phi + \sqrt{\phi^2 - \lambda^2}} \le \begin{cases} 1\\ 1/\lambda_{LT}^2 \end{cases} \tag{4.31}$$

Where:

- ϕ is calculated according to the equation (4.29)
- λ is calculated according to the equation (4.28)

4.4.3.9 Lateral torsional buckling

Non-dimensional slenderness

$$\lambda_{Lt} = \sqrt{\frac{W_y f_y}{M_{cr}}} \tag{4.32}$$

Where:

• M_{cr} is the critical bending moment of the cross-section determined by:

$$M_{cr} = \frac{C_1 \pi^2 E I_z}{L^2} \left(\frac{I_w}{I_z} + \frac{L^2 G I_T}{\pi^2 E I_z} \right)^{0.5}$$
 (4.33)

- W_{ν} is the modulus of Elasticity [cm³], which will be equal to:
 - $W_{\nu} = W_{pl,\nu}$ for cross-section class 1 or 2
 - $W_{\nu} = W_{el,\nu}$ for cross-section class 3
 - $W_v = W_{eff,v}$ for cross-section class 4

Imperfection factor

$$\phi_{Lt} = 0.5 \left[1 + \alpha_{LT} (\lambda_{LT} - \lambda_{LT,0}) + \beta \lambda_{LT}^2 \right]$$
 (4.34)

Where:

- $\lambda_{LT,0} = 0.4$ and $\beta = 0.75$ for rolled sections according to the National Annex.
- α_{LT} is the imperfection factor determined by the buckling curved according to the table 6.1 (Eurocode3 2001)

Reduction factor for lateral torsional buckling eq. (6.47) (Eurocode3 2001)

$$\chi_{Lt} = \frac{1}{\phi_{LT} + \sqrt{\phi_{LT}^2 - \beta \lambda_{LT}^2}} \le \begin{cases} 1\\ 1/\lambda_{LT}^2 \end{cases}$$
 (4.35)

Where:

- ϕ_{LT} is calculated according to the equation (3.21)
- λ_{Lt} is calculated according to the equation (3.20)

Lateral Torsional Buckling capacity check eq. (6.46) (Eurocode3 2001)

$$M_{Ed} \le M_{h,R} = \chi_{LT} W f_{\nu} / \gamma_{M1}$$
 (4.36)

Where:

- W_y is the modulus of Elasticity [cm³], which will depend on the cross-section class as it was specified before.
- χ_{LT} is the reduction factor calculated according to eq. (3.35)

Deflection verifications

4.4.3.10 Maximum vertical deflection

According to Section 7.2.1 (B), the vertical deflection for beam in roof and plates should be:

$$u_{max,v,beams} = l_1 / 200 \tag{4.37}$$

$$u_{max,v,roof} = l_2 / 150$$
 (4.38)

Where:

- l_1 is the span of simple supported and continuous beams.
- l_2 is the span of the roof plate.

4.4.3.11 Maximum deflection columns

According to Section 7.2.1 (B), the maximum horizontal deflection for columns in frames is:

$$u_{max,x} = h / 300 (4.39)$$

Where:

h is the height of the building.

Part V: RESULTS - Structural Analysis

The purpose of this chapter is to apply the method which has been proposed in the previous chapter (Part IV, 4.1), in order to study its feasibility, and study its advantages as well as disadvantages compare to the traditional working approaches. In this way it would be possible to discuss and solve the research questions posed for this master thesis.

As it was introduced in Part II (2.2), the project to study is the EKO-Canopy project [APPENDIX A: THE EKO-CANOPY PROJECT], a proposal of White Architects Sweden. The project was chosen for the great flexibility which provides, as the purpose is to build a Canopy between two existing apartment blocks. This means that is possible to play by means of the parametrical modeling with all the space in between to create a realistic and sustainable configuration.

First it is going to be explained the load analysis which has been done in appliance to the specific construction. Secondly, the method based on the parametrical model is developed and its results presented.

The following table shows the measurements for the buildings and the space in between which are essential in order to understand the considerations which have been made.

Parameter	Unit [m]	Apartment block	Space in between
Height	m	22.63	22.63
Length	m	72.57	72.57
Width	m	12.1	28.25

Table 4.4-1: Basic measurements of the site for the Canopy construction

5.1 Loads analysis

This section is focus on the study and analysis of the loads which act on the structure of the EKO-Canopy project. They are calculated following the specifications on (DS-EN 1990 – Basis of structural design 2007, sec. 4.1.1 Classifications of actions), where the actions are classified as *Permanent actions* (G), *Variable actions* (Q) and *Accidental actions* (A). For the purpose of this project only the first two categories are going to be considered. The *permanent actions* (G) are referred to the self-weight of the structure, while the *variable actions*, Q, are the wind loads, snow loads and imposed loads. In this specific case the last ones are neglected as the model taken as study, as it has been described before, is a Roof-Canopy which is not accessible for public.

Firstly are going to be calculated the variable actions. These are the different live loads which will influence the design, as the wind loads and the snow loads. Most areas of the Nordic countries have strong climatological conditions during the autumn-winter period every year; therefore a deep study of the wind and snow loads should be done. For this reason, as this master thesis form part of the NORDIC BUILD PROGRAMME (2.1), these are going to be estimated in different locations along the Nordic countries, following the premises specified in *Section 4.4*, both *4.4.1 Wind loads calculations* and *4.4.2 Snow loads calculations*.

Analyzing the loads for different locations will be used as a basis to evaluate how the loads could condition the design. Hence the wind pressures and snow loads are calculated for Sweden, Norway, Denmark, Finland and Iceland. However it should be noted that the main calculations are performed for the loads considering the climatological data from *Upplands Väsby*, Sweden, the original location of the project which is being studied.

Secondly, the permanent actions will be estimated. They consist mainly in the self-weight of the different elements of the structure. This is being automatically assumed for the calculations by the structural plug-in *Karamba* in *Grasshopper*, so its implementation on the script would be explained. The parametrical modeling using Karamba allows to easily change the material for the structure, however in order to verify the capacities of the structural elements are necessary a serial of checks, which requires a more deeply post processing analysis. Consequently, for the main models only steel has been considered as structural material. It has been considered primordial evaluate how considering a specific structural material it can be set a method which seeks for one of the optimal configurations based on form-finding and optimization processes.

5.1.1 Live loads analysis in the Nordic countries: Wind and Snow

5.1.1.1 Wind calculations

The wind actions will depend on each design situation, as it is specified in the section 3 of DS/EN 1991-1-4: 2007, which means that for each of the models they will vary. They act as pressures on the different surfaces of the EKO-Canopy, (roof and facades), as normal forces to the structural components or the cladding. The load component for *Karamba* allows to distribute the surface load on the area and project it according to the global coordinate planes (Preisinger 2015, bk. Karamba User Manual), so it will change as the parametrical model changes.

In order to determine the basic wind velocity, [eq. (4.4.1.1)], it is necessary to know the fundamental value of the basic wind speed, $v_{b,0}$, which procedure calculation is given for the corresponding National Annex. This value will then change with the different locations to be studied. In this way, the following table 5.1-1 shows the data considered for each case:

Country	Chosen location	<i>v_{b,0}</i> [<i>m</i> /s]	Reference
Denmark	Copenhagen	24	DS/EN 1991-1-4 DK NA:2015
Sweden	Upplands-Väsby	24	(BFS 2013:10 EKS 9 2011)
Finland	South part	21	(NF EN 1991-1-4 1991)
Norway	Trondheim	26	(NS-EN 1991-1-4:2005/NA:2009)
Iceland	Reykjavik	24	(IST EN 1991-1-1:2002/NA:2010 2010)

Table 5.1-1: Values for the basic wind velocity for each Country

The second parameter that should be calculated is the mean wind velocity, $v_{,m}$ [eq. (4.4.1.2)] which depends on the terrain roughness, orography and the basic wind velocity.

The terrain category for all the cases has been considered as category IV (DS/EN 1991-1-4 2007, Table no. 4.3), as the given location is an area in which at least 15% of the surface is covered with buildings. However, this implementation can be easily changed for each location as the calculations have been also performed for category III.

Furthermore, the orography has been taken as 1 as the area where the project is located is considered as valley.

Lastly, the wind pressures are calculated according to [eq. (4.4.1.6)] which is determine by the peak velocity pressure, the reference height for the external pressure and the pressure coefficient for the external pressure. As the last two parameters are related to the shape and height of the element which is being analyzed, their values will change depending on the model. To be able to carry out the calculations the wind rose of the area has been analyzed, in order to determine which wind direction is the most unfavorable. Based on the report (Söderberg and Bergström 2008) about the climatological condition in Sweden and climate

data from *Meteoblue*¹² in Upplands Vasby, it has been estimated that the most unfavorable wind direction is from Southwest (Figure 5.1-1 left). As a simplification of how the buildings are oriented with respect to the North, it has been assumed that the most unfavorable wind direction is west direction (Wind 90º for considerations in the Eurocode) as shows the Figure 5.1-1 on the right.

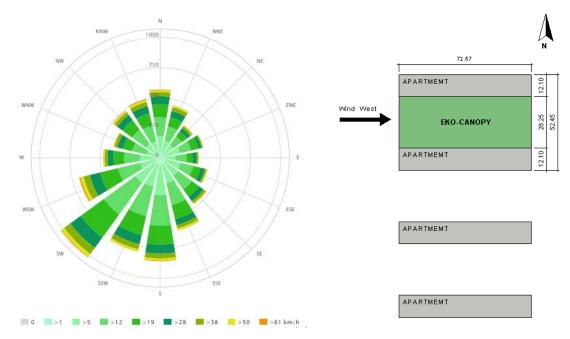


Figure 5.1-1: Wind rose for the area¹³ (left] and scheme of the are (right)

To calculate the coefficient pressures for the roof, two different assumptions have been made. For the first model (Case A), the roof is considered as a flat roof (Figure 5.1-2 left); while for the second model (Case B) the pressures on the roof are calculated for a monopitch roof (Figure 5.1-2 right).

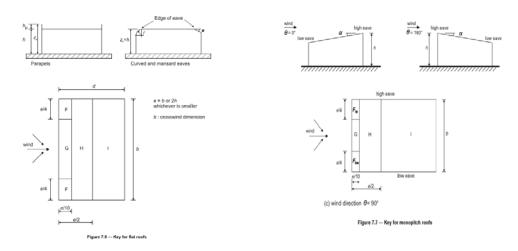


Figure 5.1-2 Flat roof wind areas (left) and monpith roof wind areas (right) a/EC1-4 (DS/EN 1991-1-4 2007)

¹³ https://www.meteoblue.com/en/weather/forecast/modelclimate/upplands-vasby_sweden_2666238

¹² https://www.meteoblue.com/en/weather/forecast/modelclimate

While in the case for the pressures on the walls of the Canopy, the premises specified in the figure 7.5 – Key for vertical walls (DS/EN 1991-1-4 2007) have been followed. The height is lower than the width so the following assumptions have been followed to estimate the wind profile acting on the facades.

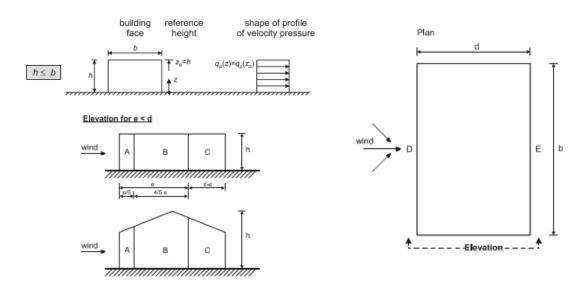
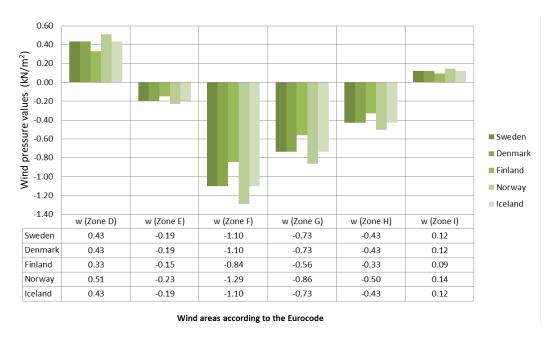


Figure 5.1-3: Diagrams from EC1-4 (DS/EN 1991-1-4 2007) to calculate pressure coefficient on the facades.

The following charter shows in a graphical way how the wind loads differs for the different countries for the flat roof case. It is seen, that according to the values from the Table 5.1-1, Norway has the most unfavorable values and Finland the most favorable. However there are not notable differences between the values.



Graph 5.1-1: Wind loads pressure values depending on the country. Flat roof case.

5.1.1.2 Snow Considerations

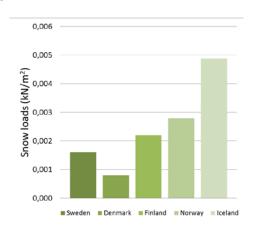
Regarding the calculation of the snow loads, the same factors specified for the wind loads will influence in its determination. The aspects that should be considered are already enumerated in 4.4.2 Snow loads calculations, and all of them, except the first one, the shape of the roof, depend on the climatological data and the area where the construction is located and its surrounding, therefore will be the same for all the models. The only factor that will change with the design is the shape coefficient, as depending on the slope of the roof (angle) the value of the load distribution will vary.

The design situation which will be adopted for the calculations is the persistent or transient situation [eq. (4.4.2.1)] and in order to calculate the snow load it is necessary to know the characteristic value of the snow load on the ground. In the same way that for the wind loads, this characteristic value depends on the location which is being taken as reference to perform the calculations. The following table listen the characteristic values for the five different countries which are being investigated.

Country	Chosen location	$s_k [kN/m^2]$	Reference
Denmark	Copenhagen	1	(DS/EN 1991-1-3 DK NA:2015 2011)
Sweden	Upplands-Väsby	2	(BFS 2013:10 EKS 9 2011)
Finland	South part	2.75	(SFS-EN 1991-1-3 NA 2003)
Norway	Trondheim	3.5	(NS-EN 1991-1-3:2005/NA:2009, n.d.)
Iceland	Reykjavik	6.5	(IST EN 1991-1-1:2002/NA:2010 2010)

Table 5.1-2: Characteristics values for the snow load on the ground for the different locations

In contrast to the wind loads, the snow loads have really extreme values for the regions of Iceland, Norway and Finland compare to Denmark o Sweden (Graph 5.1-2). This fact will have a huge influence in the design, as it is going to add a big amount of weight to consider for the calculation of the structure.



Graph 5.1-2: Snow loads for the different countries

Consequently, the snow in the Nordic regions should be deal with special attention, and the constructions should be built according to this. In the case of the EKO-Canopy project this issue

was investigated by Mikkel Knudsen in his investigation about the Canopy construction (Knudsen 2016), where he proposed to use EFTE instead of glass for the roof covering, as it has numerous advantages in comparison to glass regarding the management of snow. His proposal for the EKO-Canopy project includes the installation of diffusers on the roof, which will be used to melt the snow on the roof in the winter time. Therefore, this material as covering for the Canopy construction will be also investigated.

5.1.2 Load combinations

In order to perform the calculations it is necessary to carry out the corresponding load combinations, where the live loads and the permanent loads are considered. Its implementation in *Karamba* is a bit tricky, as the platform allows to the user to specify different load cases to later choose one to perform the calculations, but it does not consider the combinations according to the Eurocode, hence this should be implemented manually.

The capacities of the elements which form the structure should be checked for the Ultimate Limit States (ULS) based the premises of the *Eurocode 0* (DS-EN 1990 – Basis of structural design 2007) is specified in equation (5.1.2) and in the same way the deflection of the members should be check for the Serviceability Limit States (SLS) according to the equation (5.1.2).

$$q_d = \gamma_{Gj,sup} G_{Kj,sup} + \gamma_{Q,1} Q_{k,1} + \sum \psi_{Q,i} \gamma_{Q,i} Q_{k,i}$$
 (5.1)

$$q_k = G_{Ki,sup} + Q_{k,1} + \sum \psi_{O,i} Q_{k,i}$$
 (5.2)

Where:

- $\gamma_{Gi,sup}$ is design coefficient for unfavorable permanent actions taken as 1.15.
- $\gamma_{0.1}$ is design coefficient for lead variable action taken as 1.5.
- $\gamma_{0,i}$ is design coefficient for accompanying variable actions taken as 1.5.
- $\psi_{O,i}$ is the simultaneity factor taken from table A.1.1 (DS-EN 1990).
- $G_{Ki,sup}$ are the permanent actions.
- $Q_{k,1}$ is the lead variable action.
- $Q_{k,i}$ are the sum up of the accompanying variable actions.

To obtain the most unfavorable situation, five different load combinations have been implemented in Grasshopper using the VB programming component as follows:

- Load case 0 : Self-weight
- Load case 1: Self-weight + Live loads. This LC is subdivided in:
 - o Load case 1: Only wind
 - Load case 2: Only Snow
 - Load case 3: Wind Dominant + Snow accompanying.
 - Load case 4: Wind accompanying + Snow Dominant
 - Load case 5: Wind + Snow (Characteristics values ULS)

This loads combinations correspond with the following table with values listened as a generic way (Not specific values considered). The code implementation could be seen in APPENDIX F: SCRIPT IN GRASSHOPPER DEFINITION.

		Dead loa	Dead loads - Permanent actions			Imposed loads - Accompanying variable actions						Result. design
Nº Load case		Selfweigth - structure		Selfwei othe	•	Wind load - WL		/L	Snow load - SL			value
		Gk,str (kn/m2)	Υ _{Gj}	Gk,oth (kn/m2)	γ _{Gj}	Qk (kN/m2)	Ψ _{Qi}	γ _{Qi}	Qk (kN/m2)	ψ _{Qi}	γ _{Qi}	q _d [kN/m2]
	Ultimate limit states (ULS)											
0	DL (Self weight)	1.00	1.15	0	1.15	1.00	1	0	1.00	0	0	1.15
1	DL+WL	1.00	1.15	0	1.15	1.00	1	1.5	1.00	0	0	2.65
2	DL+SL	1.00	1.15	0	1.15	1.00	0	0	1.00	1	1.5	2.65
3	DL + WL (Dominant) + SL (acomp)	1.00	1.15	0	1.15	1.00	1	1.5	1.00	0	1.5	2.65
4	DL + SL (Dominant) + WL (acomp)	1.00	1.15	0	1.15	1.00	0	1.5	1.00	1	1.5	2.65
	Service limit states (SLS)											
5	DL + WL + SL	1.00		0		1	1	1	1	1	1	2.00

Table 5.1-3: Load combinations

5.2 Structural analysis of the different shape configurations

The main objective is to reach a point where the practitioner can take the decision of the optimal solution based on several factors. Normally when we are dealing with design there is not a unique best option, there are different better options depending on the aspect which is considered primordial.

Nevertheless, in all design process one of the main goals is always to achieve a sustainable construction. In relation with the structure, this objective is associated to the type of material and the total amount of mass (kg) necessaries for its construction, as these two parameters will determine the LCA calculations.

For instance, when we are talking about optimal solutions we are talking about efficiency; and the efficiency of the structure will be achieved when an equilibrium between the total amount of mass and the utilization ratio of the members which form it coexists.

As it was explained in *Section 4.3* the optimization process in done using the component Galapagos of Grasshopper in which the genes represent the different parameters to study. These have to reach a target optimization value, called fitness, which depending on the optimization process which is being carried out will vary: mass of the structure, minimal deflection, etc.

In the same way, the parameters for the optimizations processes will vary depending on the model which is being studied. Each of them should be delimited into a domain of different possibilities which represent the landscape of variations. These aspects are developed further on for the specific cases.

Therefore, in order to perform the optimization process it is necessary to previously define:

1) The characteristics of the different geometries, which set the landscape of possibilities.

- 2) The loads which are going to be applied to each specific model.
- 3) Set up the support conditions for the structures.
- 4) Describe the input to carry out the optimization process.

This is going to be explained in the following section for each of the different models. The reason of repeating the same procedure several times is that the optimization process is a complex practice which requires a deep analysis. In order to establish a simple method which can be used as the basis to create and analyze complex structures, the first approach has been to model a basic geometry between the two existing buildings based on a simple structure form by beams and columns.

Consequently, once that the optimization process has been settled down, a more complex structure has been analyzed. This second model is based on the design proposal that white architects presented, going again from a basic configuration until reach the complexity.

5.2.1 Case A: Basic Frame Configuration

5.2.1.1 Design properties (geometry):

Before analyzing complex geometries for the ECO-Canopy construction is important to study how the single elements of the structure will work against the loads acting on it. Therefore the first approach which has been considered is a simple construction, modeled as a box form by different frames in X and Y direction (Figure 5.2-1). Depending on the case, the elements will be included in the global structure or not, taking special attention to the column buckling and lateral torsional buckling of the elements as the span in some of the iterations could be considerable big.

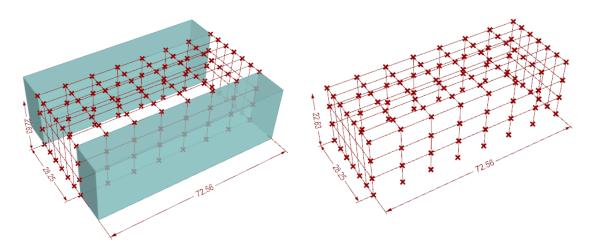


Figure 5.2-1: Case A – Parametric geometry

As it seen, the structure is placed in the space in between (measurements Table 4.4-1) so the frame will be placed in a box of maximum measurements of 28.25x72.56x22.63 m.

Thanks to the parametrical modeling is really easy to change this box domain. So, the first input of the modeler should be set these maximum measurements. Regarding the type of elements, as the loads are transferred to the ground through beams and columns, no plate elements or shells are included in the design, which simplifies the calculations.

5.2.1.2 Loads and materials:

The script developed with grasshopper allows to the user to choose between different structural materials, load cases and locations. The first study is focused on the implementation of the optimization process (form-finding) and its results. Hence a specific case has been chosen, being the parameters for the first optimization process as follows:

Location: Sweden

Structural material: Steel 275s

Cover material: Glass (Including self-weight of the auxiliary profiles)

Load case: 4 (ULS) 5 (SLS)

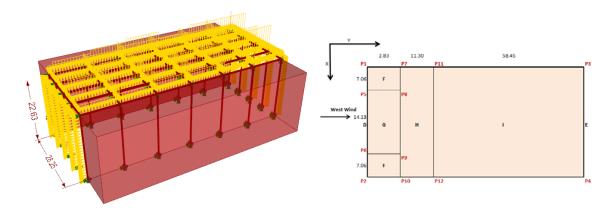


Table 5.2-1: Loads applied to the geometry (left) and roof load areas (right) case A

5.2.1.3 Support conditions

To perform the calculations it is necessary set the support conditions for the model. The connection of the columns to the foundation has been set as a totally fix supports. This means that the three translations, on the X axis, the Y axis and the Z axis, have been restrained; as well as the three rotations around the these axes, which leads to a cero DOF for all the supports at the base. This input can be seen on the APPENDIX F: SCRIPT IN GRASSHOPPER DEFINITION, where the support-component from Karamba is used.

On the other hand, the support conditions for the joints between the different elements of the structures have been set as hinge connections (as a general simplification) where the DOF for the translations in the X axis, Y axis and Z axis have been set to cero, allowing the rotations around the three axis. This input can be seen in the APPENDIX F: SCRIPT IN GRASSHOPPER DEFINITION.

5.2.1.4 Results: Structural optimization

There are three different considerations that should be taken into account when the optimization process is carried out, which are firstly the parameters to optimized (genes); secondly, the goals to achieve (fitness); and thirdly the verifications that should be fulfilled when the process is being developed (maximum stress, resistances of the elements, deflection and buckling).

Hence, for this specific model case the parameters to optimize would be:

- Number of frames in X direction.
- Number of frames in Y direction.
- Cross sections of the different elements which form the structure:
 - o Cross sections beams on top surface X direction
 - o Cross sections beams on top surface Y direction
 - o Cross sections beams on both lateral sides, left and right, Y direction
 - o Cross sections beams on front and back surfaces, X direction
 - Cross sections columns front and back
 - o Cross sections columns both lateral sides, left and right.

Following the process specified in 4.3 first all the cross sections for the frames have been chosen following the structural verifications (4.4.3) and checking the maximum deflection using the utilization model to avoid the failure of the structure during the first optimization process.

After this, the first loop of the optimization process (*Seventh step: Topology optimization 1*. Perform the 1st optimization process using Galapagos component for GH., *Eight step: Cross-section optimization*. Perform the 2nd optimization process using the component for Karamba, "Cross-section optimization", and *Ninth step: Topology optimization 2*. Perform the 3rd optimization process using Galapagos based on the results from the 8th step – cross-cross section optimization.) is carried out, where the first goal is to achieve the minimal maximum deflection of the model and the second goals is to minimize the total amount of mass (Fitness values). Meanwhile, the genes of the optimization are the number of frames in X direction and in Y direction (APPENDIX F: SCRIPT IN GRASSHOPPER DEFINITION)

In order to achieve a feasible solution it has been limited the possible total number of frames for each direction, being the domains:

- Frames in X direction [3,8]
- Frames in Y direction [3,8]

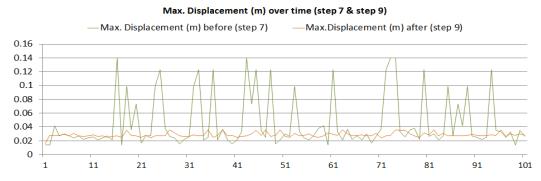
This domain reduces the variables to 36, which means that the combination of 4 frames in both directions (x=3, y=3) will have 64 elements, and the combination 9 frames in both directions (x=8, y=8) will have a total number of 244 elements.

The maximum deflection over time (number of iterations) for the steps 7 and 9 for the first optimization loop is shown in the Graph 5.2-1, where the process should try to reach a converge solution.



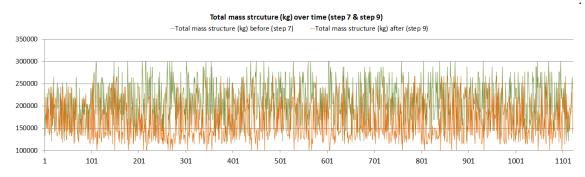
Graph 5.2-1: Maximum displacement [m] over time for the steps 7 (green) and step 9 (orange)

However it is observed that the values oscillate around the same values, between [0.01; 0.14] for the step 7 and between [0.02; 0.04] for the step 9, which means there is not an optimal minimum value, if not that there might be multiple optimum values. The repetition of the pattern is a sign of this, where several minimum values are reached. The Graph 5.2-2 shows a zoom in Graph 5.2-1 for the first 100 iterations. The reason of the short domain for the second topology optimization is than the deflection was limited to 0.03 m to perform step 8, Cross-section optimization. This process has been repeated several times (10) in order to verify that the results were correct.

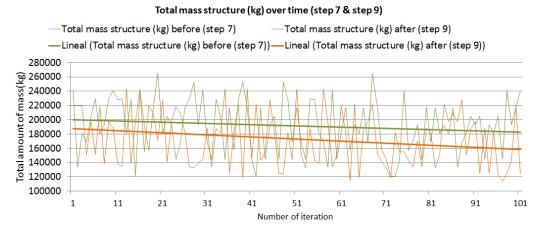


Graph 5.2-2 Zoom in first 100 iterations of maximum displacement [m] for the steps 7 (green) and step 9 (orange)

Additionally, on the Graph 5.2-3 has been plotted the total mas of the structure for the steps 7 (green) and 9 (orange). It is verified that the values of the second topology optimization where the cross-section of the elements have been already optimized, tends to have less amount of mass in comparison with the first topology optimization. This is clearly seen on the Graph 5.2-4 for the 100 first iteration, where it is plotted the tendency line for each case.



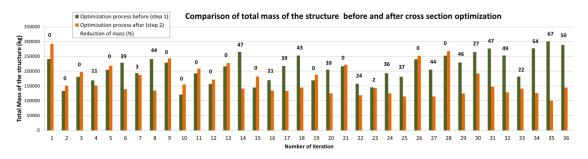
Graph 5.2-3: Total amount of mass [kg] over time for the steps 7 (green) and step 9 (orange)



Graph 5.2-4: Zoom in first 100 iterations of total amount of mass [kg] for the steps 7 (green) and step 9 (orange)

5.2.1.5 *Post-process – Quality check*

Finished the first loop of the optimization process, the results are analyzed in *Karamba* and exported to excel (*Eleventh step: Analysis of the configurations*. Extract the different combinations, analyse the optimal ones based on the amount of mass and the utilization ratio of the cross sections. Check that the elements of the structure fulfil the requirements specified in the Eurocode.) in order to verify that the elements accomplish the requirements of the Eurocode. The Graph 5.2-5 shows the comparison of mass for the 36 different structure variables (Table 5.2-2) of the model, where is indicated the percentage of saved material in each combination, being the maximum reduction of mass 67% respecting the first model, which is produced in the case number 35. (APPENDIX F: SCRIPT IN GRASSHOPPER DEFINITION)



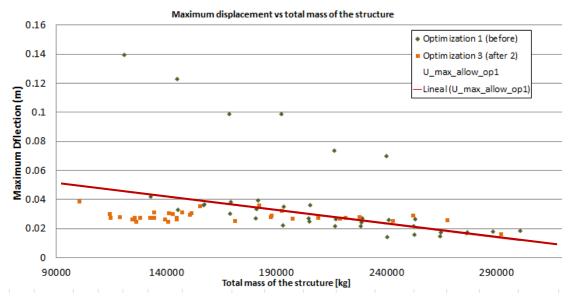
Graph 5.2-5: Comparison of total mass for the 36 different iterations

						-			
Nº it	Χ	Υ	Nº it	Χ	Υ	_	Nº it	Χ	Υ
1	7	5	13	7	3		25	4	6
2	3	4	14	7	7		26	8	3
3	5	4	15	4	3		27	4	8
4	4	5	16	3	7		28	8	4
5	6	4	17	5	7		29	5	8
6	6	6	18	7	6		30	8	5
7	5	5	19	5	3		31	8	6
8	6	7	20	5	6		32	6	8
9	7	4	21	6	5		33	3	8
10	3	3	22	3	6		34	7	8
11	6	3	23	3	5		35	8	8
12	4	4	24	4	7		36	8	7

Table 5.2-2: Combination of frames X and Y direction for the 36 variables

However, it is seen that in some of the combinations the amount of material is bigger in the second optimization than in the first one, so these arrangements have been neglected.

The Graph 5.2-6 shows the results for the 36 variables (step 7 – green and step 9 –orange), plotting the total amount of mass of the structure and the maximum deflection. As the second topology optimization (step 9) is done after the cross section optimization where it is limited the maximum deflection, it is seen that most of the values are under the red line (allowed deflection). However, when the bearing capacity of the biggest cross section of the set of HEB profiles is smaller than the minimum load bearing capacity needed, this limit is overpassed as the configuration necessary is not accepted.



Graph 5.2-6: Maximum deflection and total mass for the 36 variables

Lastly, in order to analyze which configurations are the most optimal, it has been plotted the utilization ratio of the cross sections over the total amount of mass used. The utilization ratio of the cross section indicates the percentage of the cross section which is fully working, in this way a percentage of 100% means that the bearing capacity of the element is totally reached, while a percentage of less than 50% indicates that the element is oversized, as half of the cross

section is not being utilized. Being conservatives the percentage should in between 60 % and 90% to avoid the failure of the structure.

Consequently the optimal solutions would be the ones which have a balance between the total amount of mass and the utilization ratio. The Graph 5.2-7 shows this relation for the iterations with utilization ratio less than 100%. The iterations inside the red circle are the configurations of the structure which have better performance.

Utilization ratio vs total mass of the structure

120 Utilization ratio Cross section (%) 100 80 60 Optimization 1 40 before (step 7) Optimization 3 20 after (step 9) O 100000 150000 200000 250000 300000 350000 Total mass of the strcuture [kg]

Graph 5.2-7: Utilization ratio versus total mass of the structure.

The iteration number 25 could be considered the optimal one, which has 5 frames in X direction and (x=4) and 7 frames in Y direction (y=6). This leads to a total of 106 elements out of 244 possible elements for the whole structure. The cross sections for each of the elements are:

- Beams on top surface X direction: HEB 180, HEB 220 and HEB 160
- Beams on top surface Y direction middle: HEB 300, HEB 550, HEB 450, HEB 400
- Beams on top surface Y direction sides left and right: HEB 300
- Columns front and back: HEB 700, HEB 340, HEB 300, HEB 280, H6EB 280, HEB 240
- Columns both lateral sides, left and right: HEB 140, 160

This configuration reduces in 42 % the amount of mass respect the first one. The main reason is that the initial cross sections were estimated to fulfill the requirements when the model had the minimal amount of elements (64 elem.), so when this number is potentially increased the cross sections of the models result oversized.

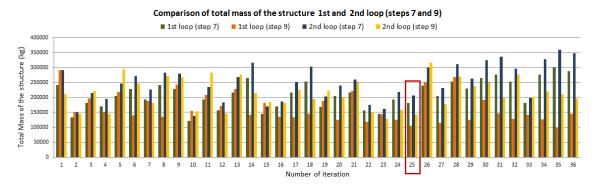
It should be taken into account that *Karamba* does not understand about construability, it might happen that a lower column has a smaller cross section than the one above or the same type of element has five different cross section; therefore all the connections between the elements which form the structure, as well as their orientation should be cautiously check after the first process of optimization.

The values for each of the process are listened in the the Appendix F. After the first optimization loop, it is carried out the *Eleventh step* where it could be checked that the cross sections suggested by *Karamba* didn't fulfill the requirements of the Eurocode for lateral buckling resistance (HEB 140), so a new iteration process was done were the cross section for the columns were set to remain the same for all the elements, (Step 11-7). Also, the orientation of the columns in the front and back façade are being rotated, as the main frames are in Y direction with this combination, so they should be placed with the strong axis perpendicular to this direction (web of the profile in Y direction).

Based on this a new optimization process incorporating the new changes has been done. The results are compared in Table 5.2-3, where it is seen that this modified configuration has decreased in a 21% the amount of material used respect the initial structure. As it was expected the reduction of mass is now less than in the previous case, as it has been included construability considerations which make to increase the size of some of the cross sections. The results for the two loops of optimization process (step 7 and 9 for each) are plotted in Graph 5.2-8.

1 st (ptimization lo	оор	2 nd optimization loop			
Step 7	Step 9	Reduction	Step 11-7	Step 11-9	Final reduction	
Total mass [kg]	Total mass [kg]	S7 / S9	Total mass [kg]	Total mass [kg]	S7 / S11-9	
	[vg]	[%]	[vg]	[vg]	[%]	
181326.04	105619.93	42	206757.76	142844.88	21%	

Table 5.2-3: Characteristics values for the snow load on the ground for the different locations



Graph 5.2-8: Comparison of total mass of the structure 1st and 2nd loop

Consequently the final configuration has the following characteristics:

- Beams on top surface X direction: HEB 180, HEB 160, HEB 140
- Beams on top surface Y direction middle: HEB 360, HEB 320, HEB 300
- Beams on top surface Y direction sides left and right: HEB 240, HEB 260
- Columns front and back: HEB 340 (set as uniform cross section)
- Columns both lateral sides, left and right: HEB 450 (set as uniform cross section)

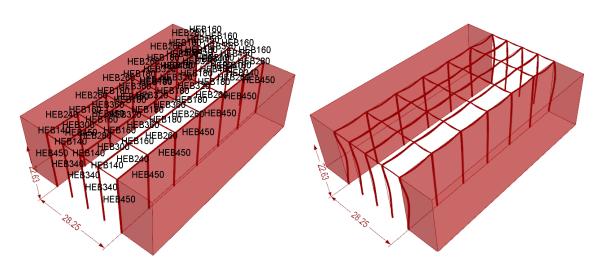


Figure 5.2-2 Final configuration. Cross section values (left) deformation of the structure (right)

The stresses and the bearing capacities for the elements, as well as the maximum deflection are checked to verify that fulfill with requirements of the Eurocode. (APPENDIX E: OUTPUT FROM GRASSHOPPER. SECTION FORCES AND VERIFICATIONS)

The parametrical modeling allows to test different materials, different class of Steel, different cover materials (EFTE) and different cross section profiles. The results for the different cases have been investigated but they are not relevant for the explanation of the method.

Lastly, it has been studied what would happen if the same construction would be located in a different country, to see the influence of the wind and snow loads on the structure.

5.2.2 Case B: White Architects configuration - From Basic to Complex Structure

The proposal of White Architects is shown in (Figure 2.2-5: Render inside Canopy (left) and scheme of overall concept of the Canopy(right) (White 2015), consisting in a combination of plate – shell surfaces and boundary frames. In order to be able to carry the loads from the top surface to the ground and reduce the loads transferred by the boundary frame, two internal columns, shaped as a 'tree', have been designed. This model would be the last approach, as it requires more complex calculations.

Accordingly, three different configurations for the same box domain (Figure 5.2-3) have been done in order to evaluate at the end the aesthetics values and efficiency, as en equilibrium should exists. The starting point consists in a simple frame construction (frames in X and Y direction) based on the Case A. The second case consists in a more complex configuration where plate surfaces have been modeled in between the boundary frames. Lastly, the third model tries to reproduce the initial proposal, having the two tree columns in the middle.

Therefore, based on the same process applied in case A, it is going to be study the EKO-canopy going from simple to complex structures following the scheme in Figure 4.3-1. In this way, this part only shows the results relevant for the investigation and they are not going to be explained with the same degree of detail than from the previous case. It is assumed that the process is already verified and settled down.

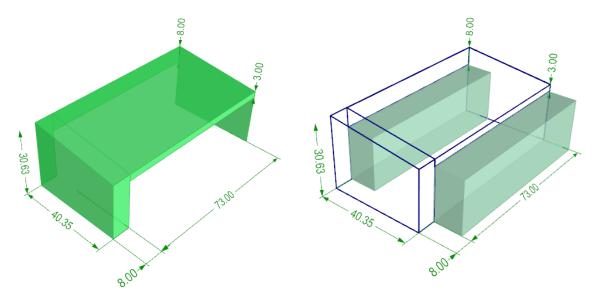


Figure 5.2-3: Box domain for case B

The geometry properties for the box domain of the three cases are based on the Table 4.4-1, to which should be added:

Parameter	Unit [m]	Measurement
Extension length front	m	8
Extension roof (high eave)	m	8
Extension roof (low eave)	m	3
Slope of the roof	[º]	7.06

Table 5.2-4: Basic measurements of the site for the Canopy construction

As all the process have been detailed described for case A, for this second case only the differences regarding the input or the definition of the script would me mentioned.

Case B - 1: Basic frame construction

5.2.2.1 Design properties:

Regarding the geometry properties, this configuration is really similar to previous one (Case A). It consists in a frame-type construction form by Beams and columns. After a first analysis, it was necessary to include transversal beams in the front and back facade to reduce the buckling length of the columns to the half.

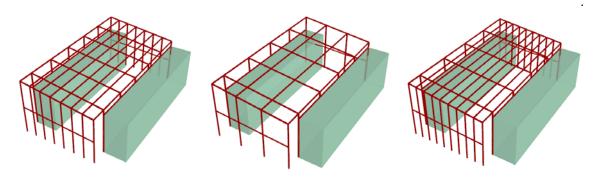


Figure 5.2-4: Possible geometries case B-1(only shown the main load bearing structure, not the covering)

5.2.2.2 Loads:

The loads for this second case are calculated for a monotpitch roof (See Appendix XX) and are applied in the different load areas as it is shown below. Karamba project the surface load in the corresponding line load in each of the elements.

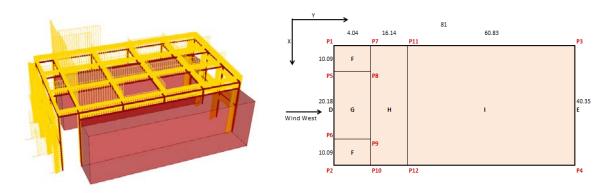


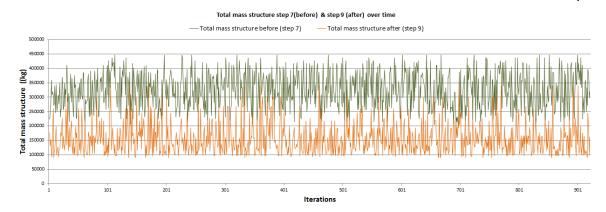
Figure 5.2-5: Loads applied to the geometry (left) and roof load areas (right) case B - 1

5.2.2.3 Support conditions

The connection of the columns to the foundation is defined as fully fix support, while the connections between the different elements are defined as hinge connections.

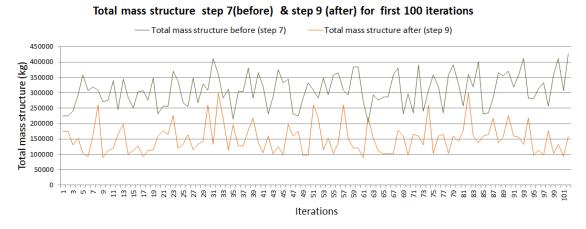
5.2.2.4 Results: Structural optimization

Regarding the optimization process, the main difference is that directly in this second model the limitation of the deflection has been input in a dynamic way. This means that for each combination of frames that Galapagos takes, the maximum allowed deflection is calculated based on the span of the elements, providing more feasible results. In this way, according to the Graph 5.2-3, the difference between the total mass of the structure before and after the cross section optimization is bigger than for the case A.



Graph 5.2-9: Total amount of mass [kg] over time for the steps 7 (green) and step 9 (orange)case B - 1

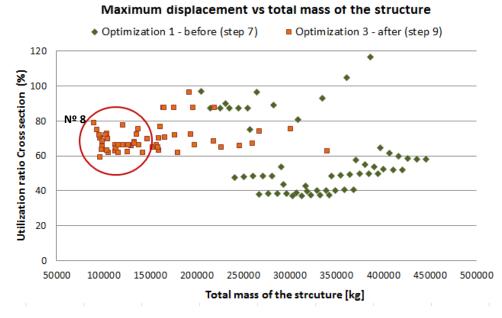
Furthermore, it can be seen that both iterations follow the same pattern (Graph 5.2-4), which indicates a better implementation of the method. Lastly, it is confirmed again that there is not an optimal solution if not several minimums, as there is not convergence in the results.



Graph 5.2-10: Zoom in first 100 iterations of total amount of mass [kg] for the steps 7 (green) and step 9 (orange)

5.2.2.5 *Post-process – Quality check*

This structure shows a bigger difference between the first optimization process (step 7) and the second optimization (step 9). In this way, according to the Graph 5.2-11, it is clearly seen than the optimal designs (inside the red circle) correspond to the iteration Nº8 (step 9).



Graph 5.2-11: Utilization ratio vs total mass of the structure.

In this case, this combination corresponds with the structure which has less amount of mass (89979.64 kg) having X equal to 4 and Y equal to 8. Furthermore the structure is working at 79% of its capacity in terms of maximum deflection. It saves 67% of total amount of mass respect the first model.

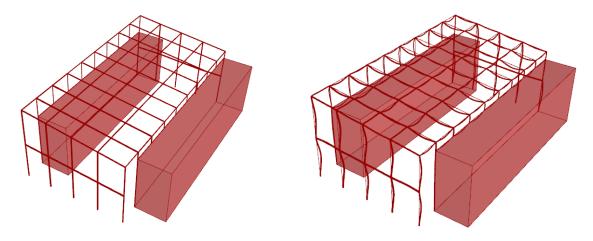


Figure 5.2-6: Configuration iteration № 8 (X=4, Y=8)

Case B - 2: Plate & Frame structure

5.2.2.6 Design properties:

The idea of this third model, second option for model B, is to move a bit further in terms of design and originality. Based on the same configuration presented before, this includes a grid between the main frames which form the structure creating a mesh geometry.

5.2.2.7 *Loads*:

The loads are applied in the same way than from the previous case, as the roof and the façade have the same geometry, so the load surfaces are calculated in the same way. Thanks to the mesh load plug-in of Karamba then these surface load are changing parametrically, and they pressures are distributing on the elements depending on the iteration which is being calculated.

5.2.2.8 Support conditions

The support conditions are defined in the same way than in the previous case, as this is a transformation of the previous one. However a new implementation has been necessary. As now there are created plate grid surfaces between the elements, it has been set all the points on the boundaries which connect the plate surfaces to the main frames as simple supports. These points would be different depending on the size of the mesh (grid) and depending also on the number of principal frames which has the structure, so to make it work it has been necessary a more complex definition. Below is shown the structure with the support definition, in the first one only the supports between the different elements which form the main frames and the connection to the foundation are shown. The picture on the right shows also the supports on the boundaries of the plates.

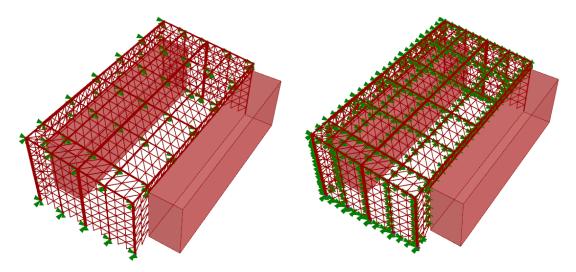
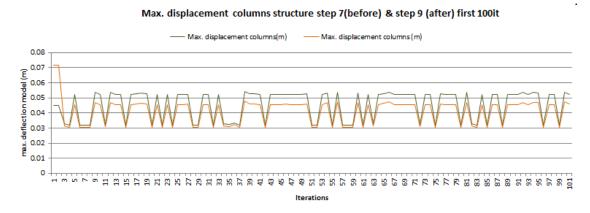


Figure 5.2-7: Support definition - beams and columns (left) and support definition beams, columns and plates (right)

5.2.2.9 Results: Structural optimization

This model has three different types of elements: beams, columns and plates. As the maximum allowed deflection is different depending on the element, it has been divided the cross section optimization in the script in this three categories, which mean more accurate results. This is verified in the Graph 5.2-12 where the maximum deflection before and after the cross-section optimization follows exactly the same pattern.



Graph 5.2-12: Maximum displacement columns case B-2

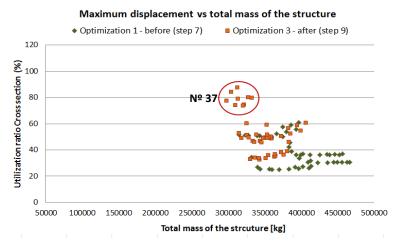
Furthermore, it is seen that this configuration tends to use much more amount of material than the previous one (Graph 5.2-13)



Graph 5.2-13: Total mass structure case 2-B

5.2.2.10 Post-process - Quality check

The results after the optimization process are analyzed, and according to the Graph 5.2-14, it is clear which options would be the optimal ones (inside the red circle).



Graph 5.2-14: Utilization of Cross-section vs Mass of the Total Structure

The combination which has less amount of mass is the number 37, having 6 frames in Y direction (X = 5) and 10 frames, taking into account the extension, in Y direction (Y = 8).

The deflections and the stresses for the elements are checked automatically in Karamba as a first overview simplification to have an indication if the members which form the structure will resist. To carry out a deep post - analysis the results are exported to excel (Step 11) (APPENDIX D: RESULTS OPTIMIZATION PROCESSES). The final configuration for the iteration number 37 is shown below.

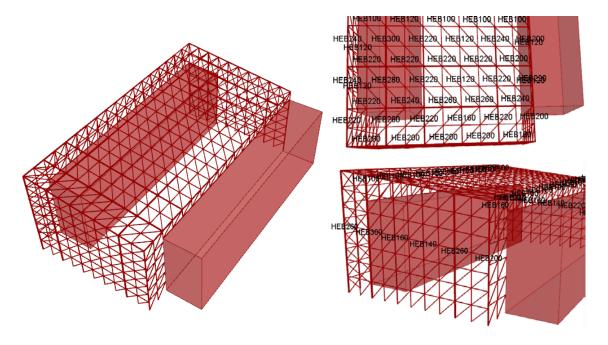
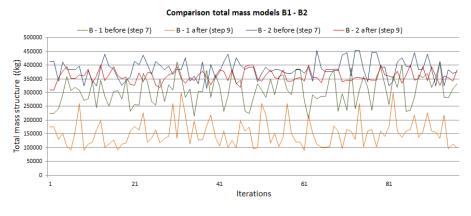


Figure 5.2-8: Final Geometry for case B - 2

The results for the optimizations processes (step 7 and step 9) for the model the model B-2 are analysed and compare with the case B-1 (Graph 5.2-15). This shows that the second model, which has incorporated the mesh grid, needs to use more amount of material. It is also observed that the difference between the first and the second topology optimization is not as big as before.



Graph 5.2-15: Comparison Total mass of structure for case B-1 and B-2

In this way, the Table 5.2-5 indicates that the second structure, B -2 would need to used 70% of steel more than the previous case, which means more impact on the environment regarding the LCA process.

Case B - 1			C	Case B - 2			
Step 7	Step 9	Reduction	Step 11-7	Step 11-9	Reduction	Increase of mass	
Total mass [kg]	Total mass [kg]	S7 / S9 [%]	Total mass [kg]	Total mass [kg]	S7 / S11-9 [%]	S9 (B2)/S9 (B1) [%]	
270,940.49	89,979.64	67	409,466.37	298,337.56	27	70	

Table 5.2-5: Comparison structures B -1 and B - 2

Case B - 3: White Architects proposal

5.2.2.11 Design properties:

Finally the third model for case 3, reproduces a similar configuration of the proposal of White Architects (Figure 2.2-5). It is a combination of beams, columns and shell surfaces. The main loads of the structure would be carried by the boundary frames and by the two columns placed in the middle of the space.

This model represents the one which has more aesthetical value, as it plays with geometry and forms reproducing an original configuration.

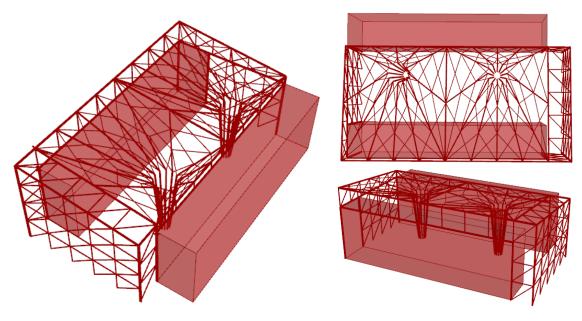


Figure 5.2-9 Geometry for case B - 3

5.2.2.12 Loads

The loads for this model are applied in the same surfaces areas described for the previous models, as the external geometry has the same measurements.

5.2.2.13 Support conditions

Regarding the support conditions, this configuration follows in general terms the same principle than the previous case. In this case the main frames are the boundary frames on the sides and the radial beams which go from the sides to the middle tree columns. In between this radial beams is created a mesh grid.

The connections of the columns, both the ones in the middles and the columns which form the boundary frames, are fully fix supported at the base. While the connections between the elements which form the main load-bearing structure are defined as hinge supports, allowing the rotation around the three axis.

5.2.2.14 Structural optimization

This model has required a deep analysis to create the geometry. As it is form by shell surfaces which are not the focus of this report and required deep analysis and new incorporations to the script, this model has been only calculated in general terms to be able to compare the total amount of mass that would require with the other two options. However, it has not investigated in deep every element which forms the structure. Therefore not topological optimization processes have been carried out for this third model. However, the geometry has been created in a way that this implementation is possible. For example, the parameters which could be investigated are the number of divisions on the sides which are connected to the number of divisions of the central columns. Additionally, the diameter of the columns is also variable.

For the main load bearing elements, beams and columns it has been implemented the cross-section optimization component, giving a save of material of 17% respecting the preliminary specified geometry.

5.2.2.15 *Post-process – Quality check*

The following table shows the total mass necessary for the construction of the three different models, B - 1, B - 2 and B - 3.

Mode	el B - 1	Mode	el B - 2	Model B - 3		
Step 7	Step 9	Step 11-7	Step 11-9	Step 11-7	Step 11-9	
T. mass [kg]						
270,940.49	89,979.64	409,466.37	298,337.56	312943.98	260569.96	

Table 5.2-6: Comparison of the total amount of mass for the three models.

Part VI: DISCUSSION

Through the developing of this master thesis project it has been studied the idea of integrating structural verifications in the early design phases of the building practices, including a method of form-finding geometries based on optimization processes. This has been done by means of parametrical modeling with the definition of a script in Grasshopper which can be interpreted as designing tool. Consequently, the purpose of this sixth part is to analyze the findings to discuss the advantages, as well as disadvantages or limitations of the applied method and the created script.

6.1 Integrated method proposal: advantages and disadvantages

Since the beginning it has been discussed the idea of replace the traditional working flow of the building process for a dynamic workflow approach. As it has been stated in Part III, this has been already happening since the last decade with the idea of Building information modeling based on the concept of the total collaboration between the building agents and the model data exchange through the different software. However, the integrated dynamic model moves a step forward, not only considering the idea of transfer the model information if not that it proposes a combination of aesthetical and technical aspects using the same tool, or a system where the data is transferred dynamically between design and calculation tools.

This has been deeply studied using the parametrical tool Grasshopper as a design platform which thanks to the incorporation of the FEM plug-in *Karamba* has allowed to perform structural calculations using the same middleware.

As it has been seen, this has provided great flexibility in terms of design, at the same time than a technical feedback is automatically provided. In this way, several structural configurations have been modeled and analyzed for the same construction, which with the traditional working approach would have been really time consuming. Within this context it should be distinguished two different things about the geometry definition. For the developing of this thesis it has been created three different models (case A, case B-1/B-2 and case B-3), and inside each of them it has been generated different structural configurations. Normally, in the building design process only one model would be created and all the alternatives would be generated based on that one. The reason of creating different models is to study the possibility of developing a script in grasshopper which can be used for several projects, in which the most of the features are already pre-defined. In this way, this tried to answer to the second research question of this thesis.

Accordingly based on the followed process, it can be concluded that it is possible to pre-define a script, which is divided in different sections; some of them can be made in a generic way and be used in several models, while others must be specific for the single projects. The different sections are:

- Calculation and definition of loads.
- Geometry definition
- Structural definition (elements, supports and cross-sections) (from GH to Karamba).
- Analysis of the structure.
- Optimization process input.
- Verifications (stresses and deflection).
- Output: Section forces and deflection.

All of them, except the geometry, could be interpreted as generic using the defined script and making the necessaries adaptations in each case. The creation of the geometry is of course specific for each case. It should be noted that special considerations should be made during its creation, taking special attention to small details. In this way to later transform the model to Karamba all the elements and connections should be modeled cautiously. This means that each line should be modeled as a single element, and all the points which connect those lines should be perfectly identified to be able to later define them as moveable, hinge or fix connections (support definition). This fact adds complexity to the designs, and it requires for its development a practitioner who has both creativity and technical skills. Therefore, it appears in this scenario the figure of the hybrid practitioner, who would be a key piece in this integrated dynamic model proposal.

Regarding the flow process between the different platforms where the data is transferred, it has been seen that it is working correctly. Once that the different ways of transferring the data have been already specified and settled down it would be really easy for an external person follow the same procedure, as all the connections have been already tested.

6.2 Results and optimization process: benefits, limitations and its future improvements

Regarding the optimization techniques applied, it has been seen that they represent a feasible method which can be easily implemented in the design process at the beginning to help to the practitioner to choose the best option for his design based on a certain number of parameters.

In this context it is clear the idea that not an unique optimal option exits. The results from the topological optimization processes carried out for the EKO-Canopy project are a sign of this, where several minimums were found for the same construction. However based on these results it is possible to do an evaluation of several variations for the same structure. Comparing the total amount of mass used for the construction and the utilization ratio of the elements which form it, it is possible to know the performance of the structure. This allows to the designer to make the corresponding decisions since the beginning to create a model which at the end, it would be possible to build without the necessity of doing several changes.

Furthermore, it has been also checked through the creation of the different models that is really important an accurate definition, as based on this, the optimization process would provide more or less worthily and trustily results. When we are dealing with the idea of calculating the structure at the same time that it is being modeled, every element which forms it should be perfectly defined. Not only with the action of generate the model is enough, if not that the modeler should move a step forward and think if that specific line which is drawing would be a beam, a column, a tension cable, etc.

Regarding the optimization process carried out for the specific model, the EKO-Canopy, it has been possible to estimate which structures would be optimal based on their efficiency after carried out several iterations, as it was summarized through the chapter. In this way it is possible to reach until 70% of savings in material respect the first configurations. However there are several factors which affect to the results of the optimization, as they are the level of definition of the joints (eccentricities), the orientation of the elements to use the maximum of its capacity, etc. Thus the process can be done one or two times, or based on the results it can be repeated multiple times until reach "the perfect solution". This also proves that increasing the level of accuracy, decreases the amount of saved material, as was demonstrated for the model A. In that process after the first loop of optimization the percentage of saved material was almost to the half, 42%, while after the new changes, the amount of saved material was 21% respect the initial configuration.

Part VII: CONCLUSION

This seventh and last part of this thesis work concludes and summarizes the main points deals during this project.

The first thing studied was the idea of substitute the linear working approach for an integrated dynamic model. Within this aspect this thesis has showed the potential benefits that this method could provide to the construction industry. However, there are still so many features for discover and study, as one of the main things that can be learned through this work, is that doing this in the frame of parametrical modeling provides endless possibilities. Grasshopper, which is the main tool used, is based on Visual Programing Language which allows to the user to incorporate under the same software several features including of codes by means of programming. Every day there are developed new plug-ins or components that allows to the user from performing energy analysis to carry out FEM calculations, as it was this specific case. Additionally, as all is based in basic geometry (lines, points, surfaces, etc) leads to the designer to create from the most simple configuration, as it was Case A, to more complex designs, as it was case B-3. Sometimes these designs could be really complicated as one of the goals of the architect is to look for original concepts. In this way the incorporation of structural aspects since the beginning would help to achieve the most original and creative geometries but being at the same time reliable.

The integration of the structural verifications in grasshopper allows to the architect to play with geometries, and to the engineering to calculate them, incorporating in this way its knowledge and suggestions from the beginning. This makes gain time during the construction

process and makes a more detailed and well-defined conceptual designs. This also demonstrates the necessity of incorporate to the building process the hybrid practitioner.

Furthermore, regarding the optimization techniques integrated in Grasshopper, it has been seen that on the one side they are easy tool to use for non-programmers as they are not based in difficult mathematical problems; and on the other side, the optimization process itself based on form-finding and sizing methods provides important savings of materials for the overall structure if it is implemented in a successful way.

This objective deals with one of the main goals of this master thesis, which was to establish a method to look for sustainable constructions in the Nordic Region. This could be used in a generic way for several projects as the script created is a kind of receipt which is modified depending on the desires of the architect. For example thanks to the loads definition on the script, it is possible to take the climatological data for several regions, calculating the same construction in different locations along the Nordic Region.

In relation with the EKO-Canopy project, the thesis provides some indications which establish the basis to let the client or the designer to choose. It has been seen that the most effective and sustainable solution is the most basic one, as it used less amount of material, which at the end benefits to the LCA process. However, in all design process, the creativity and originality are key issues, so at the end should be a balance between aesthetics and efficiency.

Lastly it can be concluded that the script presented provides some small tools which can be applied to several projects. It establishes the basis relations of components of how to define the loads, how to calculate the deflection for single elements and based on that carry out optimization processes, how to know the section forces for the different elements, etc. Therefore, when a new project is started this small parts which form the big script can be joined together to be able to develop the whole process. However, as the parametrical modeling including *Karamba* is a world itself, it should be specified which level of detail for the calculations is needed or wanted to achieve.

6.3 Further studies

As further studies for this work, it could be interesting the possibility of incorporate other optimization process including other type of structural material, as for example wood. This implies that other verifications should be added to the script. If this point is achieved, it would be really beneficial for the designs to connect in last instance the method to the already created scripts for the calculation of LCA, as it was stated in Part IV.

Additionally, it would be really interesting to continue the works in more complicated geometries, because at the end the new architecture consists in fancy and magnificent designs.

BIBLIOGRAPHY

- Bendsøe, M. P. 1989. "Optimal Shape Design as a Material Distribution Problem." *Structural Optimization* 1 (4): 193–202. doi:10.1007/BF01650949.
- BFS 2013:10 EKS 9. 2011. "Eks-9-En-2015-01-14_National_Annex_to_EC," 1-157.
- Block, Philippe, Jan Knippers, Niloy J. Mitra, and Wenping Wang. 2014. "Advances in Architectural Geometry 2014." *Advances* 1 (11): 379. doi:\url{10.1007/978-3-7091-0309-8}.
- Boeykens, S. 2012. "Bridging Building Information Modeling and Parametric Design," 453–58.
- Byrne, Jonathan, Michael Fenton, Erik Hemberg, James Mcdermott, O Neill, Elizabeth Shotton, and Ciaran Nally. 2011. "Combining Structural Analysis and Multi-Objective Criteria for Evolutionary Architectural Design," 204–13.
- Christensen, Peter W., and Anders Klabring. 2009. *An Introduction to Structural Optimization*. doi:10.1007/978-1-4020-8666-3.
- Christiansen, Asger Nyman, J. Andreas Bærentzen, Morten Nobel-Jørgensen, Niels Aage, and Ole Sigmund. 2015. "Combined Shape and Topology Optimization of 3D Structures." *Computers & Graphics* 46: 25–35. doi:10.1016/j.cag.2014.09.021.
- Davis, Daniel. 2013. "Modelled on Software Engineering: Flexible Parametric Models in the Practice of Architecture," no. February: 243. http://www.danieldavis.com/papers/danieldavis_thesis.pdf.
- DS/EN 1991-1-3 DK NA:2015. 2011. "DS/EN 1991-1-3 DK NA:2015," 1-7.
- DS/EN 1991-1-3:2007. 2007. DS-EN 1991-1-3 Dansk Standard Eurocode 1: Last På

Bærende Konstruktioner – Generelle Laster – Snelast. Edited by DS/S-19000 Eurocode Forum.

- http://sd.ds.dk.zorac.aub.aau.dk/Viewer?ProjectNr=M222442&Status=60.61&Page=0.
- DS/EN 1991-1-4. 2007. DS-EN 1991-1-4 Dansk Standard Eurocode 1: Last På Bærende Konstruktioner Generelle Laster Vindlast. http://sd.ds.dk.zorac.aub.aau.dk/Viewer?ProjectNr=M222442&Status=60.61&Page= 0.
- DS/EN 1991-1-4 DK NA:2015. 2015. DS/EN 1991-1-4 DK NA:2015 National Annex to Eurocode 1: Actions on Structures Part 1-4: General Actions Wind Actions.
- DS-EN 1990 Basis of structural design. 2007. DS-EN 1990 Dansk Standard Eurocode 0: Projekteringsgrundlag for Bærende Konstruktioner Eurocode Basis of Structural Design.
- Eurocode3, Design of steel structures. 2001. "Design of Steel Structures" 8523: 61.
- Gallas, Mohamed-anis, Kevin Jacquot, Sylvie Jancart, Frederic Delvaux, and User Cognition. n.d. "Parametric Modeling: An Advanced Design Process for" 2: 149–57.
- http://stednetwork.org/. n.d. "Sted." http://stednetwork.org/index.php?p=/article/2015/what-is-sted.
- IEA. 1997. "Integrated Design Process." *IEA Taks 23*. http://task23.iea-shc.org/integrated-design-process.
- IST EN 1991-1-1:2002/NA:2010. 2010. "Icelandic National Annexes to EurocodesSýnishorn." *Íslands, Staðlaráð*.
- Jensen, Bjarne Chr. 2007. Teknisk Ståbi. Nyt Teknisk Forlag.
- Kirsch, Uri. 1993. *Applications, Structural Optimization: Fundamentals and Applications.* Springer. https://books.google.dk/books?id=I7PoCAAAQBAJ&pg=PA262&dq=structural+optimization+process&hl=en&sa=X&ved=0ahUKEwiuzOShpb7MAhUKiSwKHWRFD0IQ6AEIKDAC#v=onepage&q=structural optimization process&f=true.
- Klitgaard, J, P H Kirkegaard, and M Mullins. 2006. "On the Integration of Digital Design and Analysis Tools" 90: 187–96. doi:10.2495/DARC060191.
- Knudsen, Mikkel Kirkeskov. 2016. "Sustainable Design in an Indoor Climate and Energy Perspective." *Investigation of EKO-Canopy Energy Renovation Concept for Swedish Million Program Apartment Buildings A Case Study of Dragonvägen*. doi:10.1017/CB09781107415324.004.
- Löhnert, Günter, Andreas Dalkowski, and Werner Sutter. 2003. "Integrated Design Process Guideline," no. April. http://task23.iea-shc.org/data/sites/1/publications/IDPGuide_internal.pdf.
- Miljonprogrammet. 1965. "No Title." http://www.stockholmskallan.se/Tema/Stockholmsbebyggelseutveckling/Miljonprogram/.
- Negendahl, Kristoffer. 2015a. "Building Performance Simulation in the Early Design Stage: An Introduction to Integrated Dynamic Models." *Automation in Construction* 54. Elsevier B.V.: 39–53. doi:10.1016/j.autcon.2015.03.002.

- ———. 2015b. "Parametric Design and Analysis Framework with Integrated Dynamic Models." *International Workshop on Design in Civil and Environmental Engineering* 23: 1–7.
- NF EN 1991-1-4. 1991. "NF EN 1991-1-4/NA Eurocode 1: Actions on Structures Part 1-4: General Actions Wind Actions National Annex to NF EN 1991-1-4:2005" 1 (1): 3-6.
- NIMBS Committe. 2007. "National Building Information Modeling Standard." Nbim, 180.
- Nordicbuilt. 2012. "No Title." http://www.nordicinnovation.org/nordicbuilt.
- NS-EN 1991-1-3:2005/NA:2009. n.d. "NS-EN 1991-1-3 Pkt. NA.4.1 (Snow).pdf."
- NS-EN 1991-1-4:2005/NA:2009. n.d. "Najsjonalt Tillegg NS-EN 1994-1-1 Tab. NA.4."
- Nyman, Asger, and Jakob Andreas. 2015. "Combined Shape and Topology Optimization Asger Nyman Christiansen."
- Preisinger, Clemens. 2015. Karamba User Manual (Version 1.1.0).
- Rutten, David. 2016. "Galapagos Evolutionary Solver." http://www.grasshopper3d.com/group/galapagos.
- SFS-EN 1991-1-3 NA. 2003. "ANNEX 4 NATIONAL ANNEX SFS-EN 1991-1-3 EUROCODE 1 : ACTIONS ON STRUCTURES Part 1 3 : General Actions . Snow Loads" 2 (2): 1–7.
- Sigmund, O. 2001. "A 99 Line Topology Optimization Code Written in Matlab." *Structural and Multidisciplinary Optimization* 21 (2): 120–27. doi:10.1007/s001580050176.
- Söderberg, Hans, and Stefan Bergström. 2008. "Wind Mapping of Sweden," no. January 2008.
- Stavric, M, and O Marina. 2011. "Parametric Modeling for Advanced Architecture." *International Journal of Applied Mathematics and Informatics* 5 (1): 9–16.
- W Young Jr., Norbert, Stephen A. Jones, Harvey M. Bernstein, and John E. Gudgel. 2009. "Getting Building Information Modeling to the Bottom Line." *Construction*, 213–22. doi:10.1080/17512549.2015.1043643.
- White, Architects. 2015. "Eko Canopy_Hus I Hus-Arkitektur Med Innovativ Användning Av Restvärme."
- Woodbury, Robert. 2010. "Elements of Parametric Design." *Taylor and Francis*.
- Xie, Y. M., and G. P. Steven. 1993. "A Simple Evolutionary Procedure for Structural Optimization." *Computers and Structures* 49 (5): 885–96. doi:10.1016/0045-7949(93)90035-C.
- Delft, T. (Janurary May de 2016). *Grasshopper Data Tree Editing*. http://wiki.bk.tudelft.nl/toi-pedia/Grasshopper_Data_Tree_Editing
- Model LAB. (January May de 2016). The Grasshopper Primer (EN), Third Edition V3.3 Mode Lab 2015. Published with GitBook. http://grasshopperprimer.com/en/index.html
- David Rutten. I Eat Bugs For Breakfast. (Januarry May de 2016).

https://ieatbugsforbreakfast.wordpress.com/2011/03/04/fitness-functions/

LIST OF FIGURES

13
the one block apartment (down) 14
e between two apartment blocks to be
15
of overall concept of the Canopy(right)15
l (Negendahl 2015a)17
of the different agents in the building 2006), The three approaches – (a) the ach provided by the prototype, showing k arrow and the detailing phase on the
proposed by Klitgaard (Klitgaard et al19
[Negendahl 2014. Figure6: a) combined kage), b) central model method (using a nodel method (utilizing a middleware)]
of overall concept of the Canopy(rig

Figure 4.2-1: Developed Integrated Dynamic model	. 25
Figure 4.2-2: Workflow for the specific method. Steps carried out in each stage	. 26
Figure 4.2-3: Applied Integrated design process	. 27
Figure 4.3-1: Optimization method scheme	. 30
Figure 5.1-1: Wind rose for the area (left] and scheme of the are (right)	. 45
Figure 5.1-2 Flat roof wind areas (left) and monpith roof wind areas (right) a/EC1-4 (DS, 1991-1-4 2007)	
Figure 5.1-3: Diagrams from EC1-4 (DS/EN 1991-1-4 2007) to calculate pressure coefficient the facades.	
Figure 5.2-1: Case A – Parametric geometry	. 50
Figure 5.2-2 Final configuration. Cross section values (left) deformation of the structure (rig	_
Figure 5.2-3: Box domain for case B	. 59
Figure 5.2-4: Possible geometries case B-1(only shown the main load bearing structure, not covering)	
Figure 5.2-5: Loads applied to the geometry (left) and roof load areas (right) case B - 1	. 60
Figure 5.2-6: Configuration iteration № 8 (X=4, Y=8)	. 62
Figure 5.2-7: Support definition - beams and columns (left) and support definition bea columns and plates (right)	
Figure 5.2-8: Final Geometry for case B - 2	. 65
Figure 5.2-9 Geometry for case B - 3	. 66

LIST OF GRAPHS

Graph 5.1-1: Wind loads pressure values depending on the country. Flat roof case40
Graph 5.1-2: Snow loads for the different countries4
Graph 5.2-1: Maximum displacement [m] over time for the steps 7 (green) and step 9 (orange
Graph 5.2-2 Zoom in first 100 iterations of maximum displacement [m] for the steps 7 (green and step 9 (orange)
Graph 5.2-3: Total amount of mass [kg] over time for the steps 7 (green) and step 9 (orange) 54
Graph 5.2-4: Zoom in first 100 iterations of total amount of mass [kg] for the steps 7 (green and step 9 (orange)
Graph 5.2-5: Comparison of total mass for the 36 different iterations54
Graph 5.2-6: Maximum deflection and total mass for the 36 variables5!
Graph 5.2-7: Utilization ratio versus total mass of the structure
Graph 5.2-8: Comparison of total mass of the structure 1st and 2nd loop5
Graph 5.2-9: Total amount of mass [kg] over time for the steps 7 (green) and step 9 (orange)case B - 16
Graph 5.2-10: Zoom in first 100 iterations of total amount of mass [kg] for the steps 7 (green and step 9 (orange)
Graph 5.2-11: Utilization ratio vs total mass of the structure62
Graph 5.2-12: Maximum displacement columns case B-264

Graph 5.2-13: Total mass structure case 2-B	64
Graph 5.2-14: Utilization of Cross-section vs Mass of the Total Structure	64
Graph 5.2-15: Comparison Total mass of structure for case B- 1 and B - 2	65

LIST OF TABLES

Table 4.4-1: Basic measurements of the site for the Canopy construction	42
Table 5.1-1: Values for the basic wind velocity for each Country	44
Table 5.1-2: Characteristics values for the snow load on the ground for the different lo	ocations
	47
Table 5.1-3: Load combinations	49
Table 5.2-1: Loads applied to the geometry (left) and roof load areas (right) case A	51
Table 5.2-2: Combination of frames X and Y direction for the 36 variables	55
Table 5.2-3: Characteristics values for the snow load on the ground for the different lo	ocations
	57
Table 5.2-4: Basic measurements of the site for the Canopy construction	59
Table 5.2-5: Comparison structures B -1 and B - 2	66
Table 5.2-6: Comparison of the total amount of mass for the three models	67

APPENDIX

APPENDIX A: THE EKO-CANOPY PROJECT

APPENDIX B: STED PROGRAM TASKS

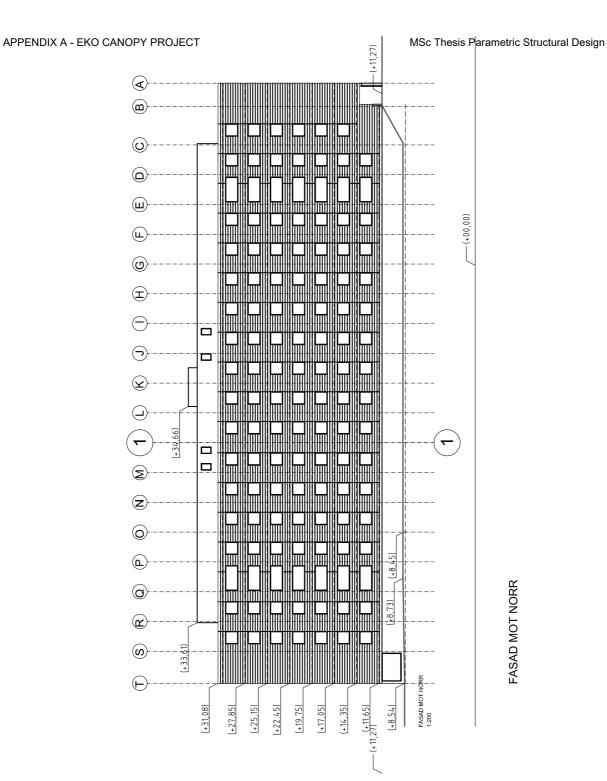
APPENDIX C: LOAD CALCULATIONS

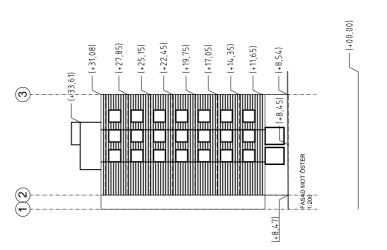
APPENDIX D: RESULTS OPTIMIZATION PROCESSES

APPENDIX E: OUTPUT FROM GRASSHOPPER. SECTION FORCES

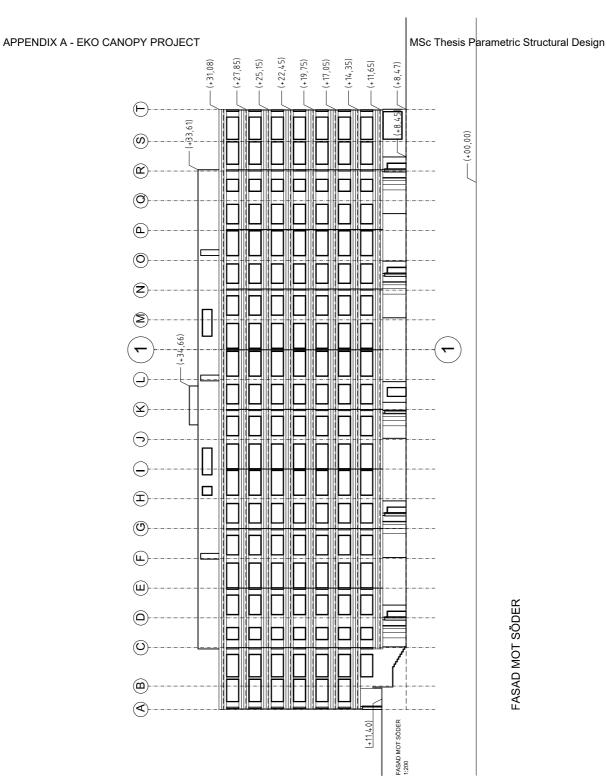
APPENDIX F: SCRIPT IN GRASSHOPPER DEFINITION

APPENDIX A: THE EKO-CANOPY PROJECT

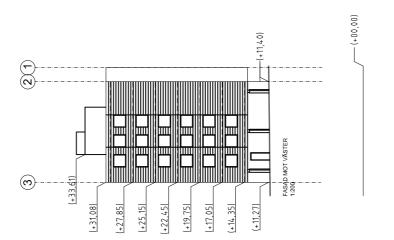




FASAD MOT ÖSTER



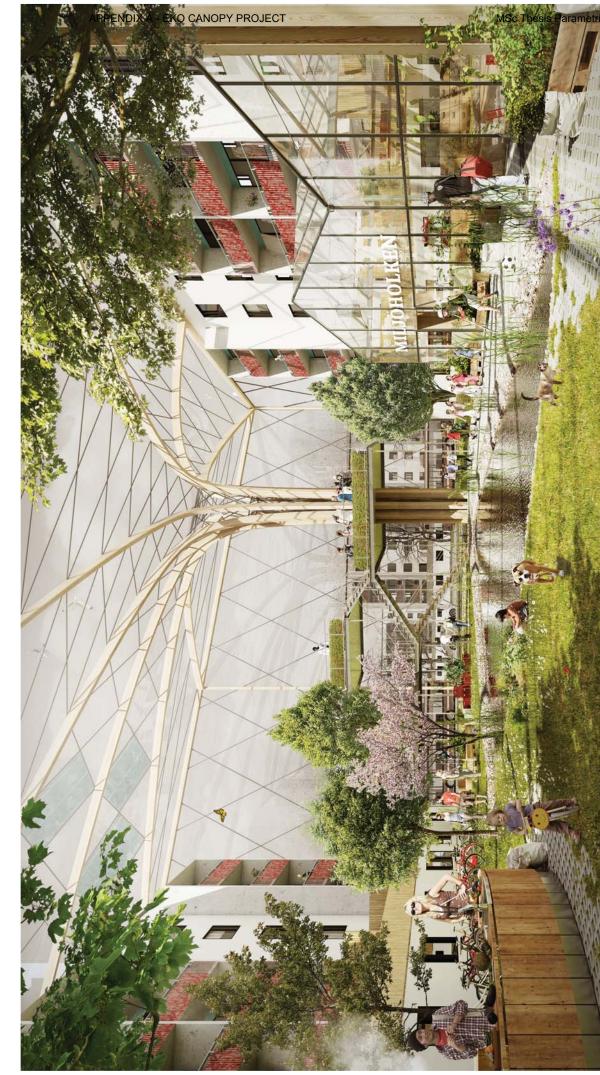
FASAD MOT SÖDER



FASAD MOT VÄSTER

13850 12160 72560

HUS 2514 PLAN 15



HUS I HUS-ARKITEKTUR MED INNOVATIV ANVÄNDNING AV RESTVÄRME

GEMENSKAPSBYGGANDE FÖR GRANNSKAPET

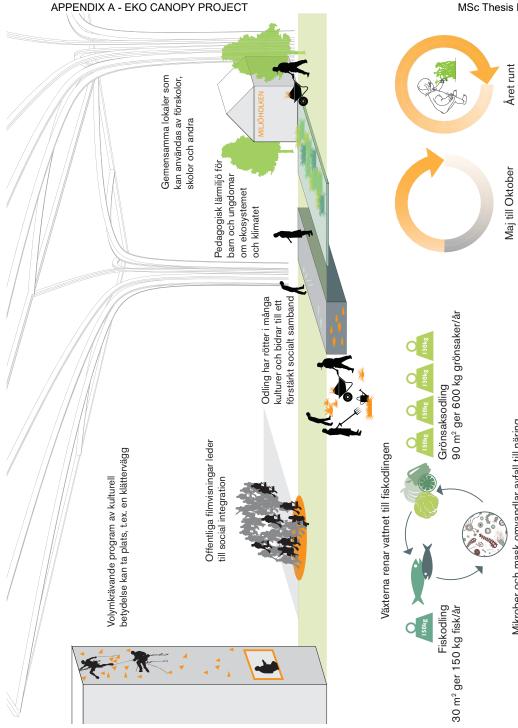
En mötesplats för socialt utbyte och en högre integration av Målbilden är att skapa ett gemensamt rum för alla grannar. människor från olika kulturer i miljonprogrammet.

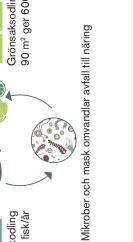
sina lägenheter och de saknas pengar till att använda gemensam plats att mötas i. De flesta bor mycket tätt Många av dagens miljonprogramsområden saknar en kostnadspliktiga kulturella mötesplatser i staden.

Även i nya bebyggelseområde kan konceptet användas, t. ex. befintliga hus inte behöver en strukturell helhetsrenovering, delvis på passivhusstandard. Dock fungerar strukturen och tradition finns en större användning av utomhusmiljön som vardagsrum. En ytterligare fördel med glasstrukturen är att kan mötas och umgås. Många i miljonprogramsområdena påverkar eller gör något ingrepp i bebyggelsestrukturen. Eko-Canopy skapar en lokal mötesplats där grannarna konceptet även i andra miljöer, eftersom det inte direkt bara ett tillägg som minskar husens energianvändning har tidigare invandrat från sydliga länder, där det av offentliga byggnader.

ETT AKVAPONISKT SYSTEM

kvävefixerande bakterier och sedan filtreras av växterna som till fiskarna. Restvärmen kan användas att hålla rätt temperasåsom fisk, skaldjur eller alger i bassänger) med hydroponik tur i bassängerna och i växthuset och sparar mycket energi 120 kvm och producerar 150 kg fisk och 600 kg grönsaker Anläggningen som ska finnas i projektet tar en yta av totalt (odling av växter i vatten) i en symbiotisk miljö. Vattnet leds tar upp näringsämnena. Det renade vattnet återförs sedan Akvaponik är ett hållbart matproduktionssystem som kom binerar konventionellt vattenbruk (odling av akvatiska djur till ett hydroponiskt system där avföringen bryts ner av jämförelse med konventionella akvaponiska system.





gemensam aktivitet och odling Sociala mötesplatser med

restvärme för ett akvaponiskt system. Turister och den lokala befolkningen är ett exempel på användning av besöker verksamheten. Här finns oerättar om de hållbara tankarna även ett integrerat museum som Iropenhaus Frutingen, Schweiz, kring projektet.



en högre grönytefaktor. Terrasserna

kan även sammanläggas med uppdelade i offentliga, semi-

balkongerna och de kan bli offentliga och privata delar.

EKO-CANOPY

möjliggör planteringsytor och växter bidrar till flera platsbildningar och

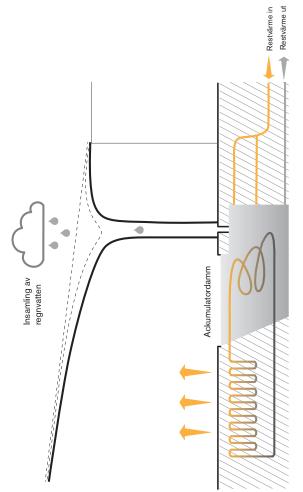
Ferrassema och strukturen

Den vertikala trädgården MFO Park Zürich, Schweiz. den vertikala dimensionen. Det

Foto: Roland Fischer







SÅ FUNGERAR SYSTEMET

Trattkantarellstrukturen i glastaket samlar regnvatten restvärmen har desto högre upp i dammen tas den in för att få en bra temperaturskiktning i dammen. Restvärmen (som har en temperatur av 30-50°C) växlas värmen över till det lokala nätet. Här löper ett rör i marken som går in under glasbyggnaden går förbi området i ett rör. Via en värmeväxlare till en ackumulatordamm. Ju högre temperatur som leds ner i dammen.

samt jämnar ut temperaturfluktuationer över dygn och I dammen i sin tur finns ett slingsystem som hämtar värme och sedan transporterar runt den i marken för som en ackumulatortank och regnvattenuppsamlare att avge värmen till "uterummet". Dammen fungerar

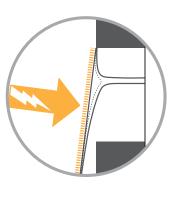
Slingsystemet fungerar likt ett golvärmesystem för att skapa en stor värmeöverförande yta. Detta gör att

värmeeffekten blir tillräckligt stor för att hålla klimatet i glasstrukturen på en behaglig temperatur året om.

kan bilda komfortabelt lagom uppvärmda bänkar och med slingor men även lufthängda rörstrukturer som Den återstående restvärmen som går ut från ackumulatordammen leds vidare för att värma upp busskurer, lekplatser mm via dels samma princip sittplatser.

som sen går in i byggnaderna och sparar på så sätt på För ventileringen av själva glashuset sker detta på helt Luckorna öppnas så inte temperaturen når över 25°C luckor i taket som öppnas via styrning på temperatur. värmeenergi. Frisk luft tas in långt ner i glasfasaden naturlig väg. För att undvika övertemperaturer finns och stiger sedan uppåt genom att den blir varmare. Det inglasade området förvärmer ventilationsluften så länge det inte är varmare utomhus.

-0



ENERGIANVÄNDNING

Transparenta solceller är integrerade i glasstrukturens tak. Eftersom största delen av taket är riktat mot söder

SOLCELLER

i bra vinkel mot solen (ca 10-15 grader) kan mycket täckningagraden 50 % för att skapa en bra balans

15 °C. Glasen ska väljas med omsorg, lågt U-värde solens strålar värmer upp luften innanför. Då solen räcker för att hålla temperaturen vintertid på cirka Samtidigt måste glaset släppa in mycket dagsljus (<0,7 W/m²,K) för att restvärmen ska räcka långt. Genom att addera ett extra klimatskal skapas en buffert till befintlig eller nybyggnation. Eftersom området är inglasat skapas en växthuseffekt där inte räcker som värmekälla kommer restvärmen via hög ljustransmittans. Istället för lågt g-värde in. Restvärmen bidrar med en värmeeffekt som på glasen (solfaktor) balanseras solen med ransparenta solceller.

brukaren befinner sig utomhus även om det egentligen

mellan energiproducering och inflöde av dagsljus. el produceras. I glastaket mellan byggnaderna är

Fönstrena är ljusa för att skapa en illusion av att

är tillstängt. Högre upp i glasstrukturen, ovanför ena

byggnadens tak, kan solcellsandelen öka ända

till 100 %.



Environment där varje rum byggts upp med en zon vardera. Programmet tar hänsyn till solinstrålning, där inkommande vatten är 40°C. Resultatet visar modellerat i modellen som ett golvvärmesystem på en 50% sänkning av energianvändningen. En energiberäkning har utförts i IES Virtual omkringliggande byggnader. Det inglasade området uppvärms med restvärme vilket är byggnadens tröghet och skuggningar från

yttre påverkan. Vind, regn och blåst hålls ute vilket ger

Genom det extra skalet fås ett naturligt skydd mot

ett bra "utomhusklimat inomhus".

20-

10 -

EXTRA KLIMAT- OCH VINDSKYDD

MAX MAX HÖST SOMMAR vÅR -EFTER FÖRE - VINTER

Samlad regnvatten kan användas för t ex toaspolning

av ett FTX-system, där plats för kanaler i byggnaden är ett

EKO-CANOPY

50 % medan uppe på bostadstaket kan täckningen sakta

populär, bland annat för olika till en "stad i trädgården" Parken har blivit väldigt regeringen i Singapore förvandla en "trädgårds: integrerad del i strategi Gardens by the bay är Gardens by the bay

Foto: Allie Caulfield evenemang.

EKONOMI

0

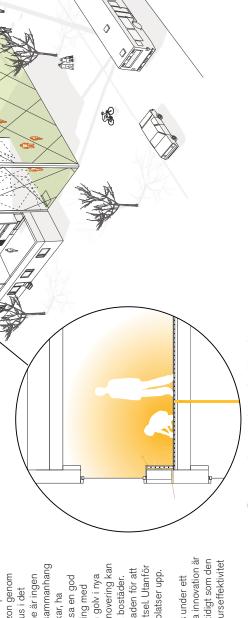
Eftersom uterummet minskar behovet av prima värme i byggnaderna kan dessa energikostnader skjutas till i tekniskt som ekonomiskt. Den uppbyggda strukturen samt energiproducerande. Det extra skalet tar också fönster och tilläggsisolera fasaderna mot uterummet byggs detta skyddande skal upp av glas. Pengarna om hand energiläckaget från de köldbryggor som strukturen ska vara genomförbar, såväl byggnadsofta finns vid balkongerna på miljonprogramshus. är i trä och materialsnål. Istället för att byta ut alla läggs på att få denna struktur tät och energisnål Grundtanken i detta förslag är att trattkantarell·

APPENDIX A - EKO CANOPY PROJECT

INNOVATION

nyhet. Att däremot sätta det i ett socialt sammanhang byggnader (se figur till höger) eller vid renovering kan täcka byggnadernas elbehov för fastighetsel. Utanför Addera solceller som integreras i glasfasaden för att Genom ett till synes enkelt glasöverdrag skapas helt glastäcket värms busshållplatser och lekplatser upp. nya unika miljöer. Vi tillåts att byta klimatzon genom att stiga utanför vår dörr. Att skapa växthus i det stora värmeöverförande ytor i väggar och golv i nya picknick eller bara hänga på en filt och läsa en god nordiska klimatet uppvärmt från restvärme är ingen restvärmen även räcka för att värma våra bostäder. bok. Genom att skapa effektiv uppvärmning med där människor kan mötas, odla, se på fiskar, ha

täcke och sätts i en social kontext. Denna innovation är enkel, genomförbar och duplicerbar samtidigt som den strävar mot ett modernt samhälle där resurseffektivitet En hel rad av befintlig teknik som samlas under ett och gemenskap är ledord.



Restvärme

lösning av vattenburen golv- och väggvärme. Restvärmen kan användas i en kombinerad

Bovieran

Ett seniorboende som

gemensam gård att vistas samman bostadshus med finns på ett antal platser Sverige där man kopplar en fjärrvärmeuppvärmd

Foto: Benno Nilsson



EKO-CANOPY

-50% Energianvändnii

nya byggnader kan värmas upp av vägg- och golvvärme bidrar till den höga mer tack vare den smarta glasstrukturen. Strukturen skapar en buffertzon som det är en elproducent. Förvärmning av ventilationsluft samtidigt som

APPENDIX B: STED PROGRAM TASKS

THE STED NETWORK NTNU SUSTAINABLE TRANSFORMATION &

Welcome to Nordic Build Sustainable Transformation & Environmental Design Network conference at NTNU Trondheim!

ENVIRONMENTAL DESIGN NETWORK

With the aim of coming together and share the knowledge we hope to have a day full of interesting reviews and transformational discussions on how we can create more quality and fastforward the track to a sustainable world.

Dialogue and exchange between academy and practitioners is key in this program, so we hope this day together will instigate new encounters for new perspectives and great ideas.

We start with an introduction exercise in order to set an open communication between every-one. After that we go through a series of fast elevator-pitches on the work that has been done. In short repetitive interval, we discuss the findings for a broad overview and so get the oppor-tunity to meet. For the afternoon we divide into 2 groups for more in depth discussions on your chosen theme.

Please note:

Practitioners please bring your research questions. What should be studied next? Are you ready to implement results?

For the ones that presents with posters, please note that it has to be hung and ready for 9.00 am.

A reminder to send all posters as pdf to Kristoffer Negendahl krnj@byg.dtu.dk

Program 8.30 - 16.30

8.30 Arrival NTNU, placing posters

9.00 Welcome by Aoife Houlihan Wiberg

9.10 The STED project, Lotte Bjerregaard Jensen

9.25 Circle event by Elise Grosse, White Arkitekter

10.00 3 minutes pitches of each poster combined with short pier-discussions.

12-13.30 tour NTNU and lunch

13.30 Choose your sessions

14.30 coffee brake

15.00 Choose your sessions

16.00 Check-out...Elise Grosse, White

16.30 End

18.30 Dinner

CONTACT

Lotte Bjerregaard Jensen Ibj@byg.dtu.dk



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Technical University of Denmark



PARAMETRIC STRUCTURAL DESIGN

OPTIMIZATION OF STRUCTURES BASED ON PARAMETRIC MODELLING & LCA

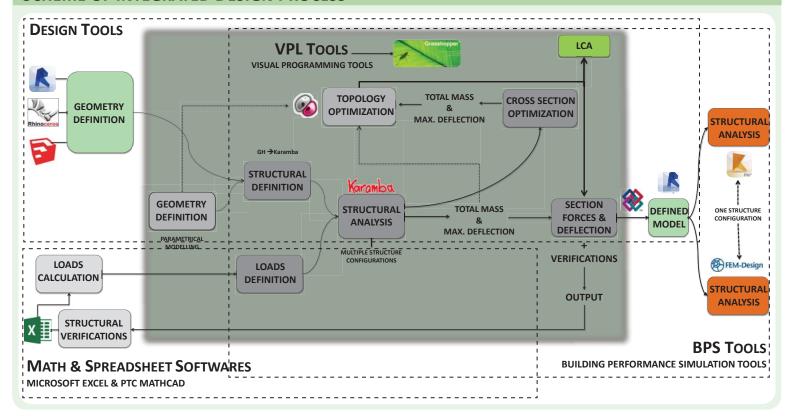
Master Thesis - Msc Architectural Engineering - Structural Design
Department of Civil Engineering - Technical University of Denmark
Advisors: Kristoffer Negendahl & Lotte Bjerregaard Jensen

Abstract - Integrated design process

The purpose of this project is to look for a method to integrate the structural aspects within the architectural designs in the first stages of designing process. As part of the STED (Sustainable Transformation & Environmental Design) programme inside of the Nordic Built, Nordic Innovation program, which principal goal is to achieve sustainable solutions in the Nordic region, the objective of the proposed method is to find reliable structures using the minimal amount of material, which means less kilograms of mass and consequently less emissions of CO2, benefiting the LCA process.

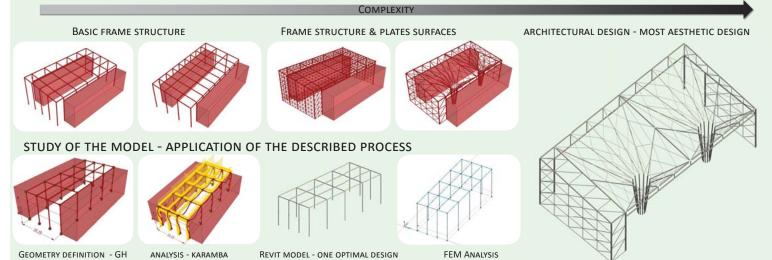
The methodology followed for this master thesis consists in an integrated process, in which the data is exchanged between several softwares (Grasshopper, Karamba, Excel, Revit and FEM-Design/Robot) in order to establish a logic flow where the information is updated automatically. Thanks to the parametrical modeling using the plug-in for Rinho 3D, Grasshopper is possible to establish a form-finding method based on optimization processes, which allows to study different possibilities for the same project in order to achieve the landscape of optimal structures.

SCHEME OF INTEGRATED DESIGN PROCESS

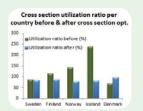


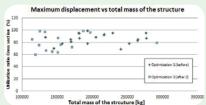
APPLICATION OF THE METHOD - BASED ON THE EKO-CANOPY PROJECT (PROPOSAL BY WHITE ARKITEKTER)

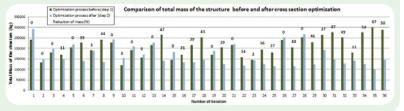
PARAMETRICAL MODELING - GEOMETRY DEFINITION -



ANALYSIS - POST PROCESSING







APPENDIX C: LOAD CALCULATIONS

APPENDIX C - 1 : WIND LOAD CALCULATIONS

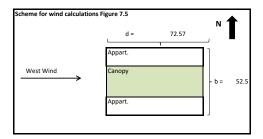
Wind Calculations depending on the Country

	1					DENMARK	SWEDEN	FINLAND	NORWAY	ICELAND
REF.			Reference		Units	DENWARK	SWE	FIN	NOR	ICE
EC-1-4 wind 4.2	Basic wind	velocity	eq. 4.1	V _b	m/s	24	24	21	26	2
NA - 4.2 (1)		basic wind velocity	cq. 4.1	V _{b,0}	m/s	24	24	21.0	26.0	2-
	Directional fac				111/5	1	1	1	26.0	24
	Season factor	.101		C _{dir}	=	1	1	1	1	
	Season ractor			C _{season}	-	1	1	1	1	•
EC-1-4 wind 4.3	Mean wind	d velocity (cat IV)	eq. 4.3	V _{m,IV}	m/s	17.54247273	17.54247273	15.34966364	19.00434546	17.5424727
	Heigth above	the terrain		z	m	22.63	22.63	22.63	22.63	22.63
	Roughness fac	ctor		C _r (z)	-	0.731	0.731	0.731	0.731	0.731
		zmin < z < zmax								
		Terrain category IV	Table 4.1	z ₀	m	1	1	1	1	1
				Z _{min}	m	10	10	10	10	10
				z _{max}	m	200	200	200	200	200
		Reference value category II		Z _{0,II}	m	0.05	0.05	0.05	0.05	0.05
		Terrain factor	eq.4.5	k _r	-	0.234	0.234	0.234	0.234	0.234
	Mean wind	d velocity (cat III)	eq. 4.3	V _{m,III}	m/s	22.34836246	22.34836246	19.55481715	24.210726	22.34836246
	Roughness fac	tor		C _r (z)	-	0.931	0.931	0.931	0.931	0.931
		zmin < z < zmax								
		Terrain category III	Table 4.1	z ₀	m	0.3	0.3	0.3	0.3	0.3
				Z _{min}	m	5	5	5	5	į
				z _{max}	m	200	200	200	200	200
		Reference value category II		Z _{O,II}	m	0.05	0.05	0.05	0.05	0.05
		Terrain factor	eq.4.5	k _r	-	0.215	0.215	0.215	0.215	0.215
	Ortography fa	ctor	A.3 Note (4)	c _o (z)	-	1	1	1	1	1
	Dard	••								
EC-1-4 wind 4.5	Реак уегос	ity pressure	eq 4.8	q _{p (IV)}	kN/m^2	0.61	0.61	0.47	0.72	0.61
				q _{p (III)}	kN/m^2	0.79	0.79	0.61	0.93	0.792
	Basic velocity		eq 4.10	q _b	N/m^2	360.00	360.00	275.63	422.50	360.00
	Exposure factor	Air density	Figure 4.2	ρ c _e (z) - IV	kg/m^3	1.25 1.7	1.25 1.7	1.25 1.7	1.25 1.7	1.25 1.7
	Exposure facto	J1	Figure 4.2	C _e (z) - III		2.2	2.2	2.2	2.2	2.2
FC-1-4 wind 5.2	Wind nress	sure on surfaces	rigure 4.2	Ge (2) III		2.2	2.2	2.2	2,2	2.2
	Pressure coef		Sectio	n 7						
		Wall (for q _{p,IV})	figure 7.4, h <b< td=""><td>w (Zone D)</td><td>kN/m^2</td><td>0.43</td><td>0.43</td><td>0.33</td><td>0.51</td><td>0.43</td></b<>	w (Zone D)	kN/m^2	0.43	0.43	0.33	0.51	0.43
						0.61	0.61	0.47	0.72	0.61
				w (Zone E)	kN/m^2	-0.19	-0.19	-0.15	-0.23	-0.19
						-0.19	-0.19	-0.15	-0.23	-0.19
		Flat roof (for q _{p,IV})		w (Zone F)	kN/m^2	-1.10	-1.10	-0.84	-1.29	-1.10
		(case 1)				-1.53	-1.53	-1.17	-1.80	-1.53
				w (Zone G)	kN/m^2	-0.73	-0.73	-0.56	-0.86	-0.73
				w (7ons II)	kN/m^2	-1.22 -0.43	-1.22 -0.43	-0.94 -0.33	-1.44 -0.50	-1.22 -0.43
				w (Zone H)	KIN/IIIZ	-0.43	-0.43	-0.56	-0.50	-0.43
				w (Zone I)	kN/m^2	0.12	0.12	0.09	0.14	0.12
						0.12	0.12	0.09	0.14	0.12
		Monotpitch roof (for q _{p,IV})		w (Zone F_up)	kN/m^2	-1.45	-1.45	-1.11	-1.71	-1.45
				/a	1817	-1.76	-1.76	-1.35	-2.07	-1.76
		(case 2)		w (Zone F_low)	kN/m^2	-1.00 -1.47	-1.00 -1.47	-0.77 -1.12	-1.18 -1.72	-1.00 -1.47
		(case 2)		w (Zone G)	kN/m^2	-1.47 -1.16	-1.47	-1.12 -0.89	-1.72 -1.36	-1.4. -1.16
				(20 0)	, 111 2	-1.50	-1.50	-1.15	-1.77	-1.5
				w (Zone H)	kN/m^2	-0.48	-0.48	-0.37	-0.56	-0.4
						-0.73	-0.73	-0.56	-0.86	-0.7
				w (Zone I)	kN/m^2	-0.42	-0.42	-0.32	-0.49	-0.42
						-0.70	-0.70	-0.54	-0.82	-0.70

APPENDIX C - 1 : WIND LOAD CALCULATIONS Parameters Wind Calculations

EC-1-4 wind 5.2 Wind pres	ssure on surfaces				
Pressure coe	efficient	Section 7	7		
	Wall				
	Case 1 figure 7.4, h <b< th=""><th></th><th></th><th></th><th></th></b<>				
	e=min(b,2h)	Figure 7.5	5 e	m	28.25
			b	m	28.25
			2h	m	45.26
			d	m	72.57
	Case 1, e <d< th=""><th></th><th></th><th></th><th></th></d<>				
	Zone	A	e/5	m	5.65
	Zone	В	e (e/5+4/5e)	m	22.60
	Zone	С	d-e	m	44.32
	Reference heigt	h Figure 7.4	ze = h	m	22.63
	Flat roof, case 1	Figure 7.6			
	Zone F	length	e/4	m	7.06
		width	e/10	m	2.83
	Zone G	length	b-2*e/4	m	14.13
		width	e/10	m	2.83
	Zone H	length	b	m	52.45
		width	e/2-e/10	m	11.30
	Zone I	length	b	m	52.45
		width	d - e/2	m	58.445
	Pitch roof, case 1	Figure 7.6			
	Cross wind direction	fig 7.7 c)	b	m	40.35
			h.1	m	30.63
			h.2	m	25.63
			2h	m	61.26
			d	m	80.57
			e		40.35
	Zone F	length	e/4	m	10.09
		width	e/10	m	4.04
	Zone G	length	b-2*e/4	m	20.18
		width	e/10	m	4.04
	Zone H	length	b	m	40.35
		width	e/2-e/10	m	16.14
	Zone I	length	b	m	40.35
		width	d - e/2	m	60.40

Canopy measu	rements - 1st mod	el	
Height	m	22.63	
Lenght	m	72.57	
Width	m	28.25	
Appartment n	neasurements		
Height	m	22.63	
Lenght	m	72.57	
Width	m	12.1	
Unit: Appartn	nent - Canopy - App	artment	
Height	m	22.63	
Lenght	m	72.57	
Width	m	52.45	
extension (len	gth) m		8
extension (roo		8	
extension (roo	f) - low eave m		3
Desnivel			5
angle			7.06



Pressure coefficients tables according to Eurocode

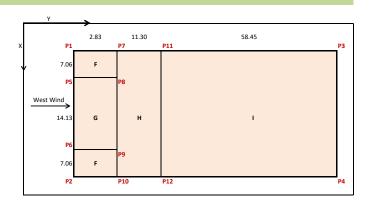
Pres	Pressure coefficient, wall		Table 7	Table 7.1 EC-1-4							
zone	e A		В		С			D			
h/d	C _{pe,1}	C _{pe,1}	C _{pe,10}	C _{pe,1}	C	e,10 C _p	e,1	C _{pe,10}		pe,10	C _{pe,1}
	1	-1.2	-1.4	-0.8	-1.1	-0.5	-0.5	0.8	1	-0.5	-0.5
	#¡DIV/0!	-1.2	-1.4	-0.8	-1.1	-0.5	-0.5	0.708	1.000	-0.316	-0.32
	0.25	-1.2	-1.4	-0.8	-1.1	-0.5	-0.5	0.7	1	-0.3	-0.3

Pressure coefficient, flat roof			Table 7.1 E	C-1-4					
zone	F		G		н		1		
h.p/h	C _{pe,10}	C _{pe,1}	C _{pe,10}	C _{pe,1}	C _{pe,10}	$C_{pe,1}$	C _{pe,10}	C _{pe,1}	
Sharp eaves	-	1.8	-2.5	-1.2	-2.0	-0.7	-1.2	0.2	0.2

Pressure o	coefficient, mor	nopitch roof	Table 7.3b	EC-1-4							\neg
zone	F_up	p	F_low		G		н		1		
angle	C _{pe,1}	0 C _{pe,1}	C _{pe,10}	$C_{pe,1}$	C_{pe}	,10 C _{pe,1}	C _{pe,10}	C _{pe,1}	C _{pe,10}	C _{pe,1}	
	5	-2.1	-2.6	-2.1	-2.4	-1.8	-2	-0.6	-1.2	-0.5	-0.5
	7.06	-2.38	-2.88	-1.64	-2.40	-1.89	-2.46	-0.78	-1.20	-0.68	-1.14
	15	-2.4	-2.9	-1.6	-2.4	-1.9	-2.5	-0.8	-1.2	-0.7	-1.2

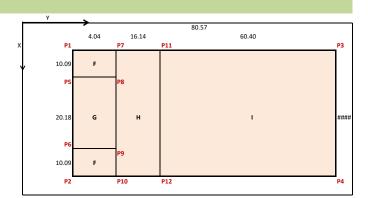
APPENDIX C - 1 : WIND LOAD CALCULATIONS CASE A -COORDINATES WIND ZONES FLAT ROOF

Coordinates				
	Relative	Absolute		
Point	х	x	у	z
P1	0	12.1	. 0	22.63
P2	28.25	40.35	0	22.63
P3	0	12.1	72.57	22.63
P4	28.25	40.35	72.57	22.63
P5	7.06	19.16	0	22.63
P6	21.19	33.29	0	22.63
P7	0	12.1	2.83	22.63
P8	7.06	19.16	2.83	22.63
P9	21.19	33.29	2.83	22.63
P10	28.25	40.35	2.83	22.63
P11	0	12.1	14.13	22.63
P12	28.25	40.35	14.13	22.63



CASE B - COORDINATES WIND ZONES MONOPITCH ROOF

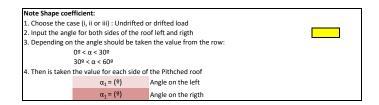
Coordinates				
	Relative	Absolute		
Point		x	У	z
P1		0.00	-8.00	30.63
P2		40.35	-8.00	30.63
P3		0.00	72.57	30.63
P4		40.35	72.57	30.63
P5		10.09	-8.00	30.63
P6		30.26	-8.00	30.63
P7		0.00	-3.97	30.63
P8		10.09	-3.97	30.63
P9		30.26	-3.97	30.63
P10		40.35	-3.97	30.63
P11		0.00	12.18	30.63
P12		40.35	12.18	30.63



APPENDIX C - 2: SNOW LOAD CALCULATIONS

Snow Calculations depending on the Country

								DENMARK	SWEDEN	FINLAND	NORWAY	ICELAND
REF.			Reference				Units	DK	SWE	FIN	NOR	ICE
EC-1-3	Snow load		5.1									
Section 5	Persistent/transier	nt design situatio	n 5.1	$s = \mu_i C_e C_t s_k$		s	kN/m^2	0.80	1.60	2.20	2.80	4.88
	Accidental design situ	uation (snow load) 5.2	$s = \mu_i C_e C_t s_{Ad}$		s	kN/m^2					
	Accidental design situ	uation (snow drift) 5.3	s = μ _i S _k		s	kN/m^2	0.67	1.33	1.83	2.33	4.07
	Carachteristic value	(location)	4.3 (1)			Sk	kN/m^2	1.00	2.00	2.75	3.50	6.10
	Exposure coefficient		5.1 (7)			Ce	-	1	1	1	1	1
	Thermal coefficient		5.1 (8)			ct	-	1	1	1	1	1
	Shape coefficient		5.3									
		Monopitch	Fig. 5.1	$0^{\circ} < \alpha < 30^{\circ}$	$\alpha_1 = (9)$ 7.064	μ_1	-	0.8	0.8	0.8	0.8	0.8
				30º < α < 60º	α ₁ = (º) 35	μ_1	-	0.67	0.67	0.67	0.67	0.67
		Pitched roof	Fig 5.3									
		Undrifted loa	d Case (i)	0º < α < 30º	$\alpha_1 = (9)$ 5	μ_1	-	0.80	0.80	0.80	0.80	0.80
				30º < α < 60º	$\alpha_1 = (9)$ 35	μ_1	-	0.67	0.67	0.67	0.67	0.67
				0º < α < 30º	$\alpha_2 = (9)$ 5	μ_1	-	0.80	0.80	0.80	0.80	0.80
				30º < α < 60º	$\alpha_2 = (9)$ 35	μ_1	-	0.67	0.67	0.67	0.67	0.67
		Drifted loa	d Case (ii)	0º < α < 30º	$\alpha_1 = (9)$ 5	μ_1	-	0.40	0.40	0.40	0.40	0.40
				30º < α < 60º	$\alpha_1 = (9)$ 35	μ_1	-	0.33	0.33	0.33	0.33	0.33
				0º < α < 30º	$\alpha_2 = (9)$ 5	μ_1	-	0.80	1.80	2.80	3.80	4.80
				30º < α < 60º	α ₂ = (⁹) 35	μ_1	-	0.67	0.67	0.67	0.67	0.67
		Drifted loa	d Case (iii)	0º < α < 30º	$\alpha_1 = (9)$ 5	μ_1	-	0.80	1.80	2.80	3.80	4.80
				30º < α < 60º	α ₁ = (⁰) 35	μ_1	-	0.67	0.67	0.67	0.67	0.67
				0º < α < 30º	$\alpha_2 = (9)$ 5	μ_1	-	0.40	0.40	0.40	0.40	0.40
				30º < α < 60º	$\alpha_2 = (9)$ 35	μ_1	-	0.33	0.33	0.33	0.33	0.33
		Cylindrical roo	f fig. 5.6									
				$\beta > 60^{\rm o}$, $\mu 3 = 0$	β = (º) 70	μ_1	-	0	0	0	0	0
		Case (i	eq. 5.4	β < 60º, μ3=0,2+10*h/b	β = (º) 35	μ_1	-	0.80	0.80	0.80	0.80	0.80
					<u></u>	μ_3	-	1.97	1.97	1.97	1.97	1.97
					h = 5		m					
					b = 28.25		m					
					h/b = 0.177	0 E *	-					
						0,5 * μ ₃	-					



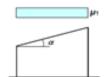


Figure 5.2: Snow load shape coefficient - monopitch roof

APPENDIX C - 3: LOADS INPUT GRASSHOPPER

Input Grasshoper

Modelo A - Basic Frame configuration

1.Win	d				C	Case	Flat roof (for qp,IV)	
					COUNTRY				
		Reference		Units	SWE	FIN	NOR	ICE	DK
Pressure	coefficient	Section	1 7						
	Wall (for qp,IV)	figure 7.4, h <b< td=""><td>w (Zone D)</td><td>kN/m^2</td><td>0.43</td><td>0.33</td><td>0.51</td><td>0.43</td><td>0.43</td></b<>	w (Zone D)	kN/m^2	0.43	0.33	0.51	0.43	0.43
			w (Zone E)	kN/m^2	-0.19	-0.15	-0.23	-0.19	-0.19
	Flat roof (for qp,IV)		w (Zone F)	kN/m^2	-1.10	-0.84	-1.29	-1.10	-1.10
			w (Zone G)	kN/m^2	-0.73	-0.56	-0.86	-0.73	-0.73
			w (Zone H)	kN/m^2	-0.43	-0.33	-0.50	-0.43	-0.43
			w (Zone I)	kN/m^2	0.12	0.09	0.14	0.12	0.12

2 . Snow Case: Monopitch

	Reference		Units	SWE	FIN	NOR	ICE	DK
Persistent/transient design situation	5.1	s = μi Ce Ct sk	kN/m^2	1.600	2.200	2.800	4.880	0.800
Accidental design situation (snow load)	5.2	s = μi Ce Ct sAd	kN/m^2	0.000	0.000	0.000	0.000	0.000
Accidental design situation (snow drift)	5.3	s = μi Sk	kN/m^2	1.333	1.833	2.333	4.067	0.667

Modelo B - White Proposal

1. Win	d				C	Case	/onotpitch roc	of (for qp,IV)	
					COUNTRY	_			
		Reference		Units	SWE	SWE	SWE	SWE	SWE
Pressure	coefficient	Section	7						
	Wall (for qp,IV)	figure 7.4, h <b< th=""><th>w (Zone A)</th><th>kN/m^2</th><th>0.43</th><th>0.33</th><th>0.51</th><th>0.43</th><th>0.43</th></b<>	w (Zone A)	kN/m^2	0.43	0.33	0.51	0.43	0.43
			w (Zone B)	kN/m^2	-0.19	-0.15	-0.23	-0.19	-0.19
			w (Zone C)	kN/m^2	0.00	0.00	0.00	0.00	0.00
	Monotpitch roof (for qp,IV)		w (Zone F_up)	kN/m^2	-1.45	-1.11	-1.71	-1.45	-1.45
			w (Zone F_low)	kN/m^2	-1.00	-0.77	-1.18	-1.00	-1.00
			w (Zone G)	kN/m^2	-1.16	-0.89	-1.36	-1.16	-1.16
			w (Zone H)	kN/m^2	-0.48	-0.37	-0.56	-0.48	-0.48
			w (Zone I)	kN/m^2	-0.42	-0.32	-0.49	-0.42	-0.42

2 . 3110W					Case:	ivionotpitch ro	of (for qp,IV)	
	Reference		Units	SWE	FIN	NOR	ICE	DK
Persistent/transient design situation	5.1	s = μi Ce Ct sk	kN/m^2	1.600	2.200	2.800	4.880	0.800
Accidental design situation (snow load)	5.2	s = μi Ce Ct sAd	kN/m^2	0.000	0.000	0.000	0.000	0.000
Accidental design situation (snow drift)	5.3	s = μi Sk	kN/m^2	1.333	1.833	2.333	4.067	0.667

No	ite:	
1.	Chose the country	
2.	Choose the case	case
	a) For wind: Zone: III or IV and type of roof	
	b) For snow: type of roof, and case (load di	istribution)

APPENDIX D: RESULTS OPTIMIZATION PROCESSES

APPENDIX D - 1 : MODEL A

MODEL A Optimization process (Results 36 variables)

Steel s275 Glass

Structural material Cladding material

Parameters considered:

Load case

S	Step 7 - B	- Before cross section opt.	tion opt.						ld. Family	Middle			Rigth/left side		
Nº of it S	Steel class	Max. displacement (m) (Structure)	X - Number of Frames	Y - Number of Frames	Beams Top - X direction		Tipe of profile Length element Allowed max. (m) Displacement beams (m)	Allowed max. Displacement beams (m)	Beams Top - Y Direction	Tipe of profile	Length element (m)	Allowed max. Displacement beams (m)	Tipe of profile	Length element (m)	Allowed max. Displacement beams (m)
1 52	s275	0.013791	7	5	42	HEB 400	4.036	0.02018	40	40 HEB 600	14.514		0.0726 HEB 450	14.514	0.07257
2		0.041698	m	4	15	HEB 400	9.417	0.04708	16	HEB 600	18.143		0.0907 HEB 450	18.143	
8		0.026916	5	4	25	HEB 400	5.650	0.02825	24	HEB 600	18.143		0.0907 HEB 450	18.143	
4		0.029933	4	5	24	. HEB 400	7.063	0.03531	25	25 HEB 600	14.514		0.0726 HEB 450	14.514	0.07257
S		0.026862	9	4	30	30 HEB 400	4.708	0.02354	28	28 HEB 600	18.143	_	0.0907 HEB 450	18.143	0.0907
9		0.023915	9	9	42	42 HEB 400	4.708	0.02354	42	42 HEB 600	12.095		0.0605 HEB 450	12.095	0.06048
7		0.022023	5	5	30	30 HEB 400	5.650	0.02825	30	30 HEB 600	14.514		0.0726 HEB 450	14.514	0.07257
80		0.025726	9	7	48	48 HEB 400	4.708	0.02354	49	49 HEB 600	10.367		0.0518 HEB 450	10.367	0.05184
6		0.021414	7	4	35	35 HEB 400	4.036	0.02018	32	32 HEB 600	18.143		0.0907 HEB 450	18.143	0.09071
10		0.139556	e	c	12	12 HEB 400	9.417	0.04708	12	HEB 600	24.190		0.1210 HEB 450	24.190	0.12095
11		0.098624	9	c	24	24 HEB 400	4.708	0.02354	21	21 HEB 600	24.190		0.1210 HEB 450	24.190	0.12095
12		0.036137	4	4	20	20 HEB 400	7.063	0.03531	20	20 HEB 600	18.143		0.0907 HEB 450	18.143	0.0907
13		0.073279	7	c	28	28 HEB 400	4.036	0.02018		24 HEB 600	24.190		0.1210 HEB 450	24.190	
14		0.017082	7	7	99	56 HEB 400	4.036	0.02018		56 HEB 600	10.367		0.0518 HEB 450	10.367	0.05184
15		0.122798	4	c	16	16 HEB 400	7.063	0.03531	15	15 HEB 600	24.190		0.1210 HEB 450	24.190	0.12095
16		0.038189	3	7	24	24 HEB 400	9.417	0.04708	28	28 HEB 600	10.367		0.0518 HEB 450	10.367	
17		0.026218	5	7	40	40 HEB 400	5.650	0.02825	42	42 HEB 600	10.367		0.0518 HEB 450	10.367	0.05184
18		0.015771	7	9	49	49 HEB 400	4.036	0.02018	48	48 HEB 600	12.095		0.0605 HEB 450	12.095	
19		0.098679	2	ĸ	20	20 HEB 400	5.650	0.02825	18	18 HEB 600	24.190		0.1210 HEB 450	24.190	
20		0.024895		9	35	35 HEB 400	5.650	0.02825	36	36 HEB 600	12.095		0.0605 HEB 450	12.095	
21		0.021635		5	36	36 HEB 400	4.708	0.02354	35	35 HEB 600	14.514		0.0726 HEB 450	14.514	
22		0.036603	æ	9	21	21 HEB 400	9.417	0.04708	24	24 HEB 600	12.095		0.0605 HEB 450	12.095	
23		0.032865	æ	2	18	18 HEB 400	9.417	0.04708	20	20 HEB 600	14.514		0.0726 HEB 450	14.514	
24		0.034876	4	7	32	HEB 400	7.063	0.03531	32	35 HEB 600	10.367	_	0.0518 HEB 450	10.367	
25		0.03332	4	9	28	28 HEB 400	7.063	0.03531	30	30 HEB 600	12.095		HEB 450	12.095	
56		899690'0	∞	c	32	32 HEB 400	3.531	0.01766	27	27 HEB 600	24.190		0.1210 HEB 450	24.190	
27		0.035951	4	∞	36	36 HEB 400	7.063	0.03531	40	40 HEB 600	9.071		0.0454 HEB 450	9.071	
28		0.021562		4	40	40 HEB 400	3.531	0.01766	36	36 HEB 600	18.143		0.0907 HEB 450	18.143	
29		0.026946	5	∞	45	45 HEB 400	5.650	0.02825	48	48 HEB 600	9.071		0.0454 HEB 450	9.071	
30		0.014579	∞	5	48	48 HEB 400	3.531	0.01766	45	45 HEB 600	14.514		0.0726 HEB 450	14.514	
31		0.016809	∞	9	99	56 HEB 400	3.531	0.01766	24	54 HEB 600	12.095		0.0605 HEB 450	12.095	
32		0.026083	9	∞	54	54 HEB 400	4.708	0.02354	26	26 HEB 600	9.071		HEB 450	9.071	
33		0.039199	3	∞	27		9.417	0.04708	32	32 HEB 600	9.071		0.0454 HEB 450	9.071	0.04536
34		0.017412		∞	63	HEB 400	4.036	0.02018	64	64 HEB 600	9.071		0.0454 HEB 450	9.071	
35		0.018243	∞	00	72	HEB 400	3.531	0.01766	72	72 HEB 600	9.071		0.0454 HEB 450	9.071	
36		0.017759	∞	7	64	. HEB 400	3.53125	0.01765625	63	HEB 600	10.36714286		0.0518 HFB 450	10 267	0.05100

ed - / dan	tep / - Berore cross section opt.	section opt.			ld. Family								
Nº of it	Columns Laterals	Tipe of profile	Length element (m)	Allowed max. Displacement columns (m)	Columns front/back	Tipe of profile	Length element (m)	Allowed max. Displacement columns (m)	max. Allow displacement	U_max_allow ?	Total number of elements	Total mass (kg)	Utilization ratio cross section (%)
1		12 HEB 450	22.630	0.0754	12	12 HEB 500	22.630	0.07543	0.02018	OK	106	240763.2335	89
2		10 HEB 450	22.630	0.0754	4	4 HEB 500	22.630	0.07543	0.04708	OK	45	133206.0864	88
3		10 HEB 450	22.630	0.0754	∞	8 HEB 500	22.630	0.07543	0.02825	ŎĶ	29	180918.7569	95
4		12 HEB 450	22.630	0.0754	9	6 HEB 500	22.630	0.07543	0.03531	ò	29	169194.2277	82
5		10 HEB 450	22.630	0.0754	10	10 HEB 500	22.630	0.07543	0.02354	NOT	78	204775.0922	114
9		14 HEB 450	22.630	0.0754	10	10 HEB 500	22.630	0.07543	0.02354	NOT	108	229038.7043	102
7		12 HEB 450	22.630	0.0754	∞	8 HEB 500	22.630	0.07543	0.02825	ŎĶ	80	193050.5629	78
80		16 HEB 450	22.630	0.0754	10	10 HEB 500	22.630	0.07543	0.02354	NOT	123	241170.5103	109
6		10 HEB 450	22.630	0.0754	12	12 HEB 500	22.630	0.07543	0.02018	NOT	88	228631.4274	106
10		8 HEB 450	22.630	0.0754	4	4 HEB 500	22.630	0.07543	0.04708	NOT	36	121074.2803	296
11		8 HEB 450	22.630	0.0754	10	10 HEB 500	22.630	0.07543	0.02354	NOT	63	192643.2861	419
12		10 HEB 450	22.630	0.0754	9	6 HEB 500	22.630	0.07543	0.03531	NOT	99	157062.4216	102
13		8 HEB 450	22.630	0.0754	12	12 HEB 500	22.630	0.07543	0.02018	NOT	72	216499.6214	363
14	1	16 HEB 450	22.630	0.0754	12	12 HEB 500	22.630	0.07543	0.02018	ŏ	140	265026.8456	82
15		8 HEB 450	22.630	0.0754	9	6 HEB 500	22.630	0.07543	0.03531	NOT	45	144930.6156	348
16		16 HEB 450	22.630	0.0754	4	4 HEB 500	22.630	0.07543	0.04708	ŏ	72	169601.5045	81
17	16	16 HEB 450	22.630	0.0754	80	8 HEB 500	22.630	0.07543	0.02825	ŏ	106	217314.175	93
18	ī	14 HEB 450	22.630	0.0754	12	12 HEB 500	22.630	0.07543	0.02018	ŏ	123	252895.0395	78
19		8 HEB 450	22.630	0.0754	8	8 HEB 500	22.630	0.07543	0.02825	NOT	54	168786.9508	349
20		14 HEB 450	22.630	0.0754	8	8 HEB 500	22.630	0.07543	0.02825	ŏ	93	205182.369	88
21		12 HEB 450	22.630	0.0754	10	10 HEB 500	22.630	0.07543	0.02354	ŏ	93	216906.8982	92
22		14 HEB 450	22.630	0.0754	4	4 HEB 500	22.630	0.07543	0.04708	ŏ	63	157469.6985	78
23		12 HEB 450	22.630	0.0754	4	4 HEB 500	22.630	0.07543	0.04708	ŏ	54	145337.8924	70
24		16 HEB 450	22.630	0.0754	9	6 HEB 500	22.630	0.07543	0.03531	OK	88	193457.8398	66
25	1	14 HEB 450	22.630	0.0754	9	6 HEB 500	22.630	0.07543	0.03531	OK	78	181326.0337	94
26		8 HEB 450	22.630	0.0754	14	14 HEB 500	22.630	0.07543	0.01766	NOT	81	240355.9566	395
27		18 HEB 450	22.630	0.0754	9	6 HEB 500	22.630	0.07543	0.03531	NOT	100	205589.6458	102
28		10 HEB 450	22.630	0.0754	14	14 HEB 500	22.630	0.07543	0.01766	NOT	100	252487.7627	122
29		18 HEB 450	22.630	0.0754	∞	8 HEB 500	22.630	0.07543	0.02825	ŏ	119	229445.9811	95
30		12 HEB 450	22.630	0.0754	14	14 HEB 500	22.630	0.07543	0.01766	ŏ	119	264619.5687	83
31		14 HEB 450	22.630	0.0754	14	14 HEB 500	22.630	0.07543	0.01766	ŏ	138	276751.3748	95
32		18 HEB 450	22.630	0.0754	10	10 HEB 500	22.630	0.07543	0.02354	NOT	138	253302.3164	111
33		18 HEB 450	22.630	0.0754	4	4 HEB 500	22.630	0.07543	0.04536	ŏ	81	181733.3106	98
34		18 HEB 450	22.630	0.0754	12	12 HEB 500	22.630	0.07543	0.02018	ŏ	157	277158.6516	98
35		18 HEB 450	22.630	0.0754	14	HEB 500	22.630	0.07543	0.01766	NOT	176	301014.9869	103
36	16	16 HEB 450	22.63	0.0754	14	14 HEB 500	22.630	0.07543	0.01766	NOT	157	288883.1808	101

Ne of it Stee 3 3 3 5 5 6 6 6 6 9 9 9 9 9 9 9 9 9 9 9 9 9 9	Steel class (m) (Structure) \$275		X - Number Y - N of Frames of F 3 3 5 5 6 6 6 7 7 7	of Frames of Frames	Beams Top - X direction		Length element (m)	Allowed max. Displacement beams (m)	Beams Top - Y		Allowed max.	Allowed max. Displacement	Tipe of profile	Allow Length element max.	Allowed max.
		0.01587 0.02955 0.02689 0.030515 0.030515 0.026896 0.026199 0.028964 0.027199 0.02557 0.02557 0.025257 0.025257 0.025257		2 4 4 5 4		Tipe of profile			Direction	Tipe of profile	(m) bea	beams (m)		(u)	Displacemen t beams (m)
2 m 4 15 10 12 10 10 10 10 10 10 10 10 10 10 10 10 10		0.029555 0.026989 0.026896 0.026896 0.028964 0.027199 0.02557 0.035501 0.035501 0.035501 0.035501 0.03524	w v 4 0 0 v 0 v w 0 4 v v	4 4 70 4	42 H	42 HEB 280, HEB 30	4.036	0.02018		40 HEB 550, HEB 400	14.514	0.0726	0.0726 HEB 360, HEB 300	14.514	0.07257
m 4 11 10 1		0.026989 0.030515 0.030515 0.026896 0.026896 0.028964 0.027199 0.02557 0.02557 0.02524 0.02524 0.02524	5 4 9 9 5 9 7 8 9 4 7 7	4 10 4	15 H	15 HEB 500, HEB 45	9.417	0.04708	16	16 HEB 1000	18.143	1 20000	HEB 1000	18.143	0.09071
4 5 9 7 8 6		0.02615 0.026896 0.026199 0.028964 0.027199 0.035501 0.027246 0.025224 0.025224 0.025224	40959789477	2 4	25 ve	25 variable	5.650	0.02825	24	24 variable	18.143	0.0907 variable	variable	18.143	0.09071
2 9 7 8 6		0.026896 0.026199 0.028864 0.027199 0.025257 0.035501 0.027246 0.025224 0.02524	0 0 5 0 7 8 0 4 7 7	4	24 v	24 variable	7.063	0.03531	25	25 variable	14.514	0.0726 variable	variable	14.514	0.07257
9		0.026199 0.028964 0.027199 0.025257 0.035501 0.027246 0.02814 0.02814	959789477		30 ve	30 variable	4.708	0.02354	28	28 variable	18.143	0.0907	variable	18.143	0.09071
7 8 6		0.028964 0.027199 0.025257 0.035501 0.027246 0.02524 0.02814	2 9 7 8 9 4 7 7	9	42 v	42 variable	4.708	0.02354	42	42 variable	12.095	0.0605	variable	12.095	0.06048
8 6		0.025257 0.035501 0.027246 0.025224 0.02514 0.02814	9 / 8 9 4 / /	2	30 v	30 variable	5.650	0.02825	30	30 variable	14.514	0.0726 variable	variable	14.514	0.07257
6		0.025257 0.035501 0.027246 0.025224 0.02814 0.024473	V & 9 4 V V	7	48 v	48 variable	4.708	0.02354	49	19 variable	10.367	0.0518 variable	variable	10.367	0.05184
		0.035501 0.027246 0.025224 0.02814 0.024473	8 9 4 7 7	4	35 v	35 variable	4.036	0.02018		32 variable	18.143	0.0907	variable	18.143	0.09071
10		0.025224 0.025224 0.02814 0.024473	9 4 7 7	co	12 vi	12 variable	9.417	0.04708	12	12 variable	24.190	0.1210 variable	variable	24.190	0.12095
11		0.025224 0.02814 0.024473	4 / /	ĸ	24 v.	24 variable	4.708	0.02354	21	21 variable	24.190	0.1210 variable	variable	24.190	0.12095
12		0.02814	^ ^	4	20 v	20 variable	7.063	0.03531	20	20 variable	18.143	0.0907	variable	18.143	0.09071
13		0.024473	7	ĸ	28 v.	28 variable	4.036	0.02018		24 variable	24.190	0.1210 variable	variable	24.190	0.12095
14		1013600		7	56 vi	56 variable	4.036	0.02018	26	56 variable	10.367	0.0518 variable	variable	10.367	0.05184
15		U.U36191	4	co	16 v	16 variable	7.063	0.03531	15	15 variable	24.190	0.1210 variable	variable	24.190	0.12095
16)	0.031259	ĸ	7	24 vi	24 variable	9.417	0.04708	28	28 variable	10.367	0.0518 variable	variable	10.367	0.05184
17	_	0.027195	2	7	40 v	40 variable	5.650	0.02825	42	42 variable	10.367	0.0518 variable	variable	10.367	0.05184
18		0.026105	7	9	49 v	49 variable	4.036	0.02018	•	48 variable	12.095	0.0605 variable	variable	12.095	0.06048
19	_	0.027773	2	m	20 v	20 variable	5.650	0.02825	18	18 variable	24.190	0.1210	variable	24.190	0.12095
20	_	0.027607	2	9	35 v	35 variable	5.650	0.02825		36 variable	12.095	0.0605 variable	variable	12.095	0.06048
21	_	0.027491	9	2	36 v	36 variable	4.708	0.02354		35 variable	14.514	0.0726 variable	variable	14.514	0.07257
22	_	0.028087	ĸ	9	21 v	21 variable	9.417	0.04708		24 variable	12.095	0.0605 variable	variable	12.095	0.06048
23	J	0.030144	ĸ	S	18 v.	18 variable	9.417	0.04708		20 variable	14.514	0.0726 variable	variable	14.514	0.07257
24	J	0.026454	4	7	32 v	32 variable	7.063	0.03531	35	35 variable	10.367	0.0518 variable	variable	10.367	0.05184
25		0.027604	4	9	28 v	28 variable	7.063	0.03531		30 variable	12.095	0.0605	variable	12.095	0.06048
56	_	0.028977	∞	c	32 v _e	32 variable	3.531	0.01766		27 variable	24.190	0.1210 variable	variable	24.190	0.12095
27		0.027226	4	∞	36 v	36 variable	7.063	0.03531	40	40 variable	9.071	0.0454 variable	variable	9.071	0.04536
28	_	0.025641	8	4	40 v	40 variable	3.531	0.01766		36 variable	18.143	0.0907 variable	variable	18.143	0.09071
59	_	0.026176	2	∞	45 v	45 variable	5.650	0.02825		48 variable	9.071	0.0454 variable	variable	9.071	0.04536
30	_	0.032354	8	2	48 v	48 variable	3.531	0.01766	45	45 variable	14.514	0.0726 variable	variable	14.514	0.07257
31	_	0.031274	8	9	56 v	56 variable	3.531	0.01766	54	54 variable	12.095	0.0605 variable	variable	12.095	0.06048
32	_	0.027399	9	∞	54 v.	54 variable	4.708	0.02354	99	56 variable	9.071	0.0454 variable	variable	9.071	0.04536
33	_	0.030796	æ	∞	27 v _c	27 variable	9.417	0.04708		32 variable	9.071	0.0454	variable	9.071	0.04536
34	_	0.024822	7	∞	63 v	63 variable	4.036	0.02018		64 variable	9.071	0.0454 variable	variable	9.071	0.04536
35	_	0.038684	8	∞	72 v.	72 variable	3.531	0.01766	72	72 variable	9.071	0.0454 variable	variable	9.071	0.04536
36)	0.027174	8	7	64 v.	64 variable	3.531	0.01766	63	63 variable	10.367	0.0518 variable	variable	10.367	0.05184

Step 9 - A	After cross se	ection o	- After cross section opt. With max. Displacement lim	isplaceme	ent limitation	ion							
				Allowed									
Nº of it	Columns Laterals	Tipe of profile	Length element (m)	max. Displacemen t columns	Columns front/back	Tipe of profile	Length element A (m)	Length element Allowed max. Displacement (m) columns (m)	max. Allow displacement	U_max < U_max_allow ?	Total number of elements	Total mass (kg)	Utilization ratio cross section (%)
,	JIT 90	0.74	000 00	(m)		770	000 00	CA3500	0,000	Carcagany	000	0004 4000	OL.
1 0	30 HE	30 HEB 220 HEB?	22.630	0.0754	121	12 HEB 220 HEB	22.630	0.07543	0.02018	VERDADERO	130	150815 165	7.9
ı m	30 var	30 variable	22.630	0.0754	- >0	8 variable	22.630	0.07543	0.02825	VERDADERO	87	197407.2028	96
4	36 var	36 variable	22.630	0.0754	> 9	6 variable	22.630	0.07543	0.03531	VERDADERO	91	151349.4117	98
2	30 var	30 variable	22.630	0.0754	10 v	10 variable	22.630	0.07543	0.02354	FALSO	98	218906.1107	114
9	42 var	42 variable	22.630	0.0754	10 v	10 variable	22.630	0.07543	0.02354	FALSO	136	139382.0882	111
7	36 vai	36 variable	22.630	0.0754	8	8 variable	22.630	0.07543	0.02825	FALSO	104	188061.4174	103
80	48 variable	riable	22.630	0.0754	10 v	10 variable	22.630	0.07543	0.02354	FALSO	155	134519.1627	116
6	30 vai	30 variable	22.630	0.0754	12 v	12 variable	22.630	0.07543	0.02018	FALSO	109	243248.9379	125
10	24 vai	24 variable	22.630	0.0754	4	4 variable	22.630	0.07543	0.04708	VERDADERO	52	155231.2059	75
11	24 vai	24 variable	22.630	0.0754	10 v	10 variable	22.630	0.07543	0.02354	FALSO	79	209011.1311	116
12	30 vai	30 variable	22.630	0.0754	9	6 variable	22.630	0.07543		VERDADERO	26	171122.5809	71
13	24 vai	24 variable	22.630	0.0754	12 v	12 variable	22.630	0.07543	0.02018	FALSO	88	228003.4945	139
14	48 vai	48 variable	22.630	0.0754	12 v	12 variable	22.630	0.07543	0.02018	FALSO	172	140902.7456	121
15	24 vai	24 variable	22.630	0.0754	9	6 variable	22.630	0.07543	0.03531	FALSO	61	182082.579	102
16	48 vai	48 variable	22.630	0.0754	4	4 variable	22.630	0.07543	0.04708	VERDADERO	104	134401.0309	99
17	48 vai	48 variable	22.630	0.0754	8	8 variable	22.630	0.07543	0.02825	VERDADERO	138	132736.0331	96
18	42 var	variable	22.630	0.0754	12 v	12 variable	22.630	0.07543	0.02018	FALSO	151	144768.465	129
19	24 vai	24 variable	22.630	0.0754	8	8 variable	22.630	0.07543	0.02825	VERDADERO	70	187616.3391	86
50	42 vai	42 variable	22.630	0.0754	8	8 variable	22.630	0.07543	0.02825	VERDADERO	121	125469.1604	86
21	36 vai	36 variable	22.630	0.0754	10 v	10 variable	22.630	0.07543	0.02354	FALSO	117	221430.7919	117
22	42 vai	42 variable	22.630	0.0754	4	4 variable	22.630	0.07543	0.04708	VERDADERO	91	118954.0111	09
23	36 vai	36 variable	22.630	0.0754	4 v	4 variable	22.630	0.07543		VERDADERO	78	143012.0607	64
24	48 vai	48 variable	22.630	0.0754	9 v	6 variable	22.630	0.07543	0.03531	VERDADERO	121	124680.5959	75
25	42 var	variable	22.630	0.0754	9	6 variable	22.630	0.07543		VERDADERO	106	105619.9322	78
56	24 vai	24 variable	22.630	0.0754	14 v	14 variable	22.630	0.07543		FALSO	97	252232.0768	164
27	54 vai	54 variable	22.630	0.0754	9	6 variable	22.630	0.07543		VERDADERO	136	114680.7465	77
28	30 vai	riable	22.630	0.0754	14 v	14 variable	22.630	0.07543		FALSO	120	267762.6405	145
29	54 vai	54 variable	22.630	0.0754	8	8 variable	22.630	0.07543	0.02825	VERDADERO	155	124971.0919	93
30	36 vai	36 variable	22.630	0.0754	14 v	14 variable	22.630	0.07543		FALSO	143	192348.5354	183
31	42 var	variable	22.630	0.0754	14 v	14 variable	22.630	0.07543		FALSO	166	147153.763	177
32	54 vai	54 variable	22.630	0.0754	10 v	10 variable	22.630	0.07543	0.02354	FALSO	174	128198.752	116
33	54 vai	54 variable	22.630	0.0754	4	4 variable	22.630	0.07543	0.04536	VERDADERO	117	141013.5563	89
34	54 vai	54 variable	22.630	0.0754	12 v	12 variable	22.630	0.07543	0.02018	FALSO	193	126144.5894	123
35	54 va	54 variable	22.630	0.0754	14 v	14 variable	22.630	0.07543	0.01766	FALSO	212	100381.5571	219
36	48 var	variable	22.630	0.0754	14 v	14 variable	22.630	0.07543	0.01766	FALSO	189	144637.7197	154

		1 st	1 st optimization loop - Model A	Model A		
	Before	Before cross section optimization	ization	After	After cross section optimization	sation
Nº of iteration	Total mass (kg)	Utilization ratio cross section (%)	Utilization ratio cross section (%)	Total mass (kg)	Utilization ratio cross section (%)	Utilization ratio cross section (%)
1	240763.2335	89	89	292264.1999	79	79
2	133206.0864	88	89	150815.165	63	63
3	180918.7569	95	95	197407.2028	96	96
4	169194.2277	85	85	151349.4117	86	86
2	204775		0	218906	114	0
9	229039	102	0	139382	111	0
7	193050.5629	78	78	188061		0
∞	241171	109	0	134519		0
б	228631	106	0	243249	125	0
10	121074	296	0	155231.2059	75	75
11	192643	419	0	209011	116	0
12	157062	102	0	171122.5809	7.1	71
13	216500	363	0	228003	139	0
14	265026.8456	85	85	140903	121	0
15	144931	348	0	182083	102	0
16	169601.5045	81	81	134401.0309	99	99
17	217314.175	93	93	132736.0331	96	96
18	252895.0395	78		144768	129	0
19	168787	349	0	187616.3391	86	86
20	205182.369	88	88	125469.1604	98	86
21	216906.8982	92	92	221431	117	0
22	157469.6985	78	78	118954.0111	09	09
23	145337.8924	70	70	143012.0607	64	64
24	193457.8398	66	66	124680.5959	75	75
25	181326.0337	94	94	105619.9322	78	78
26	240356	395	0	252232	164	0
27	205590	102	0	114680.7465	77	77
28	252488	122	0	267763	145	0
29	229445.9811	95	95	124971.0919	93	93
30	264619.5687	83	83	192349	183	0
31	276751.3748	95	95	147154	177	0
32	253302	111	0	128199	116	0
33	181733.3106	86	86	141013.5563	89	89
34	277158.6516	86	86	126145	123	0
35	301015	103	0	100382		0
36	288883	101	0	144638	154	0

	COMPARISON	SISON	
Reducction of mass (kg)	Reducction of mass (%) respect the initial value	Reducction of mass (%) respect the initial value	Nº of iteration
-51500.96649	-21	0	1
-17609.07866	-13	0	2
-16488.44586	6-	0	8
17844.81595	11	11	4
-14131.01859	2-	0	r.
89656.61603	39	39	9
4989.145529	m	c	7
106651.3476	44	44	∞
-14617.51054	9-	0	6
-34156.92555	-28	0	10
-16367.84495	8-	0	11
-14060.15926	6-	0	12
-11503.87316	-5	0	13
124124.0999	47	47	14
-37151.96342	-26	0	15
35200.47364	21	21	16
84578.14199	39	39	17
108126.5745	43	43	18
-18829.38821	-11	0	19
79713.20863	39	39	20
-4523.893718	-2	0	21
38515.68735	24	24	22
2325.831755	2	2	23
68777.24387	36	36	24
75706.10153	42	42	25
-11876.12021	-5	0	26
90908.89938	44	44	27
-15274.87786	9-	0	28
104474.8892	46	46	29
72271.03333	27	27	30
129597.6117	47	47	31
125103.5643	49	49	32
40719.75424	22	22	33
151014.0622	54	54	34
200633.4298	29	29	35
144245.4611	20	50	36

APPENDIX D - 1 : MODEL A

MODEL A - Comparison Loop 1 & Loop 2 (36 variables)

				1 st optimization loop - Model A	ion loop - I	Model A				2	nd optimiza	2nd optimization loop - Model A	odel A			before	after	before 1st loop
<u> </u>	Genes opt. Befor	Before cross section optimization	ptimization	*	After cross section optimization	ptimization		Be	efore cross section optimization	timization		After cross section optimization	nization			1st/2nd loop	1st/2nd loop	after 2nd loop
Nº of iteration	x Y To	Total mass (kg)	Utilization ratio cross section (%)	Utilization ratio cross section (%)	Total mass (kg)	Utilization ratio Uti	Utilization ratio Utilization ratio cross Reduction mass cross section (%) section (%) (%)	uction mass (%)	Total mass (kg)	Utilization ratio cross Utilization ratio cross section (%)	Itilization ratio cross section (%)	Total mass (kg) Uti	Utilization ratio cross Utilization ratio cross section (%)	tilization ratio cross section (%)	Reduction mass (%)	Increment mass (%)	Ē	Reduction mass
-	7 5	240763.2335		68,34477876	292264.1999	78.64778761	78,64778761	ľ	291626.693	79.89168142	79.89168142	211214,2109	96.64707965	96.64707965	28	17	÷	
2	3 4	133206.0864		88.56212389	150815.165	62.77168142	62.77168142	-13	150160.5729	96.81769912	96.81769912	145250.9902	98.79716814	98.79716814	m	11	, ,	
e	5 4	180918.7569	95.27787611	95.27787611	197407.2028	95.53628319	95.53628319	ę,	214827.7299	99.83362832	99.83362832	221259.2628	96.30088496	96.30088496	ή	16	11	-22
4	4 5	169194.2277	84.7660177	84.7660177	151349.4117	86.41415929	86.41415929	11	194625.9575	87.26088496	87.26088496	145814.0412	93.51646018	93.51646018	25	13		
2	6 4	204775.0922	114.1040708	0	218906.1107	114.2484956	0	-2	247161.3085	119.7833628	0	295579.3666	92.64	92.64	-20	17	26	4
9	9 9	229038.7043	101.5858407	0	139382.0882	111.2877876	0	39	271424.9206	136.0056637	0	246361.9074	103.0088496	0	6	16	43	
7	5 5	193050.5629	77.95752212	77.95752212	188061.4174	102.5274336	0	8	226959.536	109.0477876	0	180917.1655	95.66017699	95.66017699	20	15		
00	2 9	241170.5103	109.2785841	0	134519.1627	115.5355752	0	44	283556.7266	139.9773451	0	272039.0849	112.6300885	0	4	15	51	-13
6	7 4	228631.4274	106.1224779	0	243248.9379	125.1674336	0	φ	279494.887	93.96106195	93.96106195	267834.1788	106.499115	0	4	18	6	- ti-
10	3	121074.2803	296,4021239	0	155231.2059	75.40035398	75.40035398	-28	138028.7668	231.9334513	0	152588.2407	98.21946903	98.21946903	-11	12		'7
11	6 3	192643.2861	418.9338053	0	209011.1311	115.7352212	0	οp	235029.5024	335.0952212	0	283563.6966	93.32389381	93.32389381	-21	18		4
12	4 4	157062.4216	102.3348673	0	171122.5809	71.43079646	71.43079646	φ	182494.1514	79.90088496	79.90088496	146805.2857	97.87752212	97.87752212	20	14		
13	7 3	216499.6214	363.1525664	0	228003.4945	139.4548673	0	ι'n	267363.0809	297.8307965	0	276627.5889	95.79469027	95.79469027	ņ	19	18	-58
14	7 7	265026.8456	84.65415929	84.65415929	140902.7456	121.2821239	0	47	315890.3051	90.2739823	90.2739823	214822.6275	92.5040708	92.5040708	32	16		19
15	4 3	144930.6156		0	182082.579	102.4877876	0	-26	170362.3454	223.5440708	0	185422.5825	98.32778761	98.32778761	6-	15		-58
16	3 7	169601.5045		81.10938053	134401.0309	66.39079646	66.39079646	21	186555.991	112.8042478	0	182253.2463	98.70159292	98.70159292	2	6		
17	5 7	217314.175	92.80707965	92.80707965	132736.0331	96.26548673	96.26548673	39	251223.1481	116.640708	0	224356.2317	99.50442478	99.50442478	11	13		
18	9 /	252895.0395		78.15716814	144768.465	129.3699115	0	43	303758.4991	86.30442478	86.30442478	195704.8123	94.02548673	94.02548673	36	17		
19	5 3	168786.9508		0	187616.3391	98.31150442	98.31150442	-11	202695.9239	279.3451327	0	222679.3553	99.10088496	99.10088496	-10	17		-32
20	5 6	205182.369		88.12389381	125469.1604	97.72389381	97.72389381	39	239091.342	113.5575221	0	201933.1763	92.38230088	92.38230088	16	14		
21	9	216906.8982	-	91.90088496	221430.7919	116.7759292	0	-5	259293.1145	130.8955752	0	249076.7814	91.48884956	91.48884956	4	16	11	-15
22	3 6	157469.6985		77.74088496	118954.0111	59.65380531	59.65380531	24	174424.185	111.1454867	0	151646.9018	97.88814159	97.88814159	13	10		
23	3 2	145337.8924		69.80176991	143012.0607	64.02265487	64.02265487	2	162292.3789	107.2672566	0	127783.7653	100.6343363	0	21	10		
24	4 7	193457.8398	98.76389381	98.76389381	124680.5959	74.9139823	74.9139823	36	218889.5696	93.31823009	93.31823009	158924.7663	100.9755752	0	27	12		18
25	4 6	181326.0337	6	94,35752212	105619.9322	78.17061947	78.17061947	42	206757.7635	90.66477876	90.66477876	142844.8782	97.77274336	97.77274336	31	12	56	
56	e 00	240355.9566		0	252232.0768	164.1175221	0	ι'n	299696.6594	236.9982301	0	315958.682	94.81061947	94.81061947	ς	8	20	
27	4 8	205589.6458		0	114680.7465	77.10017699	77.10017699	44	231021.3756	93.53628319	93.53628319	178271.6092	95.95752212	95.95752212	23	11		
28	8 4	252487.7627	122.1210619	0	267762.6405	145.2233628	0	φ	311828.4655	81.76707965	81.76707965	269214.6779	110.2782301	0	14	19		-
59	2 8	229445.9811	95.3840708	95.3840708	124971.0919	92.65840708	92.65840708	46	263354.9541	116.9061947	0	238017.8861	100.3504425	0	10	13		
30	8 5	264619.5687		82.57132743	192348.5354	183.2438938	0	27	323960.2715	53.37486726	53.37486726	249326.1138	85.04637168	85.04637168	23	18		
31	9 8	276751.3748	0,	95.20141593	147153.763	177.1270796	0	47	336092.0776	58.56283186	58.56283186	198259.6977	109.8477876	0	41	18		
32	8 9	253302.3164	П	0	128198.752	116.3851327	0	49	295688.5327	140.2916814	0	277085.1594	116.4276106	0	9	14	54	
33	00 I	181733.3106		86.4246934	141013.5563	67.89802949	67.89802949	22	198687.7971	117.4744385	0	202327.2142	100.3478021	0	-5	6		
34	7 8	277158.6516		86.28955752	126144.5894	123.0116814	0	54	328022.1112	91.15610619	91.15610619	219403.073	92.14230088	92.14230088	33	16	43	
32	∞	301014.9869		0	100381.5571	219.0952212	0	29	360355.6897	63.19575221	63.19575221	209625.5505	96.67964602	96.67964602	42	16	5	98
36	8 7	288883.1808	100.5819469	0	144637.7197	153.9058407	0	20	348223.8836	61.90442478	61.90442478	196916.2513	99.23964602	99.23964602	43	17	2	,

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MODEL B	•		-			200														
	1	Optimization process	orocess (Kesuic	(Results 56 Variables)	הוכטו														
Paramete	arameters considered:	Structural material Cladding material Load case	Steel s275 Glass 4 & 5																	
Canopy m Height	measurements - 1st mode m	_	direct	_ •	extension (length) m extension (roof) - high eave m	n) m high eave m		∞ ∞		Appartment Height	measurements m	22.63 Z d	lirect		Jnit: Appartmen Jeight n	t - Canopy - Appartment n		Z direct		
Lenght Width	EΕ	72.57 Y dirct 28.25 X direct	dirct direct	40	extension (roof) - low eave Desnivel angle	- low eave m		3 5 7.06		Lenght Width	Lenght m Width m	72.57 Y dirct 12.1 X direct	dirct direct		Lenght m Width m	c c	52.45	72.57 Y dirct 52.45 X direct		
	B - 1:Step	7 - Before cross	section opt		ld. Family			9	ld. Family			ā	ld. Family							
Nº it	Steel class	nent re)	X - Number Y of Frames of	Y - Number of Frames	Beams Top - X direction	Tipe of profile	Length A	ax.	Beams Top - Y Direction Extension	Tipe of profile	Length Allor element (m) Disp	ax.	Beams Top - Y Direction central part top	Tipe of profile	Length A	Allowed max. Displacement	Beams Top - Y Direction central sides	Tipe of profile e	Length Al element (m) Di	Allowed max. Displacement
1	s275	0.034992	6	9	24 1	HEB 400	13.450	beams (m) 0.06725	4	HEB 600		beams (m) 0.0400	24 H	HEB	12.095	beams (m) 0.060475	12 H	HEB	be 12.095	beams (m) 0.060475
2		0.019008	4	Ŋ	28		10.088	0.05044	LO I		8.000	0.0400	25		14.514	0.07257	10		14.514	0.07257
w 4		0.01469	9 ~	9 6	42		5.764	0.03363	7 8		8:000	0.0400	35		14.514	0.07257	10	~	14.514	0.07257 0.040316667
10 00		0.015462	2 7	б 1/1	55		8.070	0.04035	9 80		8.000	0.0400	42. 44		8.063333333	0.040316667	18	~	8.063333333	0.040316667
7		0.023315	. ~	4	42		5.764	0.02882	0 00		8:000	0.0400	32		18.1425	0.0907125	, ee		18.1425	0.0907125
x 0	OI	0.015282	4 10	9	40		8.070	0.05044	9		8:000	0.0400	36		12.095	0.04535625	12		12.095	0.04535625
10		0.011579	7	7	63		5.764	0.02882	80		8.000	0.0400	95		10.36714286	0.051835714	14		10.36714286	0.051835714
11		0.034992	m o	00 ц	30		13,450	0.06725	4 0		8.000	0.0400	32		9.07125	0.04535625	16		9.07125	0.04535625
13		0.01537	0 10	0 1	45		8.070	0.04035			8:000	0.0400	4 4		10.36714286	0.051835714	14		10.36714286	0.051835714
14		0.019207	4	9	32		10.088	0.05044	ı		8.000	0.0400	30		12.095	0.060475	12		12.095	0.060475
15		0.012458	9 1	9 0	48		6.725	0.03363	7		8.000	0.0400	42		12.095	0.060475	12		12.095	0.060475
17		0.035985	. 4	0 4	24		10.088	0.05044	o ro		8.000	0.0400	20		18.1425	0.0907125	8		18.1425	0.0907125
18		0.030009	ı,	4	30		8.070	0.04035	9		8.000	0.0400	24		18.1425	0.0907125	∞ :		18.1425	0.0907125
20		0.034992	m on	ס נח	333		13.450	0.06725	10		8:000	0.0400	36		8.063333333	0.040316667	8 9	~	8.063333333	0.040316667
21		0.015116	. 10	ı ın	35		8.070	0.04035	9		8.000	0.0400	30		14.514	0.07257	10		14.514	0.07257
22		0.011515	7	9 0	26		5.764	0.02882	ω ς		8.000	0.0400	48		12.095	0.060475	12		12.095	0.060475
24		0.023481	ი თ	y 4	54		4.483	0.02242	10 10		8:000	0.0400	05 4 04		8.005333333 18.1425	0.0907125	g 80	~	8.003333333 18.1425	0.0907125
25		0.030003	9	4	36		6.725	0.03363	7		8.000	0.0400	28		18.1425	0.0907125	00		18.1425	0.0907125
26		0.012541	9 **	7	54		6.725	0.03363	7 4		8.000	0.0400	49		10.36714286	0.051835714	4 5		10.36714286	0.051835714
28		0.012358	. 6	9	72		4.483	0.02242	10		8.000	0.0400	09		12.095	0.060475	12		12.095	0.060475
29		0.012459	00 Y	7	72		5.044	0.02522	0 1		8.000	0.0400	63		10.36714286	0.051835714	14		10.36714286	0.051835714
31		0.012505	0 00	0 00	8 80		5.044	0.02522	, 6		8:000	0.0400	27		9.07125	0.04535625	16		9.07125	0.04535625
32		0.012628	9 •	0 0	99		6.725	0.03363	7		8.000	0.0400	63		8.063333333	0.040316667	18		8.063333333	0.040316667
34		0.012526	4 00	ח ס	\$ 8		5.044	0.02522	0		8:000	0.0400	81.8		8.063333333	0.040316667	18 18	~ ~	8.063333333	0.040316667 0.040316667
35		0.0388	es i	4	18		13.450	0.06725	4		8.000	0.0400	16		18.1425	0.0907125	∞ !		18.1425	0.0907125
36		0.015432	ın m	00 1	50		8.070	0.04035	9 4		8.000	0.0400	8 8		9.07125	0.04535625	16		9.07125	0.04535625
38		0.012004	6		81		4.483	0.02242	10		8.000	0.0400	02		10.36714286	0.051835714	14		10.36714286	0.051835714
39		0.011762	തം	00 4	90		4.483	0.02242	10		8.000	0.0400	8 5		9.07125	0.04535625	16		9.07125	0.04535625
4		0.011828	10	0 00	100		4.035	0.02522	11		8.000	0.0400	¥ 88		9.07125	0.04535625	16		9.07125	0.04535625
42		0.019323	4	7	36		10.088	0.05044	r.		8.000	0.0400	35		10.36714286	0.051835714	14		10.36714286	0.051835714
43		0.013008	10	2 4	70		4.035	0.02018	11		8.000	0.0400	35		14.514	0.07257	10		14.514	0.07257
45		0.02348	10	4	9 9		4.035	0.02018	11		8.000	0.0400	8 4		18.1425	0.0907125	0 00		18.1425	0.0907125
46		0.01254	∞ ç	10	96		5.044	0.02522	σ,		8.000	0.0400	8 8		7.257	0.036285	20		7.257	0.036285
44 48		0.011/44	9 9	n w	80		4.035	0.02018	1 11		8.000	0.0400	£ 99		8.U03333333 12.095	0.040316667	12	~	8.003333333 12.095	0.060475
49		0.012071	10	7	90		4.035	0.02018	11		8.000	0.0400	77		10.36714286	0.051835714	14		10.36714286	0.051835714
S 12		0.011656	6 6	10	108		4.483	0.02242	10		8:000	0.0400	100		7.257	0.036285	20		7.257	0.036285
52		0.01172	10	10	120		4.035	0.02018	11		8.000	0.0400	110		7.257	0.036285	20		7.257	0.036285
53		0.012645	9 1	10	72		6.725	0.03363	7		8.000	0.0400	2 8		7.257	0.036285	20		7.257	0.036285
, 13		0.01942	n 4	10	84		10.088	0.05044	ם נח		8.000	0.0400	8 8		7.257	0.036285	20		7.257	0.036285
56		0.034992	3	10	36		13.450	0.06725	4		8.000	0.0400	40		7.257	0.036285	20		7.257	0.036285

APPENDIX D - 2 : MODEL B - 1

MODEL B - 1 Optimization process (Results 56 variables)

Parameters considered Structural mater Steel 6275

Load case 48 5

B - 1 : St	: Step 7 - Befc	7 - Before cross sect	tion opt.									-		ld. Family		
Nº it	Beams back facade	Tipe of profile	Length element (m)	Allowed max. Displacement beams (m)	Beams front facade	Tipe of profile	Length element (m)	Allowed max. Displacement beams (m)	Columns Laterals left short	Tipe of profile e	Length slement A (m) D	Length element Allowed max. (m) Displacement beams (m)	Columns Laterals rigth short	Tipe of profile	Length element (m)	Allowed max. Displacement columns (m)
1	,	3 HEB	13.45		c	HEB	13.45	0.06725	9	HEB	00	0.026666667		HEB 450	3.000	0.0100
2	, ,	4 '	10.0875	0	4 (10.0875	0.0504375	ı,		00 (0.02666667	LO I		3.000	0.0100
ηV	٠,١	۰ ۵	6.725 5.764285714	0.033625	9 1		5.764285714	0.033625	nσ		00 00	0.025555557	Ωσ		3.000	0.0100
2	. 41	. 10	8.07		· R		8.07	0.04035	. 6		0 00	0.026666667	0		3.000	0.0100
9	.`	7	5.764285714		7		5.764285714	0.028821429	5		∞	0.026666667	S		3.000	0.0100
7		_	5.764285714	0.028821429	7		5.764285714	0.028821429	4		∞ •	0.026666667	4		3.000	0.0100
0	01		8.07	0.04035	5		8.07	0.04035	9		o ∞	0.026666667	9		3.000	0.0100
10	,,	7	5.764285714	0.02	7		5.764285714	0.028821429	7		∞	0.026666667	7		3.000	0.0100
11	***	3	13.45		3		13.45	0.06725	00		00	0.026666667	80		3.000	0.0100
12	~ !	ω.	5.04375	0.0	00 L		5.04375	0.02521875	1 12		∞ 0	0.026666667	1 02		3.000	0.0100
13	, <	o -	10.0875	0.04035	v 4		8.07	0.04035	, 9		xo ox	0.025555557	, 4		3.000	0.0100
15	. •	+ 10	6.725		† 9		6.725	0.033625	9		0 00	0.026666667	9		3.000	0.0100
16	,	7	5.764285714	0.0	7		5.764285714	0.028821429	00		00	0.02666667	∞		3.000	0.0100
17		4	10.0875	ö	4		10.0875	0.0504375	4		00	0.026666667	4		3.000	0.0100
18		υ.	8.07		S		8.07	0.04035	4		00 (0.02666667	4		3.000	0.0100
19		m	13.45	0.06725	m a		13.45	0.06725	D II		00 O	0.026666667	பை		3.000	0.0100
21			4.465333333		ח נה		4:403333333	0.022418887	חני		0 00	0.026666667	nır		3.000	0.0100
22		. ~	5.764285714	0.02	^		5.764285714	0.028821429	9		0 00	0.026666667	9		3.000	0.0100
23		. 6	4.48333333		6		4.483333333	0.022416667	6		00	0.026666667	6		3.000	0.0100
24		6	4.483333333	0.0	6		4.483333333	0.022416667	4		∞	0.026666667	4		3.000	0.0100
25		9 '	6.725		9		6.725	0.033625	4 1		00 (0.026666667	4		3.000	0.0100
26		9 .	6.725	0	9 6		6.725	0.033625	,		00 0	0.026666667			3.000	0.0100
77		n ~	13.45 4.48333333	0.027416667	nσ		4.48333333	0.022416667	א ע		xo oc	0.026666667	n u		3.000	0.0100
29			5.04375		. 00		5.04375	0.02521875	7		000	0.026666667	7		3.000	0.0100
30		S	6.725		9		6.725	0.033625	80		00	0.026666667	00		3.000	0.0100
31		00	5.04375	0	00		5.04375	0.02521875	00		00	0.026666667	00		3.000	0.0100
32		φ.	6.725		9 •		6.725	0.033625	6		00 0	0.026666667	6		3.000	0.0100
33		4 ~	10.0875	0.0504375	4 0		E 04375	0.0504375	no		xo ox	0.026666657	no		3.000	0.0100
3.5		۰ ~	13.45		o m		13.45	0.06725	v 4		0 00	0.026666667	v 4		3.000	0.0100
36		10	8.07		20		8.07	0.04035	- 00		00	0.026666667	- 00		3.000	0.0100
37		3	13.45		8		13.45	0.06725	7		00	0.026666667	7		3.000	0.0100
38		6	4.483333333		6		4.483333333	0.022416667	7		00	0.026666667	7		3.000	0.0100
39		n ~	4.483333333	0.022416667	nα		4.483333333	0.022416667	90 U		00 O	0.026666667	00 U		3.000	0.0100
41	•		4.035		10		4.035	0.020175	000		000	0.026666667	0 00		3.000	0.0100
42		4	10.0875	_	4		10.0875	0.0504375	7		00	0.026666667	7		3.000	0.0100
43		0	4.035		10		4.035	0.020175	S.		00	0.026666667	5		3.000	0.0100
44	•	∞ <u>c</u>	5.04375	0.02521875	∞ 5		5.04375	0.02521875	4 4		00 0	0.026666667	4 4		3.000	0.0100
4.5		. ~	4,033	-	۹		4.033	0.0201/3	+ 5		0 00	0.025066667	4 5		3.000	0.0100
47			4.035		10		4.035	0.020175	6		0 00	0.026666667	9 6		3.000	0.0100
48	•	0	4.035	0.020175	10		4.035	0.020175	9		00	0.02666667	9		3.000	0.0100
49	•	0	4.035		10		4.035	0.020175	7		∞	0.026666667	7		3.000	0.0100
20		o '	4.483333333		ை 1		4.48333333	0.022416667	10		00 0	0.026666667	10		3.000	0.0100
51	•		5.764285714	0.028821429	7 01		5.764285714	0.028821429	10		oo o	0.026666667	10		3.000	0.0100
53		o 10	6.725		9		6.725	0.033625	10		0 00	0.02666667	9 01		3.000	0.0100
54		. 10	8.07		. 20		8.07	0.04035	10		00	0.026666667	10		3.000	0.0100
52		4	10.0875	Ö	4		10.0875	0.0504375	10		00	0.026666667	10		3.000	0.0100
56			13.45	0.06725	m		13.45	0.06725	10		œ	0.026666667	10		3.000	0.0100

APPENDIX D - 2: MODEL B - 1

MODEL B - 1 Optimization process (Results 56 variables)

Parameters considered: Storetonal S

Column C	io.	: Step 7 - Before cross section	ore cros		opt.				•				•		Minimal Mass [kg] =	ass [kg] =	205204.1627	
151199 0.0111 1.0 0.0111 1	Colum	ins front	Tipe of profile	Length element (m)	Allowed max. Displacement columns (m)	Columns back			Allowed max. Displacement columns (m)	Columns Front Ablock	Tipe of profile	Length element (m)	Allowed max. Displacement columns (m)	max. Allow displacement	U_max < U_max_allow ?	Total number of elements	Total mass (kg)	Utilization ratio cross section (%)
15.11 10.00 10.00 1.00		∞ :		15.3150		∞ ;	HEB 500	15.315	0.05105	2		22.63	0.075433333	0.04000) OK	100	225097.859	87
1,11,119 0,055,11 1, 10, 11,119 1, 10, 11,119 1, 10, 11,119 1, 10, 11,119 1,		5 4		15.3150	0.0511			15.315	0.05105	w 4		22.63	0.075433333	0.04000	šδ	109	241099.9416 292996.2937	8 4 4 8 44
131190 0.05511 1.0		16		15.3150	0.0511			15.315	0.05105	5		22.63	0.075433333	0.02882		244	358731.539	40
131190 0.0551 1.5		12		15.3150	0.0511			15.315	0.05105	9		22.63	0.075433333	0.04000		191	306835.6612	39
131190 10561 100 11319 1010 11319 1010 11319 1010 11319 1010 11319 1010 11319 11319 1010 11319 113		16		15.3150	0.0511			15.315	0.05105	V 80		22.63	0.075433333	0.02882		170	318944.1464 308997.2982	81
153190 106511 12 15315 105109 110 1268 104953338 104009 0.K 1519 151319 105119 113119 105119 131319 131319 105119 1313		10		15.3150	0.0511			15.315	0.05105	6		22.63	0.075433333	0.04000		154	270940.4861	48
1,13,19, 0.007		12		15.3150	0.0511			15.315	0.05105	10		22.63	0.075433333	0.04000		150	276995.1167	38
1,131.99 0.0011 1.0 1.0 1.5 1.5 0.00500 1.1 2.5 0.00500		16		15.3150	0.0511			15.315	0.05105	11		22.63	0.075433333	0.02882		212	338837.8427	40
153199 0.0011 1.0 1.0 1.5		× 82		15.3150	0.0511			15.315	0.05105	13		22.63	0.075433333	0.04000		195	344891.8912	8/
153159 0.0511 1.0 1.5 1.0 1.5 1.0 1.5 1.0 1.5 1.0 1.5 1.0 1.5 1.0 1.5 1.0 1.5 1.0 1.5		12		15.3150	0.0511			15.315	0.05105	14		22.63	0.075433333	0.04000		169	286941.9649	38
153199 0.0551 1 1 15319 0.05510 1 1 15319 0.05510 1 1 1 15319 0.05510 1 1 1 1 1 1 1 1 1		10		15.3150	0.0511			15.315	0.05105	15		22.63	0.075433333	0.04000		134	251046.7897	48
131190 10511 10511 10511 131190 10510 11 12511 10510 11 12511 10510 11 12511 10510 11 12511 10510 11 12511 1		14		15.3150	0.0511			15.315	0.05105	16		22.63	0.075433333	0.03363		177	302943.1419	37
133150 105211 105213 1		16		15.3150	0.0511			15.315	0.05105	17		22.63	0.075433333	0.02882		237	348784.6908	40
1,2,1,2,2,2,3,3,3,4,3,4,3,4,4,4,4,4,4,4,4,4,4		9 9		15.3150	0.0511			15.315	0.05105	18		22.63	0.075433333	0.04000		111	231153.0934	90
1,13,150 1,00,111 1,00 1,13,150 1,00,101 1,10 1,13,150 1,00,13,133 1,00,13,133 1,00,10,10 1,13,130 1,13,130 1,13,130 1,13,130 1,13,130 1,13,130 1,13,130 1,13,130 1,13,13,130 1,13,13,130 1,13,13,130 1,13,13,130 1,13,13,130 1,13,13,130 1,13,13,130 1,13,13,130 1,13,13,130 1,13,13,130 1,13,13,130 1,13,13,130 1,13,13,130 1,13,13,13,130 1,13,13,13,130 1,13,13,13,130 1,13,13,13,13,130 1,13,13,13,130 1,13,13,13,130 1,13,13,13,130 1,13,13,13,13,13,13,13,13,13,13,13,13,13		71 %		15.3150	0.0511			15.315	0.05105	19		22.63	0.075433333	0.04000		151	25/101.4204	78
151500 0.0531 1.0		70		15.3150	0.0511			15.315	0.05105	21		22.63	0.07543333	0.02242		222	370839.5641	58
15.3156 0.0561 16 15.315 0.055105 2.5 2.5 0.07543333 0.072242 0.05 2.5 0.072423 0.05 0		12		15.3150	0.0511			15.315	0.05105	22		22.63	0.075433333	0.04000		147	267048.2686	38
15.3150 0.0561 1.0 1.3 1.0 0.055105 2.4 2.268 0.07943333 0.02224 ACK 2.268 ACK 2.268 0.07943333 0.02224 ACK 2.268 ACK 2.26		16		15.3150	0.0511			15.315	0.05105	23		22.63	0.075433333	0.02882		205	328890.9945	40
153350 0.0511 2.0 15335 0.05511 2.0 153350 0.0511 4.0 15335 0.0510 2.0 0.0513333 0.05234 7.4 0.0510 2.0 0.0513333 0.05234 7.4 0.0510 1.3 0.0510 2.0 0.0513333 0.05333 0.05333 0.05333 0.05333 0.05333 0.05030 0.0 0.0511 1.3 0.0510 1.3 0.0510 1.3 0.0510 1.3 0.0520 2.0 0.05233 0.0520 0.0		70		15.3150	0.0511			15.315	0.05105	24		22.63	0.075433333	0.02242		317	410626.9568	52
15,3150 0,050 1		50		15.3150	0.0511			15.315	0.05105	25		22.63	0.075433333	0.02242		203	360892.716	105
1,23,150 0,0051 1, 1, 1, 1, 1, 1, 1, 1, 1, 1, 1, 1, 1,		4 5		15.3150	0.0511			15.315	0.05105	26		22.63	0.075433333	0.03363		153	283049.4456	89
15.3150 0.0511 2.0 0.5510 0.0543333 0.0243333 0.024243 0.02520 0.02500 2.2 0.07543333 0.02520 0.07543333 0.02520 0.07543333 0.02520 0.075403433 0.02520 0.075403433 0.02520 0.075403433 0.02520 0.075403433 0.075403433 0.075403433 0.075403433 0.075403433 0.075403433 0.075403433 0.075403433 0.0754034333 0.075403433 0.075403433 0.075403433 0.075403433 0.075403433 0.075403433 0.075403433 0.075403433 0.075403433 0.075403433 0.0754034333 0.0754034333 0.075403433 0.075403433 0.07540343333 0.07540343333 0.0754034333 0.075		\$ ∞		15.3150	0.0511			15.315	0.05105	28		22.63	0.075433333	0.04000		115	215151.0108	87
15.319 0.06511 18 15.315 0.055108 31 2.263 0.075433333 0.02622 0.K 2.244 15.315 0.0511 14 15.315 0.05106 33 2.268 0.07543333 0.03232 0.K 2.245 15.315 0.0511 14 15.315 0.05106 34 2.268 0.07543333 0.03232 0.K 2.245 15.315 0.0511 18 15.315 0.05106 34 2.268 0.07543333 0.03020 0.K 2.245 15.315 0.0511 18 15.315 0.05106 34 2.268 0.07543333 0.03020 0.K 2.245 15.315 0.0511 18 15.315 0.05106 37 2.268 0.07543333 0.00000 0.K 1.11 15.315 0.0511 18 15.315 0.05106 37 2.268 0.07543333 0.00000 0.K 1.11 15.315 0.0511 2.0 1.24 <td< td=""><td></td><td>20</td><td></td><td>15.3150</td><td>0.0511</td><td></td><td></td><td>15.315</td><td>0.05105</td><td></td><td></td><td>22.63</td><td>0.075433333</td><td>0.02242</td><td></td><td>253</td><td>380786.4123</td><td>55</td></td<>		20		15.3150	0.0511			15.315	0.05105			22.63	0.075433333	0.02242		253	380786.4123	55
153150 0.0531 14 15315 0.05105 31 2.56 0.075433333 0.03563		18		15.3150	0.0511			15.315	0.05105			22.63	0.075433333	0.02522		254	364785.5875	49
153150		14		15.3150	0.0511			15.315	0.05105	31		22.63	0.075433333	0.03363		226	322836.8382	37
15.3150		18		15.3150	0.0511			15.315	0.05105	32		22.63	0.075433333	0.02522		277	374732.4357	50
153150		4 5		15.3150	0.0511			15.315	0.05105	33		22.63	0.075433333	0.03363		245	332/83.6864	300
153150 0.0511 8 15.315 0.0510 36 2.68 0.07543333 0.04000 0.K 1.12 15.3150 0.0511 1 15.315 0.05105 37 2.68 0.07543333 0.04000 0.K 1.02 15.3150 0.0511 2 15.315 0.05105 39 2.68 0.07543333 0.04000 0.K 2.07 15.3150 0.0511 2 15.315 0.05105 39 2.68 0.07543333 0.04000 0.K 2.07 15.3150 0.0511 2 15.315 0.05105 41 2.68 0.07543333 0.02242 0.K 2.07 15.3150 0.0511 2 15.315 0.05105 42 2.68 0.07543333 0.02020 0.K 2.07 15.3150 0.0511 2 15.315 0.05105 42 2.68 0.07543333 0.02020 0.K 2.07 15.3150 0.0511 2 15.315 0.05105		2 2		15.3150	0.0511			15.315	0.05105	4. c.		22.63	0.07543333	0.04000		301	384679 2838	0 0 0 0
153150 10511 12 15315 105105 33 2.263 107543333 104000 10K 20F		00		15.3150	0.0511			15.315	0.05105	36		22.63	0.075433333	0.04000		112	205204.1627	97
153150 0.0511 8 15.315 0.05105 38 2.66 0.0743333 0.004000 0K 147 147 143 153150 0.0511 2.0 15.315 0.05105 3.9 2.66 0.07543333 0.02242 0K 2.06 2.06 15.315 0.05105 3.0 2.68 0.07543333 0.02242 0K 2.06 2.		12		15.3150	0.0511			15.315	0.05105	37		22.63	0.075433333	0.04000		207	296888.8131	39
153150 0.0511 2.0 15.315 0.05105 3.9 2.565 0.074543333 0.02242 0.0K 2.565 0.074543333 0.02021 0.0K 2.565 0.074543333 0.0Z522 0.0K 2.565		00		15.3150	0.0511			15.315	0.05105	38		22.63	0.075433333	0.04000		147	235044.7072	87
153150 0.0511 2.0 15315 0.05103 4.0 2.68 0.07543333 0.02224 0.04		8 8		15.3150	0.0511			15.315	0.05105	39		22.63	0.075433333	0.02242		286	390/33.2605	44.
153150 15317 15315 153		78		15.3150	0.0511			15.315	0.05105	4 4		22.63	0.075/13333	0.02242		310 244	354838 7393	92
153159 0.0511 1.0 1.5.315 0.05105 4.4 2.66 0.07543333 0.04000 0.K 1.75 153150 0.0511 2.2 15.315 0.05105 4.4 2.263 0.07543333 0.02018 0.K 2.64 153150 0.0511 2.2 15.315 0.05105 4.4 2.263 0.07543333 0.02018 0.K 2.64 153150 0.0511 2.2 15.315 0.05105 4.5 2.263 0.07543333 0.02018 0.K 2.24 153150 0.0511 2.2 15.315 0.05105 4.4 2.263 0.07543333 0.02018 0.K 2.24 153150 0.0511 2.2 15.315 0.05105 4.9 2.263 0.07543333 0.02018 0.K 2.24 153150 0.0511 2.2 15.315 0.05105 4.9 2.263 0.07543333 0.02018 0.K 2.24 153150 0.0511 2.2 15.315 0.05105 5.2 2.263 0.07543333 0.02018 0.K 2.24 153150 0.0511 2.2 15.315 0.05105 5.2 2.263 0.07543333 0.02018 0.K 2.24 153150 0.0511 1.4 15.315 0.05105 5.2 2.263 0.07543333 0.02018 0.K 2.24 153150 0.0511 1.4 15.315 0.05105 5.2 2.263 0.07543333 0.02018 0.K 2.24 153150 0.0511 1.4 15.315 0.05105 5.2 2.263 0.07543333 0.02018 0.K 2.24 153150 0.0511 1.4 15.315 0.05105 5.2 2.263 0.07543333 0.02018 0.K 2.24 153150 0.0511 1.4 15.315 0.05105 5.2 2.263 0.07543333 0.02028 0.K 2.24 153150 0.0511 1.4 15.315 0.05105 5.2 2.263 0.07543333 0.02629 0.K 2.25 153150 0.0511 1.4 1.5 1.5 0.05105 5.2 2.263 0.07543333 0.05629 0.K 2.25 153150 0.0511 1.5 0.05105 5.2 2.263 0.07543333 0.05629 0.K 2.25 153150 0.0511 1.5 0.05105 5.5 2.263 0.07543333 0.05629 0.K 2.25 153150 0.0511 1.5 0.05105 5.5 2.263 0.07543333 0.05629 0.K 2.25 153150 0.0511 1.5 0.05105 5.5 0.07543333 0.05629 0.K 2.25 153150 0.0511 1.5 0.05105 5.5 0.07543333 0.05629 0.K 2.25 153150 0.0511 1.5 0.05105 0.05105 0.05050 0.05050 0.05050 0.05050 0.05050 0.05050 0.0		22		15.3150	0.0511			15.315	0.05105	42		22.63	0.07543333	0.02018		337	426627.7313	65
153150 0.0511 22 15.315 0.05105 44 2.63 0.07943333 0.02018 0.04 0.05 15.3150 0.0511 18 15.315 0.05105 45 2.68 0.07943333 0.02522 0.05 0.05 15.3150 0.0511 18 15.315 0.05105 48 2.68 0.07943333 0.02522 0.05 0.05 15.3150 0.0511 12 15.315 0.05105 48 2.263 0.07943333 0.02018 0.07 0.29 15.3150 0.0511 22 15.315 0.05105 49 2.263 0.07943333 0.02018 0.07 0.29 15.3150 0.0511 16 15.315 0.05105 51 2.263 0.07943333 0.02018 0.07 0.29 15.3150 0.0511 14 15.315 0.05105 52 0.07543333 0.02018 0.07 0.29 15.3150 0.0511 14 15.315 0.05105 52 0.07543333 0.02018 0.07 0.29 15.3150 0.0511 14 15.315 0.05105 52 0.07543333 0.02018 0.07 0.29 15.3150 0.0511 14 15.315 0.05105 52 0.07543333 0.02018 0.07 0.29 15.3150 0.0511 14 15.315 0.05105 52 0.07543333 0.02028 0.07 0.29 15.3150 0.0511 18 15.315 0.05105 52 0.07543333 0.05629 0.07 0.25 15.3150 0.0511 18 15.315 0.05105 52 0.07543333 0.05629 0.07 0.25 15.3150 0.0511 18 15.315 0.05105 0.07543333 0.05629 0.07 0.25 15.3150 0.0511 18 0.05105 0.07543333 0.05629 0.07 0.25 15.3150 0.0511 18 0.05105 0.07543333 0.05629 0.07 0.25 15.3150 0.0511 18 0.05105 0.07543333 0.05629 0.07 0.25 15.3150 0.0511 18 0.05105 0.07543333 0.05629 0.07 0.25 15.3150 0.0511 18 0.05105 0.05105 0.07543333 0.05629 0.07 0.25 15.3150 0.0511 0.0511 0.05105 0.05005 0.05		101		15.3150	0.0511			15.315	0.05105	43		22.63	0.075433333	0.04000		175	260993.6379	48
153150 0.0511 18 15.315 0.05105 45 2.056 0.07543333 0.00552 0.0K 2.06 15.315 0.0K 2.0K 2.0K 0.07543333 0.0K 2.0K		22		15.3150	0.0511			15.315	0.05105	44		22.63	0.075433333	0.02018		264	396787.1868	64
15.3150 0.0511 22 15.315 0.05105 46 2.263 0.07543333 0.02012 FALSO 241 15.315 15.315 0.05101 22 15.315 0.05105 48 2.263 0.07543333 0.02012 FALSO 241 15.315 15.315 0.05105 48 2.263 0.07543333 0.02013 0.02013 0.0511 15.315 0.05105 59 2.663 0.07543333 0.02013 0.02013 0.0511 15.315 0.05105 51 2.263 0.07543333 0.02024 0.07		18		15.3150	0.0511			15.315	0.05105	45		22.63	0.075433333	0.02522		206	334945.043	93
15.3150 0.0511 18 15.315 0.05105 47 2.66 0.07943333 0.02222 0.04 3.34 15.3150 0.0511 22 15.315 0.05105 48 2.66 0.07943333 0.02222 0.04 3.34 15.3150 0.0511 22 15.315 0.05105 59 2.68 0.07943333 0.02018 0.04 2.94 15.3150 0.0511 20 15.315 0.05105 51 2.68 0.07943333 0.02018 0.04 3.05 15.3150 0.0511 16 15.315 0.05105 52 2.68 0.07943333 0.02022 0.04 3.05 15.3150 0.0511 14 15.315 0.05105 53 2.68 0.07943333 0.02028 0.04 3.05 15.3150 0.0511 14 15.315 0.05105 55 2.68 0.07943333 0.02036 0.04 2.58 15.3150 0.0511 10 15.315 0.05105 57 2.263 0.07943333 0.08529 0.04 2.25 15.3150 0.0511 10 15.315 0.05105 57 2.263 0.07943333 0.08529 0.04 2.25 15.3150 0.0511 10 15.315 0.05105 57 2.263 0.07943333 0.08529 0.04 2.25 15.3150 0.0511 10 15.315 0.05105 57 2.263 0.07943333 0.08529 0.04 2.25 15.3150 0.0511 10 15.315 0.05105 57 2.263 0.07943333 0.08529 0.04 2.25 15.3150 0.0511 10 10.315 0.05105 57 2.263 0.07943333 0.08529 0.04 2.25 15.3150 0.0511 10 10.315 0.05105 57 2.263 0.07943333 0.08529 0.04 2.25 15.3150 0.0511 10 10.315 0.05105 0.07943333 0.08529 0.04 2.25 15.3150 0.0511 10 0.05105 0.07943333 0.08529 0.04 2.25 15.3150 0.0511 10 0.05105 0.07943333 0.08529 0.04 2.25 15.3150 0.0511 10 0.05105 0.05105 0.07943333 0.08529 0.04 0.05105		22		15.3150	0.0511			15.315	0.05105	46		22.63	0.075433333	0.02018		241	386840.3386	116
15.3150 0.0511 2.2 15.315 0.05103 4.9 2.268 0.07543333 0.02008 0.04 0.04 0.		8 2		15.3150	0.0511			15.315	0.05105	47		22.63	0.075433333	0.02522		334	394626.132	50
15.3150 0.0511 2.2 15.315 0.05105 5.0 2.68 0.07543333 0.02082 0.04 3.67 15.315 0.05105 5.1 2.68 0.07543333 0.0282 0.04 3.67 15.315 0.0511 14 15.315 0.05105 5.1 2.68 0.07543333 0.0282 0.04 3.67 15.315 0.05105 5.1 2.68 0.07543333 0.0282 0.04 3.67 15.315 0.05105 5.1 2.68 0.07543333 0.0282 0.04 3.67 15.315 0.05105 5.1 2.68 0.07543333 0.0829 0.04 2.28 15.315 0.05105 5.1 2.263 0.07543333 0.0829 0.04 2.25 15.315 0.05105 5.1 2.263 0.07543333 0.0829 0.04 2.25 15.315 0.05105 5.1 2.263 0.07543333 0.0829 0.04 2.25 15.315 0.05105 5.1 2.263 0.07543333 0.0829 0.04 2.25 15.315 0.05105 5.1 2.263 0.07543333 0.0829 0.04 2.25 15.315 0.05105 5.1 2.263 0.07543333 0.0829 0.04 2.25 15.315 0.05105 0.07543333 0.0829 0.04 2.25 15.315 0.05105 0.07543333 0.0829 0.04		77 62		15.3150	0.0511			15.315	0.05105	8 4 6		22.63	0.075433333	0.02018		368	4365/4.5/94	58
15.3150 0.0511 20 15.315 0.05105 51 2.68 0.075433333 0.02242 0.07 367 367 15.3150 0.0511 16 15.315 0.05105 53 2.263 0.07543333 0.02882 0.07 3.0 15.3150 0.0511 14 15.315 0.05105 54 2.263 0.075433333 0.03863 0K 3.0 15.3150 0.0511 14 15.315 0.05105 54 2.263 0.075433333 0.03863 0K 2.283 15.3150 0.0511 12 15.315 0.05105 55 2.263 0.075433333 0.03863 0K 2.283 15.3150 0.0511 12 15.315 0.05105 55 2.263 0.075433333 0.03863 0K 2.25 15.3150 0.0511 18 15.315 0.05105 57 2.263 0.075433333 0.03863 0K 2.25 15.3150 0.0511 8 1		77		15.3150	0.0511			15.315	0.05105	50		22.63	0.07543333	0.02018	šš	320	416680.8831	09
15.3150 0.0511 16 15.315 0.05105 5.2 2.63 0.07943333 0.0282 0.0K 310 1.5		20		15.3150	0.0511			15.315	0.05105	51		22.63	0.075433333	0.02242		367	420573.8049	52
15.3150 0.0511 22 15.315 0.05105 53 22.63 0.07543333 0.02018 0K 258 15.315 0.02018 15.315 0.0511 12 15.315 0.05108 55 22.63 0.07543333 0.0353 0K 228 15.315 0.0511 10 15.315 0.05108 55 22.63 0.07543333 0.0352 0K 228 15.315 0.0511 10 15.315 0.05108 57 22.63 0.07543333 0.0352 0K 228 15.315 0.0511 10 15.315 0.05108 57 22.63 0.07543333 0.0352 0K 227 15.315 0.0511 18 15.315 0.05108 57 22.63 0.07543333 0.0352 0K 129		16		15.3150	0.0511			15.315	0.05105	52		22.63	0.075433333	0.02882		310	368678.3871	40
15.3150 0.0511 14 15.315 0.05105 54 22.65 0.07543333 0.03659 0K 283 15.315 0.0511 10 15.315 0.05105 57 22.63 0.07543333 0.03629 0K 255 15.315 0.0511 10 15.315 0.05105 57 22.63 0.07543333 0.03629 0K 255 15.315 0.0511 8 15.315 0.05105 57 22.63 0.07543333 0.03629 0K 199		22		15.3150	0.0511			15.315	0.05105	23		22.63	0.075433333	0.02018		398	446521.4276	28
15.3550 0.0511 10 15.315 0.05105 57 22.63 0.075433333 0.05629 0K 22.53 15.3550 0.0511 18 15.315 0.05105 57 22.63 0.075433333 0.05629 0K 22.50 17.54733333 0.05629 0K 22.50 17.5473333 0.05629 0K 22.50 17.547333 0.05629 0K 22.50 17.5473333 0.05629 0K 22.50 17.5473333 0.05629 0K 22.50 17.5473333 0.05629 0K 22.50 17.547333 0.05629 0K 22.50 17.5473333 0.05629 0K 22.50 17.547333 0.05629 0K 22.50 17.54733 0.05629 0K 22.50 0		4 5		15.3150	0.0511			15.315	0.05105	54		22.63	0.075433333	0.03363		283	342730.5345	38
15.3350 0.051 8 15.3315 0.05156 57 2.268 0.0743333 0.0829 0.K		1 5		15.3150	0.0511			15.315	0.05105	0 0 0 1		22.63	0.075433333	0.03629		727	290834 1824	64 67
		00		15.3150	0.0511			15.315	0.05105	57		22.63	0.075433333	0.03629		199	264885.2516	96

MODEL B - 1 Optimization process (Results 56 variables)

Parameters considered Structural mars a reel 8.274
Cladding mark
Load case 4.8.5

	B - 1 : Ste	ep 9 - After cr	: Step 9 - After cross section opt.	ld. Family				ld. Family				ld. Family						
Nº it	Steel class	Max. displacemen X t (m) F (Structure)	X - Nº of Y - Nº of Frames Frames	Beams Top - X direction	Tipe of profile	Length All element (m) Dis be	Allowed max. Displacement beams (m)	Beams Top - Y Direction T Extension	Tipe of profile el	Length Allowed max. element (m) Displacement beams (m)		Weams Top - Y Direction Tipe of central part profile top	Length Allowed max. element (m) Displacement beams (m)		Beams Top - Tip Y Direction pro	Tipe of Length profile element (Length Allowed max. element (m) Displacement beams (m)	max. nent n)
1	s275	0.035095	3	24	HEB 400	13.450	0.06725	4 4	HEB 600	8.000	0.0400	24 HEB		090:0	12 HEB	12		0.060475
2		0.026401	4			10.088	0.05044	5		8.000	0.0400	25		0.073	10	14	14.514	0.07257
m s		0.021933	9 1	42		6.725	0.03363	7 8		8.000	0.0400	35	14.514 0	0.073	9 9	14.514		0.07257
4 ru		0.021001	, Ln			8.070	0.04035	0 4		8.000	0.0400	54		0.040	9 12	8.06333333		0.040316667
י ע		0.022155				5.764	0.02882			8.000	0.0400	40		0.073	101	14		0.07257
7		0.019352	7			5.764	0.02882	0 80		8.000	0.0400	32		0.091	. so	18.		0.0907125
∞		0.031528	4	40		10.088	0.05044			8.000	0.0400	40		0.045	16	0.0	0	04535625
ο ;	OI.	0.026487	n 1	40		8.070	0.04035	9 0		8.000	0.0400	36		0.060	12	12		0.060475
3 5		0.022442	` *	30 30		5.764	0.02882	× 4		8.000	0.0400	3.2	9.071	0.052	14	10.36/1429	5	0.051835714
12		0.016785	. 00	22		5.044	0.02522	6		8.000	0.0400	45		0.073	10	14		0.07257
13		0.028869	2	7 45		8.070	0.04035	9		8.000	0.0400	42	10.367 0	0.052	14	10.3671429	_	0.051835714
14		0.025142	4	32		10.088	0.05044	ı, ı		8.000	0.0400	30		0.060	12	12		0.060475
15		0.02224	9 1	3 48		6.725	0.03363	7 8		8.000	0.0400	42	12.095 0	0.060	12	172	12.095	0.060475
1 1		0.028193	. 4	4 24		10.088	0.05044	0 10		8.000	0.0400	50 2		0.091	g ∞	18.		0.0907125
18		0.028869				8.070	0.04035	9		8.000	0.0400	24		0.091	000	18.		0.0907125
19		0.035159				13.450	0.06725			8.000	0.0400	36		0.040	18	8.06333333		0.040316667
50		0.014595				4.483	0.02242	П		8.000	0.0400	20		0.073	10	14		0.07257
7 7		0.027163				8.070	0.04035	9 8		8.000	0.0400	30	14.514 0	0.073	9 5	14	14.514	0.07257
73 23		0.01509		066		5.764 4,483	0.02242	10		8.000	0.0400	8 06		0.040	18	8.06333333		0.040316667
54		0.016884				4.483	0.02242	10		8.000	0.0400	40		0.091	00	18.		0.0907125
25		0.023045		36		6.725	0.03363	7		8.000	0.0400	28		0.091	∞	18.		0.0907125
56		0.021741				6.725	0.03363			8.000	0.0400	49		0.052	14	10.3671429	_	0.051835714
72 82		0.035095	m o	5 21		13.450	0.06725	4 5		8.000	0.0400	70	14.514 0	0.073	10	12	14.514	0.07257
2 2		0.016785		72		5.044	0.02522			8.000	0.0400	63		0.052	14	10.3671429		0.051835714
30		0.023458				6.725	0.03363	7		8.000	0.0400	26		0.045	16	9.07125		0.04535625
31		0.016785		80		5.044	0.02522	6		8.000	0.0400	72		0.045	16	0.6		0.04535625
32		0.022087				6.725	0.03363	7		8.000	0.0400	63	8.063 0	0.040	18	8.06333333		0.040316667
8 %		0.026193				5.044	0.05044			8.000	0.0400	5 - 2		0.040	9 2	8.06333333		0.040316667
32		0.035099				13.450	0.06725			8.000	0.0400	16		0.091	∞	18.		0.0907125
36		0.025575				8.070	0.04035	9		8.000	0.0400	48		0.045	16	9.07125		0.04535625
37		0.035154				13.450	0.06725	4 6		8.000	0.0400	28	10.367 0	0.052	14	10.3671429		0.051835714
9 8		0.016884				4.483	0.02242	9 9		8.000	0.0400	08		0.045	16	10.36		0.04535625
40		0.016785	8			5.044	0.02522			8.000	0.0400	54		0900	12	12		0.060475
41		0.013133		100		4.035	0.02018	11		8.000	0.0400	88		0.045	16	9.07125		0.04535625
45		0.024819				10.088	0.05044	2 ;		8.000	0.0400	35		0.052	14	10.3671429	_	0.051835714
8 4		0.018302		48		5.044	0.02018	11		8,000	0.0400	3,6	18 143 D	0.073	9 «	78.	18.145 0	0.0071257
45		0.0127				4.035	0.02018	T		8.000	0.0400	44		0.091	00	18.		0.0907125
46		0.016785				5.044	0.02522	6		8.000	0.0400	06		0.036	20	7		0.036285
47		0.014047	0 6	9 110		4.035	0.02018	11		8.000	0.0400	66	8.063 0	0.040	18	8.06333333	0.0	0.040316667
9 4		0.014399				4.035	0.02010	1 1		8,000	0.0400	99		0.052	14	10.3671429		0.051835714
2 05		0.014004				4.483	0.02242			8.000	0.0400	100		0.036	50	7		0.036285
51		0.018212				5.764	0.02882	00		8.000	0.0400	80		0.036	20	7		0.036285
52		0.014593	10	120		4.035	0.02018	11		8.000	0.0400	110	7.257 0	0.036	20	7		0.036285
8 24		0.024263				8.070	0.04035	, 9		8.000	0.0400	? 09		0.036	70		7.257	0.036285
. 55		0.024855	4 11			10.088	0.05044	. S		8.000	0.0400	20		0.036	20	7		0.036285
26		0.034932	3 1.			13.450	0.06725	4		8.000	0.0400	40		0.036	20	7		0.036285

APPENDIX D - 2: MODEL B - 1

MODEL B - 1 Optimization process (Results 56 variables)

Parameters considered: Structural Steel 5278

Clading T Glass

Load case 48 5

B - I : Step	p 9 - Arter cr	9 - After cross section opt.										ŀ								
				Allowed max.	Beams	i	1		Columns	i	1		Columns	i	Length	Allowed max.				Allowed max.
Nº it	facade pr	i ipe of Lengtn profile (r	Lengtn element D (m) nt (n	Displaceme nt beams (m)	front	npe or profile	Length element (m)	Allowed max. Displacement beams (m)	Laterals left short	npe or profile	Lengtn , element (m) (Allowed max. Displacement beams (m)	Laterals rigth short	ripe or profile	element (m)	Displacement columns (m)	Columns front	ripe of profile	Lengtn element Dis (m) co	Displacement columns (m)
1 2	3 HEB	88	13.450	0.06725	8 4	HEB	13.45	0	9 2	HEB	80 80	0.0267		HEB 450	3.000	0.0100 0.0100		HEB	15.3150 15.3150	0.0511
e 4	9		6.725	0.033625	9 2		5.764285714	0.033625	10 6		00 00	0.0267	in o		3.000	0.0100	0 14		15.3150	0.0511
5	LO I		8.070	0.04035	LO I		8.07	0.04035			00	0.0267	6		3.000	0.0100			15.3150	0.0511
7	, ,		5.764 0	0.02882143 0.02882143	` '		5.764285714	0.028821429 0.028821429	v 4		∞ ∞	0.0267	v 4		3.000	0.0100			15.3150 15.3150	0.0511
80 0	4 n		10.088	0.0504375	4 n		10.0875	0.0504375	80 4		80 0	0.0267	80		3.000	0.0100	0 10		15,3150	0.0511
9 01	۷ ۲			0.02882143	0 7		8.U/ 5.764285714	0.02			× ×	0.0267	7		3.000	0.0100			15.3150	0.0511
11 5	e		13.450	0.06725	m		13.45		00 L		οο c	0.0267	00 L		3.000	0.0100			15.3150	0.0511
12	90 LN		5.044 C	0.02521875	∞ r∪		5.04375	0.02521875	0 10		∞ ∞	0.0267	2 7		3.000	0.0100	0 18		15.3150 15.3150	0.0511
14	4 4		10.088	0.0504375	4 4		10.0875	0.0504375	9 9		00 0	0.0267	9 4		3.000	0.0100			15.3150	0.0511
19	2 /			0.02882143	7		5.764285714	0.0	0 00		0 00	0.0267	· •		3.000	0.0100			15.3150	0.0511
17	4 .		10.088	0.0504375	4 1		10.0875	0	4 4		00 0	0.0267	4 -		3.000	0.0100	0 10		15.3150	0.0511
19	n m		13.450	0.06725	nm		13.45	0.04035	4 0		0 00	0.0267	4 0		3.000	0.0100			15.3150	0.0511
70	6			0.02241667	6		4.483333333	0.02	15		00	0.0267	10		3.000	0.0100	2		15.3150	0.0511
21	10.1		8.070	0.04035	1 12		8.07	0.04035	10.0		00 0	0.0267	r.		3.000	0.0100			15.3150	0.0511
73	6		5.764 U	0.02241667	, 6		5.764285714 4.483333333	0.022416667	0 00		0 00	0.0267	0 6		3.000	0.0100	20		15.3150	0.0511
24	6			0.02241667	6		4.483333333	0.0	4		00	0.0267	4		3.000	0.0100			15.3150	0.0511
25	9 4		6.725	0.033625	9 4		6.725	0.033625	4 1		00 O	0.0267	4 1		3.000	0.0100			15.3150	0.0511
27	o m		13.450	0.06725	m		13.45	0.06725	0		0 00	0.0267	· ru		3.000	0.0100			15.3150	0.0511
28	6			0.02241667	6		4.483333333	0.022416667	9		00	0.0267	9		3.000	0.0100	0 20		15.3150	0.0511
82 82	00 U			0.02521875	00 U		5.04375	ō			00 (0.0267	7		3.000	0.0100			15.3150	0.0511
31	۰ ۵		5.044 0	0.033625	ω ω		6.725 5.04375	0.033625	0 00		× ×	0.0267	× ×		3.000	0.0100			15.3150	0.0511
32	9			0.033625	9		6.725				80	0.0267	6		3.000	0.0100			15.3150	0.0511
33	4 (0.0504375	4 (10.0875				00 (0.0267	6		3.000	0.0100	0 10		15.3150	0.0511
35 25	00 m		13.450	0.02521875	xo m		5.04375	0.02521875	D 4		∞ ∞	0.0267	p 4		3.000	0.0100			15.3150	0.0511
36	2		8.070	0.04035	2		8.07		80		∞	0.0267	80		3.000	0.0100	-		15.3150	0.0511
37	m d		13.450	0.06725	e o		13.45	0.06725	7 7		оо о	0.0267	7		3.000	0.0100	8 6		15.3150	0.0511
33 8	6			0.02241667	0		4.483333333		. 8		0 00	0.0267	~ 60		3.000	0.0100			15.3150	0.0511
9 ;	ω (0.02521875	∞ (5.04375	ō	9 0		∞ 0	0.0267	9 0		3.000	0.0100			15.3150	0.0511
4 4	O1 4		10.088	0.0504375	DI 4		10.0875	0.0504375			0 00	0.0267	۸ ٥		3,000	0.0100	10		15.3150	0.0511
43	10		4.035	0.020175	10		4.035		2		∞	0.0267	2		3.000	0.0100			15.3150	0.0511
4 4	80 Ç		5.044 0	0.02521875	∞ 5		5.04375	0	4 4		00 C	0.0267	4 -		3.000	0.0100	18		15.3150	0.0511
5 4	P 8			0.02521875	OT 88		5.04375	0.02521875	-		× ×	0.0267	4 01		3.000	0.0100			15.3150	0.0511
47	10			0.020175	10		4.035				80	0.0267	6		3.000	0.0100			15.3150	0.0511
48	10		4.035	0.020175	10		4.035		9		80	0.0267	9		3.000	0.0100			15.3150	0.0511
6 S	10		4.035	0.020175	10		4.035	0.020175	7 01		00 00	0.0267	, ot		3.000	0.0100	22		15.3150	0.0511
51.5				0.02882143	7		5.764285714	0.028821429			0 00	0.0267	10		3.000	0.0100			15.3150	0.0511
25	10		4.035	0.020175	10		4.035				00	0.0267	10		3.000	0.0100			15.3150	0.0511
23.23	9 11		6.725	0.033625	9 1		6.725	0.033625	10		00 0	0.0267	01 5		3.000	0.0100			15.3150	0.0511
* 14	o 4		10.088	0.0504375	υ 4		10.0875	Ö			0 00	0.0267	10		3.000	0.0100			15.3150	0.0511
26	m		13.450	0.06725	m		13.45				00	0.0267	10		3.000	0.0100	8		15.3150	0.0511

APPENDIX D - 2 : MODEL B - 1

MODEL B - 1 Optimization process (Results 56 variables)

Parameters considered: Statutula in: Stat

	Utill ratio																																														
	Total mass (kg)		129959.8037					112724.6648				101909.9169					164398.5864						Н	195/53.238					96493.65643		•	165254.9456		Н				267333.6231						Η.	135554.434		
20410101010	Total number of elements	100	109	244	191	170	152	150	212	132	195	169	177	237	111	129	151	147	205	317	203	153	205	115	254	226	277	245	301	112	207	147	310	244	337	175	264	206	334	368	294	320	367	310	398	255	227
3	ر max < اسax = السام			ťξ	šš	ŏ	ŏ			ŏ	ŏ	šδ	šδ	ŏ	ŏ	š			šě	ŏ			žδ			ŏ	ŏ	šδ	žč	šš		šδ				š	š							šί	ž č	šš	ŏŏ
	max. Allow displacement	0.04000	0.04000	0.03363	0.04000	0.02882	0.02882	0.04000		0.04000	0.02522	0.04000	0.03363	0.02882	0.04000	0.04000	0.04000	0.02242	0.02882	0.02242	0.02242	0.03363	0.03363	0.04000	0.02522	0.03363			0.04000			0.04000						0.02522	0.02018	0.02018	0.02018	0.02018	0.02242	0.02882	0.02018	0.03629	
	Allowed max. Displacement columns (m)	0.07543	0.07543	0.07543	0.07543	0.07543	0.07543	0.07543	0.07543	0.07543	0.07543	0.07543	0.07543	0.07543	0.07543	0.07543	0.07543	0.07543	0.07543	0.07543	0.07543	0.07543	0.07543	0.07543	0.07543	0.07543	0.07543	0.07543	0.07543	0.07543	0.07543	0.07543	0.07543	0.07543	0.07543	0.07543	0.07543	0.07543	0.07543	0.07543	0.07543	0.07543	0.07543	0.07543	0.07543	0.07543	0.07543
	Length element (m)	22.63	22.63	22.03	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63
	Tipe of profile	2 HEB	m =			7	m		_	61					~	•		_ ^		_	10			~ ~		_	61			. 10	_	~ .		_	61	~	-	0.10			•	_		6 1 6	~ -		10.6
	Columns Front Ablock							o 01		12	13	41 1	1 %			S1 1				77	25	3 1	7 7						¥ %	3 %						43		45						25 1	7, 17	1 15	55
	Allowed max. Displacement columns (m)	0.05105	0.05105	0.05105	0.05105	0.05105	0.05105	0.05105	0.05105	0.05105	0.05105	0.05105	0.05105	0.05105	0.05105	0.05105	0.05105	0.05105	0.05105	0.05105	0.05105	0.05105	0.05105	0.05105	0.05105	0.05105	0.05105	0.05105	0.05105	0.05105	0.05105	0.05105	0.05105	0.05105	0.05105	0.05105	0.05105	0.05105	0.05105	0.05105	0.05105	0.05105	0.05105	0.05105	0.05105	0.05105	0.05105
ion opt.	Length element (m)	15.315	15.315	15.315	15.315	15.315	15.315	15.315	15.315	15.315	15.315	15.315	15.315	15.315	15.315	15.315	15.315	15.315	15.315	15.315	15.315	15.315	15.315	15.315	15,315	15.315	15.315	15.315	15.315	15.315	15.315	15.315	15.315	15.315	15.315	15.315	15.315	15.315	15.315	15.315	15.315	15.315	15.315	15.315	15.315	15.315	15.315
oss sect	Tipe of profile	HEB 500																																													
p 9 - After cross section opt.	Columns back	8	10	16	12	16	16	10	16	∞	18	12	14	16	10	12	× 6	12	16	20	20	14	14	8 00	18	14	18	14	10	φ ∞	12	∞ ç	20	18	22	10	22	18	18	22	22	22	20	16	22	12	10
B - 1 : Step	Nº it	1	7	ο 4	ı,	9	7	∞ σ	10	11	12	13	12	16	17	18	19	20	22	23	24	25	26	/7	29	30	31	32	33	35	36	37	39	40	41	42	43	44	45	47	48	49	20	51	52	24 3	55
		<u> </u>					•																																								

MODEL B - 1 Optimization process (Results 56 variables)

Parameters considered: Structural materi s275

Cladding materia 60ase tax 5

Load case tax 5

1 225097839 Custinator ratio Custinator rat				
225097.855 241099.416 29299.237 2829371.539 308835.6612 318941.4446 308937.252 27699.5167 33887.9427 244991.5553 34489.915.553 34489.915.553 34289.99 251016.70.82 251016.70.945 312889.99 251016.70.945 312889.99 251016.70.945 312889.99 251016.70.945 312889.99 251016.70.945 312889.99 251016.70.945 312889.99 251016.70.945 312889.99 251016.70.945 312889.99 32788.686 32888.833 3778.4457 38079.3866 38078.4457 38079.3866 38078.4457 38079.3866 38078.4457 38079.3866 38078.4457 38079.3866 38078.4457 38079.3866 38078.4457 38079.3866 38078.4457 38079.3866 38078.4457 38079.3866 38078.4457 38079.3866 38078.4457 38079.3868 38089.3871 38080.3386	ilization ratio Utilization ratio ss section (%) cross section (%)	Total mass (kg)	Utilization ratio cross section (%)	Utilization ratio cross section (%)
241099 9416 241099 9416 241099 9416 30683 56612 310893 7, 2882 310893 7, 2882 27094 14564 31883 74827 24489 14553 24489 14553 30294 1419 34784 6508 25104 7, 756 25104 7, 756 25104 7, 756 25104 7, 756 3115 30 994 25115 10 10 34784 6508 3115 30 994 25115 10 10 34784 6508 3115 30 994 25115 10 10 3478 5886 3686 353 3664 7, 2457 3208 33 7, 75 26093 7, 16 25008 7, 3457 3200 4, 557 3200 4, 557 320 534 320 534 32	87 87	175567.8109	88	38
382731.539 308835.6612 3108835.6612 3108835.6612 270840.4464 208997.2982 270840.4467 244991.553 344891.8812 289941.9649 251046.7897 307943.419 348784.82889 251014.204 254988.09945 4106.6.958 307889.99 215151.0108 307889.99 215151.0108 30888.3342 31472.4457 32888.33342 334945.604 34888.73342 3488.73342 3488.73342 3488.73342 3488.73342 3488.73342 3488.73342 3486.64 3486.64 3486.6132 3486.64 3486.6132 3486.64 3486.6132 3486.6132 3486.6133 3486.6133 3486.6133 3486.6133 3486.6132 3486.6132 3486.6132 3486.6132 3486.6132 3486.6132 3486.6132 3486.6132 3486.6132 3486.6132 3486.734.579 34773.0134 3486.734.132 34723.594 34723.514.77		129959.8037	99	99 ;
3080.54.1233 3080.55.61.233 3080.97.282 27.093.71167 3189.71.827 3189.71.827 32.809.92.71.82 32.809.92.71.82 32.809.94.56 32.10.10.10.08 32.11.83.09.94 25.10.67.08.26.66 32.809.94.56 31.288.99 25.10.67.08.26.66 32.809.94.56 31.288.99 25.10.67.08.26.66 36.709.71.82 36.609.71.82 36.609.71.82 36.609.71.82 36.609.71.82 36.609.72 37.72.44.57 37.72.44.57 37.72.44.57 37.72.44.57 37.72.44.57 39.77.34.86.8 34.86.7.73.13 26.009.36.73 36.60.73.36.99 36.67.73.38.09 36.67.73.38.09 36.67.73.38.09 36.67.73.38.09 36.67.73.38.09 36.67.73.38.09 36.67.73.38.09 36.67.73.38.09 36.73.45.59.44 40.67.74.59.83 31.72.52.50.44 26.00.83.11.22 31.72.22.50.63 36.72.23.23 36.72.33.23 36.73.33.2		152998.3162	65	6
3,00832,0012,3884,1464,38844,1464,38884,1464,38884,1464,38884,1464,38884,1464,38886,999,155,587,3115,309,487,4106,7897,3886,999,497,4106,2890,945,4106,290,945,4106,290,290,290,290,290,290,290,290,290,290	000	103647.6514	7.	ν. Γ
208997,2882 270900 4661 270900 4662 270901,5553 244891,5553 244891,5553 244891,5553 244891,5649 251046,7897 251047		160654.6896	77	
77080048667 270804048667 270804115553 24499115553 24499115553 24499115553 24499115553 24499115553 24499115553 249913013934 251006 2310130934 251006 2310130934 251006 2310130934 2510101008 2151		260093.0246	67	19
276995,1167 288378,427 24991,5553 244891,15812 28994,19649 251046,7897 302943,1419 34278,6508 231115,3034 25701,4204 257083,5641 267083,5641 267083,5641 267083,5641 267083,5641 267083,5641 267083,5641 267083,5641 267083,5641 267083,5641 267083,5641 267083,5641 267083,5641 267083,5641 267083,5641 267083,5641 267084,456 312786,4123 34772,4357 34772,4357 36773,3664 36773,3674 36773,3674 36773,3674 3676,77313 36784,773 36787,3868 39787,1878,1878 31787,2879	48 48	89979.64269	79	7
338837,8427 24991,5553 344891,5912 286941,5649 251046,7897 302943,1419 346784,6908 231133,0934 251704,4204 257701,4204 257701,4204 25701,4204 25702,436 310082,716 280082,716 280082,716 280082,716 280082,716 280082,716 280082,716 280082,716 38073,2457 32772,4457 32773,2457 3273,2457 3273,2457 3273,2457 3273,2457 3273,2457 3273,2457 3273,2504 3273,2504 3273,2504 3273,2504 3273,2504 3273,2504 3273,2504 3273,2504 3273,2504 3273,2504 3273,2504 3273,2504 3273,2504 3273,2504 3273,2504 3273,2504 3273,2504 3273,2504 3273,2504	38	112724.6648	99	99
24499.1.5553 34499.1.8912 286941.9649 25.0647.897 30.943.4149 30.943.41419 30.943.4		120965.2629	78	24
24489112 286941.9649 25106.7897 30293.3419 30293.3419 30293.3419 30293.3419 30288.033 30889.956 312889.99 215151.01008 32776.64123 36778.5875 32786.888.8131 2002.04.7072 39773.2437 39773.3664 2006.01086 39867.347 39773.366 39787.1868		164572.2303	88	88
25,004,7897 25,004,7897 25,7004,7897 25,7004,7898 22,7004,426 32,8880,994 41065,6,956 32,880,994 41065,956 38,709,944 38,709,944 38,709 215,151,010 28,709 215,151,010 28,709 215,151,010 28,709 215,151,010 28,709 215,151,010 28,709 215,151,010 21,709 21,7		197916.1096	67	4 6
3029431419 340784608 241734094 2573101,4204 2573101,4204 2573101,4204 2573101,4204 2573101,4204 2573101,4204 2573101,4204 2573101,4204 2573101,4204 2573101,4204 2573101,4204 2573101,4204 257313 25741,427 25688 8131 25504 1657 25504	38 38	112292.4592	7.7	7 9
348794.6508 231133.0934 25700.4704 254938.4635 370839.5641 257700.4704 254938.5685 370839.5641 28607.945 312889.99 21151.0108 387786.4123 36778.5875 312889.99 21151.0108 387786.4123 38773.346.4123 38773.346.4123 3773.346.4123 3773.346.4123 3773.346.4123 3773.346.4123 3773.346.4123 3773.366.77313 260933.657 396773.868 334945.043 38786.77313 426577.313 42657.31477 426577.313 42657.31477 426577.313 42657.31477 42657.31477 42657.31477 42657.31477 42657.31477 42657.31477 42657.31477 42657.31477 42657.31477 42657.31477 42657.31477 42657.31477 42657.31477 42657.31477 42657.31477 42657.3147		127268.9785	99	99
2311334,2034,2511334,2034,2511334,2034,2511334,2034,2511334,20394,2710,40089,564,1289,99,2189,218		115794.8587	62	29
2-5-701-4204 2-5-701-4204 2-5-701-4204 2-5-701-62-59-68 3-7083-9-56-41 3-7083-9-9-45-6 3-7083-9-9-45-6 3-70-70-9-45-6 3-70-70-9-45-6 3-70-70-9-6-41-6-7 3-70-70-8-6-41-6-7 3-70-70-8-6-41-6-7 3-70-70-8-6-7 3-70-70-8-6-7 3-70-70-8-8-8-8-8-8-8-8-8-8-8-8-8-8-8-8-8		159475.3364	70	2 1
2.6708.9.5641.3 2.6708.9.5641.3 2.6708.2.716 2.8209.0.945 4.10.0.6.6.9568 3.8208.9.2.716 2.820.9.4.5.95 2.151.11.0.108 3.80786.4.12.3 3.847.9.2.835 3.847.9.2.838 3.847.9.2.838 3.847.9.2.838 3.847.9.2.838 3.847.9.2.838 3.847.9.2.838 3.847.9.2.838 3.847.9.3.838 3.847.9.3.838 3.848.9.	75 75	176763.3887	72	7 00
267048, 2686 32880, 9445 4106.66, 9548 4106.66, 9548 4106.66, 9548 31089, 97, 16, 18, 18, 18, 18, 18, 18, 18, 18, 18, 18		226203.7475	92	8 6
378890,9945 41066,6968 38099,4456 31289,99 215151,0108 38078,4423 36778,5875 37772,4457 3278,8664 3278,8664 3278,824,4767 29688131 25604,7072 296783,866 340634,57 39073,2605 400630,1086 3406734,1086 3406736,1086		132814.7193	89	8 89
41062.6.9568 360892.7.16 283049.4456 31288.99 215151.0108 38778.6.4123 32788.6.8823 32773.4357 32773.8664 280887.3342 286644.7072 286688.8131 286644.7072 286688.8131 286644.7072 286733.2605 390733.2605 390733.2605 390733.2605 34265.7.7313 28689.73811 28689.8387 48657.7313 28689.8387 48657.7313 28689.8387 48657.7313 38689.8387 48657.7313 28689.8387 48657.7313 28689.8387 48657.7312 28678.8371 28678.8371 28678.8371 28678.8371 28678.8371 28678.8371 28678.8371 28678.8371 28678.8371 286878.8371 286878.8371 286878.8371 286878.8371 286878.8371 286878.8371 286878.8371 286878.8371 286878.8371 286878.8371 286878.8371 286878.8371 286878.8371 286878.8371 286878.8371 286878.8371 286878.8371 2868878.8371		142114.8203	62	29
360892.716 28009.4456 213189.99 21510.08 36.745.2875 36.745.2875 37.742.4357 37.742.4357 380.745.24.4357 296888.8131 2950.41677 296888.8131 2950.41677 296888.8131 2950.41677 390.733.866 3946.873.886 3946.873.886 3946.873.886 3946.873.886 3946.873.886 3946.873.886 3946.873.886 3946.873.886 396.873.871 446.873.886 396.873.871 446.873.871 36.873.871 26.883.871 26.883.871 26.883.871 26.883.871 26.883.871 26.883.871 26.883.871 26.883.871 26.883.872 26.883.871 26.883.871 26.883.871 26.883.871 26.883.871 26.883.871 26.883.871 26.883.871 26.883.871 26.883.871 26.883.871 26.883.871 26.883.871 26.883.871 26.883.871 26.883.871 26.883.871 26.883.871 26.883.871	S	133022.1894	29	(6
	105 0	300785.837	75	22 33
		113238.8942	69	6 6
		195753.238	88	8 88
		179296.3725	62	.9
	49 49	138107.2742	67	9 7
	ù :	126359 7154	79	< 16
		98836.51048	99	9
		96493.65643	70	7
		121231.1673	29	.9
	97	219055.397	88	8 0
		165254.9456	† 00 0	ė š
		159279.8195	63	3 iii
		137349.805	75	7.
	49	156982.3197	67	9 0
		104997.4742	62	6 6
		245558.4923	99	9
	93 93	267333.6231	74	77
		339611.2444	63	9
	50	115606.7108	79	6
	58 58	14/401.3/16	0/	× 1.
		165629.8893	71	. 7
	52 52	126093.7827	62	.9
		103393.6785	63	9
	0 00	135554.434	27 cr	
		95944.79825	59	. 15
		98633.26296	89	39
		191961.801	96	0,

		I	
Reduction of mass	Reducction of mass	Reducction of mass (%)	Š J
(kg)	(%) respect the initial value	respect the initial value	itera tion
49530.04807	22	22	1
111140.1379	46	46	2
139997.9775	48	48	3
255083.8876	71	71	4
214088.8987	70	70	2
158289.4568	50	50	9
48904.2736	16	16	7
180960.8434	29	67	8
164270.452	65	59	6
217872.5798	64	64	10
80419.325	33	33	11
146975.7816	43	43	12
185032.048	64	64	13
138754.3306	55	55	14
175674.1634	58	58	15
232989.8321	29	29	16
71677.757	31	31	17
80338.0317	31	31	18
90539.81705	36	36	19
144635.8166	39	39	20
134233.5493	50	50	21
186776.1742	57	57	22
277604.7673	89	99	23
60106.87903	17	17	24
64868 17582	23	73	25
199651 0958	3 2	1 2	26
90077 70001			2,7
10347.7.200	n E	u [77
CCLC 0523CC	3 5	3 6	9 6
2200/0.3133	79	92	67
218132.1047	89	89	30
248372.7202	99	99	31
233947.1759	70	70	32
184393.6778	99	99	33
263448.1165	89	89	34
-13851.23428	-2	0	35
198932.472	29	29	36
69789.76156	30	30	37
231453.441	59	59	38
263330.3036	99	99	39
197856.4196	56	26	40
268419.2807	63	63	41
155996.1637	09	09	42
151228.6945	38	38	43
67611.41991	20	20	4
47229.0942	12	12	45
279019 4211	7.1	7.1	46
289173 2078	99	99	47
212075 0266	000	000	
251050 9938	09	9	
204480.0333	9 6	1 0	
294480.0223	70	0/	
265284./08/	7/	7 /	
310966.9936	70	70	
246942.8802	72	72	
220837.7111	70	70	54
192200.9194	99	99	52
		1	

MODE	MODEL B - 2 O	MODEL B - 2 Optimization process (Results 48 variables) Parameters considered: Structural material Steel 8275	esults 48 vari	ables)											
		Cladding material Glass Load case 4 & 5													
Height Lenght Width	EEE	22.63 Z direct 72.57 Y diret 28.25 X direct	Unit: Appartment - Canopy - Appartmen Height m Lenght m	Canopy - Appartm m m	₽	lirect lirct									
Appartment Height Lenght Width	Appartment measurements Height m Enght m Width m	ts 22.63 Z direct 72.57 Y direct 12.1.X direct	Width Roof - high eave Desnivel Roof - Low eave	EEEE	52.45 x direct 8 5 3	direct									
	B - 2: Ste	Step 7 - Before cross section opt.								ld. Family					
Number of iteration	Steel dass	Max. X - Number Y - Number displacement (m) of Frames of Frames (Structure)	ber Max displacement les beams	Beams Top - X direction	Length	Allowed max. Displacement beams (m)	Beams Top - Y Direction Extension	Length Celement (m)	Allowed max. Displacement beams (m)	Beams Top - Y Direction central part top	Length element (m)	Allowed max. Displacement beams (m)	Beams Top - Y Direction central sides	Length Allowed ma element (m) Displaceme	Allowed ma Displaceme beams (m)
	1 \$275	0.064856 5	7 0.001547		8.070	0.04035	9	8.000		42	10.36714286	5 0.051835714	14	10.3671429	0.0518
	€ 1	0.294949 3	12 0.003042 14 0.012734	42	13.450 20.175	0.06725	4.6	8.000	0.0400	48	5.183	0.02	24 28	5.18	0.03
				39	13.450	0.06725	4	8.000		44	6.597272727	7 0.032986364	22	6.59727273	0.0329
	2	1.116553 2	5 0.012579		20.175	0.10088	en e	8.000	0.0400	15	14.514	4 0.07257	10	14.514	0.000
	0 0	0.29517 3		24	13.450	0.06725	0.4	8.000		24			12	0.0	0.0
	0 -	1.1081 2	12 0.012718		20.175	0.10088	en ur	8.000	0.0400	36	6 597777777	5 0.0302375	24	6.0475	0.03
	12	1.1089 2			20.175	0.10088	n en	8.000	0.0400	27	8.063		18	8.06	0.0403
	m •	1.113359 2	6 0.012614 7 0.012652	16	20.175	0.10088	en en	8.000		18	12.095	5 0.051835714	12	12.095	0.00
-	· N	0.295068 3	8 0.003363		13.450	0.06725	4	8.000		32			16	9.07125	0.045
	9	0.108188 4	9 0.001104		10.088	0.05044	v 4	8.000		45	5.183571429	3 0.040316667 9 0.025917857	18	5.18357143	0.0403
-		0.108232 4	13 0.001115		10.088	0.05044	S.	8.000		99	5.5823		26	5.58230769	0.027
- 2	6.0	1.108728 2 0.10818 4	10 0.012714 8 0.001103		20.175	0.05044	en un	8.000		30	7.257	7 0.036285	20	7.257	0.0
7	e	0.108229 4	10 0.001106		10.088	0.05044	r.	8.000		20			20		0.0
. 7	3 2	0.38885 3	7 0.003015		13.450	0.06725	4 4	8.000		28	10.36714286	5 0.051835714	14	10.3671429	0.0518
7 0	4 :	1.119293 2	4 0.012555		20.175	0.10088	ε ι	8.000		12	,	•	« ;		0.09
	. u	0.108162 4	13 0.00175		10.088	0.05044	u w	8.000		9 8	10.36714286	5 0.051835714 2 0.027911538	14	10.3671429	0.0518
2	7	1.109989 2			20.175	0.10088	es ·	8.000		24			16	9.07125	0.045
.7 2	80 m	0.29502 3	13 0.003274		13.450	0.06725	4 4	8.000	0.0400	36	8.063333333	3 0.040316667 2 0.027911538	18	8.06333333	0.0403
· m		0.157972 4			10.088	0.05044	. ru	8.000		25			10		0.0
io m	2	0.170862 4	12 0.0011632		10.088	0.05044	in in	8.000		20	18.1425	5 0.0907125	8 24	18.1425	0.03
, m	1 10	0.137617 4			10.088	0.05044	י אי	8.000		30			12		
en 6	4 1	0.108242 4	14 0.001114		10.088	0.05044	LO 4	8.000		70	5.183571429	9 0.025917857	28	5.18357143	0.0259
, en	1 9	0.050644 5	0.000678		8.070	0.04035	9	8.000		54	8.063	0.0	18	8.06	0.0403
, e	8	0.044529 5	11 0.000661		8.070 8.070	0.04035	9	8.000		8 4	9.07125 6.597272727	5 0.04535625 7 0.032986364	16 22	9.07125 6.59727273	0.0329
m «	0.0	0.083341 5	5 0.004863		8.070	0.04035	9 4	8.000		30	-	4 0.07257	10	-	9.0
4.4	0 €	0.044486 5	12 0.000602		8.070	0.04035	9	8.000	0.0400	72	Ψ	0	20 24	•	0.03
4 4	2 5	0.294976 3	15 0.003237		13.450	0.06725	4 4	8,000	0.0400	60	4.838	3 0.02419	30	4,838	0.0
4	4 4	0.04454 5			8.070	0.04035	, 6	8.000			5.58		26	5.5	0.0279
4 4	. v	0.044505 5	14 0.000599		8.070	0.04035	9 17	8.000			5.183571429	0.0		5.18357143	0.0259
4 4	47	0.04479 5	15 0.012725	385	8.070	0.04035	9 **	8.000	0.0400	90		0.02419	30		8 8

DEL B - 2 Optimization proces
sidered: Structural mate Steel s275
Cladding materi Glass
Load case 4 & 5

	ĺ	tion ross (%)	23	569	904	241	241	911	241	904	88	902	206	241	88	241	902	88	88	241	31/	88	904	906	241	129	139	112	88	67	4 7	36	89	36	241	67	36	98 39	36
2683	İ	Utilization s) ratio cross section (%)	5714	3279	1921	1683	3939	2964	7387	3832	7801	3442	2202	9396	6609	7929	3367	5229	3118	0163	866	1885	3285	0162	797	1714	1433	7517	5271	5112	3674	60.6	9476	2879	1941	3195	2462	1433 5804	2793
315252.5683		Total mass (kg)	413176.571	340763.827	386007.092:	383342.468	396948.393	323768.296	344279.996	393494.383	439332.780	352910.344	343297.220	356419.939	412650.509	402751.792	366346,936	399480.522	425939.811	360330.016	332248.099	402819.188	375211.328	339587.016	391910.79	383361.171	374845.443	387103 325	446057.627	397242.5112	422/69.09	449759.09	393647.647	436226.287	413631.194	385131.919	445090.679	455886.443	466721.2793 396841.9281
s [kg]=		Total number of elements	135	76	144	134	146	99	94	132	178	107	91	122	157	190	123	148	175	118	86 74	140	157	113	191	119	107	212	240	154	186	232	143	219	226	132	268	284	301
Minimal Mass [kg] =		U_max < U_max_allow ?	ŏ	Not	Not	Not	Not	to 1	ğ ğ	Not	ŏ	to to	N TO	Not	ŏ	Not Not	S to	ŏ	ŏ	Not	S S	ŏ	Not	z z	g to	Not	Not	ŏŞ	ŏ	ŏδ	š <mark>š</mark>	ŏ	ŏ	ŏδ	No	ŏ	ŏ	ŏŏ	Not Not
		max. Allow displacement	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252
		U_max < U_max_allow ?	ŏ	ŏ	ŏ	ŏ	ŏ	ŏδ	š š	ŏ	ŏ	žδ	έĕ	ŏ	ŏ	žδ	á ă	ŏ	ŏ	ŏ	ž č	šš	ŏ	ŏδ	š š	ŏ	ŏ	šč	ŏ	ŏδ	š č	ŏ	ŏ	ŏδ	ž č	ŏ	ŏ	ŏŏ	ŏŏ
		Allow Max displacement columns	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921
		Allowed max. Displacement columns (m)	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921
		Length element (m)	27.63	27.63	27.63	27.63	27.63	27.63	27.63	27.63	27.63	27.63	27.63	27.63	27.63	27.63	27.63	27.63	27.63	27.63	27.63	27.63	27.63	27.63	27.63	27.63	27.63	27.63	27.63	27.63	27.63	27.63	27.63	27.63	27.63	27.63	27.63	27.63	27.63
		Columns Front Ablock	2	e	S	9	7	00 0	9	11	12	13	15	16	17	18	20	21	22	23	24	26	27	28	30	31	32	33	35	36	38	39	40	41	43	44	45	46	48
		Allowed max. Displacement columns (m)	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210
		Columns back Length element (m)	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630
			11 6	4 4	3 1	21 4	4	3	4 3	3	5	e e	3 0	21 4	5	4 4		5	5	4	4 %	5	3	3	4 4	5	5			9 0			9	9 9	. 4		9	2 0	3 6
		Allowed max. Displacement columns (m)	0 0.0921						0.0921				0 0.0921									0 0.0921			0.0921			0 0.0921				0.0921			0.0921		0 0.0921		
		Length element (m)	6 27.6300	4 27.6300	3 27.6300	4 27.6300		3 27.6300	4 27.6300	3 27.6300	5 27.6300	3 27.6300	3 27.6300	4 27.6300	5 27.6300	27.6300	3 27.6300	5 27.6300	5 27.6300	4 27.6300	3 27.6300	5 27.6300	3 27.6300	3 27.6300	4 27.6300	5 27.6300	5 27.6300	5 27.6300	5 27.6300	6 27.6300			6 27.6300	5 27.6300	4 27,6300	6 27.6300		5 27.6300	6 27.6300 3 27.6300
_		Columns front						0.6																				0.0						0.6					
		Allowed max. Displacement columns (m)		0.0100				0.0100				0.0100				0.0100				0.0100				0.0100				0.0100		0.0100				0.0100				0.0100	
3		Length element (m)	3:000					3.000			3.000	3.000	3.000	3.000		3.000				3.000	3.000	3.000		3.000		3.000		3.000		3.000		3.000		3.000				3,000	
E		Columns Laterals rigth short	7	2 22	14	10	11	v.	9	12	11	5		80	6	14	1 01	8	10		4 4		13		13 4	5	4	12	14	9 0	n 00	11	S.	1 10	15	4	13	15 15	21 21
angle	opt.	Allowed max. Displacement beams (m)	0.026666667	0.02666665	1	0.02666665	0.02666666	0.026666667	0.026666667	0.026666667	0.02666666	0.02666666	0.02666665	0.02666666	0.02666666	0.02666666	0.02666665	0.02666666	0.02666665	0.02666666	0.02666666	0.02666665	0.026666657	0.02666665	0.026666667	0.026666667	0.02666666	0.02666665	0.02666665	0.026666667	0.02666666	0.02666665	0.02666665	0.02666665	0.02666665	0.02666665	0.026666657	0.026666667	0.02666666
	ss section	Length A element (m) b	8.00	8.00	200	8.00	8.00	8.00	8.00	8.00	8.00	8.00	8.00	8.00	8.00	8.00	8.00	8.00	8.00	8.00	8.00	8.00	8.00	8.00	8.00	8.00	8.00	8.00	8.00	8.00	8.00	8.00	8.00	8.00	8.00	8.00	8.00	8.00	8.00
12.1 X direct	- Before cr	Columns Laterals left short	7	5 21		10	11	υ.	11 9	12	11	5	7	00	6	14	10	00	10	7	4 4		13	00 (13	S	4	12	14	9 0	n «	11	S	10	15	4	13	14	15
E	B - 2 : Step 7 - Before cross section opt.	Max displacement columns	0.044936	0.032771	1000	0.03192	0.031864	0.053659	0.032512	0.052324	0.052181	0.052572	0.052882	0.032133	0.052413	0.031855	0.052469	0.052523	0.052278	0.032252	0.033346	0.052647	0.052322	0.05272	0.031865	0.053234	0.053856	0.052179	0.052127	0.045259	0.044835	0.044529	0.045422	0.044599	0.031819	0.045998	0.04454	0.044505	0.044479
Width		No it	-	2 %	1	Ŋ	9	7	xo on	10	11	12	14	15	16	17	19	20	21	22	23	25	26	27	29	30	31	32	34	35	37	38	39	40	42	43	44	45	47

MODEL	B - 2 Op	Q	(Result	s 48 variabl	les)									
Parameters considered	nsidered:	Structural materia Steel s275 Cladding material Glass Load case 4 & 5												
Height r Lenght r Width r	EEE	22.63 Zdírect 72.57 Y dirct 28.25 X direct	Unit: Appartn Height Lenght	Jnit: Appartment - Canopy - Appartmen Height m .enght m		22.63 Z direct 72.57 Y dirct								
Appartment m Height r	measurements m	22.63 Z direct	Width m (roof) - low eam	E E E	52.45	52.45 X direct 8								
	E E	12.1 X direct		€ €	n en									
	B - 2 : Step	Step 9 - After cross section opt	opt.						ld . Family					
Number of iteration	Steel dass	Max. X - Number of Y - Number of frames for fearers	Y - Number of Frames	Max displacement beams	Beams Top - X direction	Length element (m)	Allowed max.	Beams Top - Y Direction	Length Allowed max.		op- tion part	All Length mi	Allowed max.	Beams Direction
		(m) (structure)					beams (m)	Extension	beams (m				t beams (m)	š
110	s275	0.064856 5	7	0.021193		8.070	0.04035	9 4	8.000	0.0400		10.3671429 (0.05183571	
3 8		0.294949	12				0.06725	1 4		0.0400			0.0302375	
4 1		1.107277 2	14	0.014698	32	20.175	0.10088	e 4		0.0400		5.18357143 (0.02591786	
9		0.294995 3	11		3 8		0.06725	4		0.0400			0.03298636	
7		1.116553 2	2.1	0.014616	14	20.175	0.10088	60 6	0008	0.0400		14.514	0.07257	
9 6	~	0.29517 3	9		24		0.06725	0.4		0.0400			0.060475	
10		1.1081 2	12	0.014682	8 6	20.175	0.10088	m u	8.000	0.0400		6.0475	0.0302375	
12		1.1089	6		22 22		0.10088			0.0400			0.04031667	
13		1.113359 2	9 1	0.014647	16		0.10088	en e		0.0400			0.060475	
15		0.295068 3	- 80	0.010028			0.06725	υ 4		0.0400		9.07125	0.04535625	
16		0.108188 4	6	0.016735		10.088	0.05044	50 4	8.000	0.0400			0.04031667	
18		0.108232 4	13				0.05044	4 10		0.0400		5.1835/143 U	0.02791186	
19		1.108728 2	10				0.10088	8		0.0400			0.036285	
21 20		0.10818 4	10	0.014926	8 8	10.088	0.05044	יט יט	000.8	0.0400	20 40	9.07125 0	0.04535625	
22		0.295249 3	7		22		0.06725	4		0.0400			0.05183571	
24		0.388885 3	4 4	0.011422			0.06725	4 K	000.8	0.0400	16	18.1425	0.0907125	
25		0.108162 4	7				0.05044	r.		0.0400			0.05183571	
27		1.109989 2	13	0.014697	8 8	20.175	0.10088	m m	8:000	0.0400		9.07125	0.02791154	
28		0.29502 3	6		83		0.06725	4		0.0400		_	0.04031667	
ଅନ		0.157972 4	13	0.009154	28 45	13,450	0.05044	4 ru	8.000	0.0400		5.58230769 C	0.02791154	
31		0.170862 4	4				0.05044	ľ		0.0400	20		0.0907125	
32		0.10824 4	12	0.011426		10.088	0.05044	un ur		0.0400	9 09	12.095	0.0302375	
3 %		0.108242 4	14		4 49		0.05044	n in		0.0400			0.02591786	
35		0.082663 5	9 0	0.013468	9 5	8.070	0.04035	9 4	8.000	0.0400		12.095	0.060475	
37 %		0.058982 5	n oo	0.020834	S 28		0.04035	9		0.0400	48 6.0		0.04535625	
8 8		0.044529 5	11	0.015368		8.070	0.04035	9 9	8.000	0.0400		6.59727273 (0.03298636	
9 9		0.044599 5	10				0.04035	9		0.0400	9	7.257	0.036285	
41		0.044486 5	12	0.017211	8 5	13.450	0.04035	9 4	8.000	0.0400	72	6.0475	0.0302375	
43		0.08209 5	4		8 8		0.04035	9		0.0400			0.0907125	
44 45		0.04454 5	13	0.015535	£ 8	8.070	0.04035	9 9	8.000	0.0400		5.58230769 0	0.02591154	
46		0.108229 4	15			1	0.05044	, ru		0.0400			0.02419	
47		0.044479 5	15	0.013939			0.04035	9 6	8.000	0.0400	90	4,838	0.02419	

	ODEL B - 2 Optimization process (Results 48 variables)				
2	proce	Steel s275	Glass	4 & 5	
: MODEL B -	Optimizatior	Structural mat	Cladding mate	Load case	
APPENDIX D - 3: MODEL B - 2	MODEL B - 2 (Parameters considered:			

	Utilization ratio cross section (%)	23	269	241	904	241	241	911	904	241	904	88 8	000	407	241	88	241	88	902	88	88	241	317	914	88 8	5 6	241	241	129	139	88	112	8 19	41	48	39	89	36	38	241	79	g 15	R 88	36	904
ļ		9449	1843	.489	1911	0044	1453	1798	3248	8666	179	57.75	106	7.45	9468	:412	3305	3076	3944	1054	8579	159	1614	2201	.025	000	5587	3535	13.75	.671	1183	619	1156	22.0	2959	.326	3827	1281	209	1031	1821	5005	3363	5753	333
	Tota		m		,	,	,	,	,	,	,	343727.377		,		325910.2412	372617.8305	338860.8076	372745.894	318644.405		,		m	328675.02			,	æ		,	336486.2619			7				.,	., .	353392.1821		,,	,	
	Total number of elements	135	84	16.	150	142	15,	72	127	10	138	188	87	7.6	130	167	198	221	129	158	185	126	94	80	150	103	154	199	129	117	222	145	166	212	198	244	15.	231	262	234	144	ő Ž	275	313	203
	U_max < U_max_allow ?	ю	Not	X N	Not	Not	Not	ŏ	Not	ŏ	Not	ŏ	ŏ	Not	Not	Not	ŏ	No.	Not	Not	Not	Not	ŏ	Ş	έŏ	ŏ	Ж	ŏ	ŏ	ŏ	ŏ;	Not	ŏδ	śč	έĕ	ŏ	Not								
-	max. Allow displacement	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12232	0 12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12232	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252	0.12252
	U_max < U_max_allow ?	ОК	OK	ŏ	o,	ŏ	OK	ŏ	ŏ	OK.	ŏ	× ×	ž č	ž	ok Ok	ОК	OK	ОК	ОК	ОК	ОК	OK	OK	o _K	o X	5 6	žě	o X	ОК	OK	ŏ	ŏŏ	žŏŏ	ОК	ОК	OK	OK	OK	ŏ) OK	ŏ, ŏ	ž č	žŏ	ò	ОК
	Allow Max displacement columns	0.0921	0.0921	0.0921 OK	0.0921	0.0921	0.0921 OK	0.0921 OK	0.0921 OK	0.0921 OK	0.0921 OK	0.0921 OK	0.0921 OK	0.0921 OK	0.0921 OK	0.0921 OK	0.0921 OK	0.0921 OK	0.0921 OK	0.0921	0.0921 OK	0.0921 OK	0.0921 OK	0.0921 OK	0.0921 OK	0.0921 OK	0.0921 OK	0.0921 OK	0.0921	0.0921 OK	0.0921	0.0921 OK	0.0921	0.0921	0.0921 OK										
	Allowed max. Displacement columns (m)	0.0754	0.0754	0.0754	0.0754	0.0754	0.0754	0.0754	0.0754	0.0754	0.0754	0.0754	42.00	0.0754	0.0754	0.0754	0.0754	0.0754	0.0754	0.0754	0.0754	0.0754	0.0754	0.0754	0.0754	47.00	0.0754	0.0754	0.0754	0.0754	0.0754	0.0754	0.0754	0.0754	0.0754	0.0754	0.0754	0.0754	0.0754	0.0754	0.0754	0.0754	0.0754	0.0754	0.0754
	Length element (m)	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.03	22 63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.03	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.63	22.03	22.63	22.63	22.63
	Columns Front Ablock	2	3	4	2	9	7	00	6	10	11	12	14	15	16	17	18	19	20	21	22	23	24	25	26	7 00	29	30	31	32	33	34	36	37	38	39	40	41	42	43	44 4	45	47	48	49
•	Allowed max. Displacement columns (m)	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210	0.09210
	Length element (m)	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.030	27.630	27.630	27.630	27.630	27.630	27.630	27,630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630
	Columns back	9	00	00	9	00	∞	9	9	00	9 9	10	9 4	9 (2	00	10	80	10	9	10	10	80	00	9	10	0 (0 00	00	10	10	10	10	12	12	12	12	12	12	12	ω :	12	12	10	12	9
	Allowed max. Displacement columns (m)	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921	0.0921				0.0921		0.0921							0.0921			_		_		0.0921			0.0921	0.0921	_			0.0921	0.0921				
	Length element (m)	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630					27.630	27.630	27.630	27.630	27.630	27.630	27.030	27.630	27.630	27.630	27.630	27.630	27.630	27,630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630	27.630
	Columns	9	00	00	9	80	00	9	9	00	9 9	10	9	9	. 00	10	80	10	9	10	10	00	00	9	10		0 00	00	10	10	10	0 0	12	12	12	12	12	12	12	xo ç	77 22	1 1	1 01	12	9
•	Allowed max. Displacement columns (m)	0.0100	0.0100	0.0100	0.0100	0.0100	0.0100	0.0100	0.0100	0.0100	0.0100	0.0100	0.0100	0.0100	0.0100	0.0100	0.0100	0.0100	0.0100	0.0100	0.0100	0.0100	0.0100	0.0100	0.0100	0.0100	0.0100	0.0100	0.0100	0.0100	0.0100	0.0100	0.0100	0.0100	0.0100	0.0100	0.0100	0.0100	0.0100	0.0100	0.0100	0.0100	0.0100	0.0100	0.0100
	Length element (m)	3.000	3.000	3.000	3.000	3.000	3.000	3.000	3.000	3.000	3.000	3.000	3.000	3 000	3.000	3.000	3.000	3.000	3.000	3.000	3.000	3.000	3.000	3.000	3.000	3.000	3.000	3.000	3.000	3.000	3.000	3.000	3.000	3.000	3:000	3.000	3.000	3.000	3.000	3.000	3.000	3.000	3.000	3.000	3.000
	Columns Laterals rigth short	7	S	12	14	10	11	S	11	9	12	11			. 00	6	14	13	10	00	10	7	4	4	7 ,	n o	0 0	13	5	4	12	9 -	9	6	80	11	r.	10	12	15	4 6	14	15	15	15
opt.	Allowed max. Displacement beams (m)	0.026666667	0.026666667	0.026666667	0.026666667	0.026666667	0.026666667	0.026666667	0.026666667	0.026666667	0.026666667	0.026666667	0.022000007	0.02666667	0.026666667	0.026666667	0.026666667	0.02666667	0.026666667	0.026666667	0.026666667	0.026666667	0.026666667	0.026666667	0.026666667	0.02000000	0.02666667	0.026666667	0.026666667	0.026666667	0.02666667	0.026666667	0.026666667	0.026666667	0.026666667	0.026666667	0.026666667	0.026666667	0.026666667	0.026666667	0.026666667	0.026666667	0.026666667	0.026666667	0.026666667
s section	Length element (m)	8.00	8.00	8.00	8:00	8.00	8.00	8.00	8:00	8.00	8.00	8.00	8.00	8 00	8.00	8.00	8.00	8.00	8.00	8.00	8.00	8.00	8.00	8.00	8:00	0.00	8.00	8.00	8.00	8.00	8.00	8.00	8.00	8.00	8.00	8.00	8.00	8.00	8.00	8.00	8.00	8 00	8.00	8.00	8.00
- After cros	Columns Laterals left short	7	'n	12	14	10	11	'n	11	9	12	11	n va		- 00	6	14	13	10	80	10	7	4	4	7 ;	n o	0 0	13	5	4	12	9 1	t 9	6	8	11	'n	10	12	15	4 0	14	15	15	15
B - 2 : Step 9 - After cross section opt.	Max displacement columns	0.071539	0.031474	0.030409	0.045561	0.030464	0.030414	0.047123	0.04561	0.031124	0.045592	0.045473	0.045613	0.046204	0.030687	0.045703	0.030406	0.045451	0.045707	0.045857	0.045569	0.030805	0.032183	0.048116	0.046015	0.04033	0.030569	0.030428	0.046739	0.047653	0.04549	0.046334	0.053918	0.077302	0.077438	0.077181	0.054213	0.077219	0.07716	0.030379	0.058264	0.077204	0.045433	0.077164	0.045561
<u>a</u>	Number of iteration	1	2	m	4	S	9	7	00	σ .	10	11	13	1 1	15	16	17	18	19	20	21	22	23	24	25	2 6	28	29	30	31	32	33	35	36	37	38	39	40	41	42	43	45	46	47	48

APPENDIX D		L B - 2								
MODEL B		Optimization process	(Result	(Results 48 variables	es)					
Parameters considered	idered:	Structural material Cladding material Load case	Steel s275 Glass 4 & 5							
Height Lenght Width	EEE	22.63 72.57 28.25	22.63 Z direct 72.57 Y diret 28.25 X direct	Unit: Appartment - Canopy - Appartment Height Lenght m	anopy - Appartn m m		22.63 Z direct 72.57 Y diret			
Appartment measurements	surements			Width	Ε	52.45	52.45 X direct			
Height Lenght Width	EEE	22.63 72.57 12.1	22.63 Z direct 72.57 Y dirct 12.1 X direct	(roof) - Iow eave Desnivel angle	EEE	3 2 2				
	NOSIGNARYOS	NOSIG				Gray	misson continuous social	motion	After	the notation of
	COMPA	KISON				Before	Before cross section optimization	ization	Arter	After cross section opti
Reducction of mass (kg)	Reducction of mass (%) respect the initial value	Reducction of mass (%) respect the initial value	Nº of iteration		Nº of iteration	Total mass (kg)	Utilization ratio cross section (%)	Utilization ratio cross section (%)	Total mass (kg)	Utilization ratio cross section (%)
103388.0265			1		1	413176.5714	26	26	309788.5449	
-14290.65643	4		2		2	340763.8279	23	23	355054.4843	.,
33888.1018 -8438.024014			n 4		n 4	386007.0921	22	22	3/6/42.489	
31483.46394			S		2	383342.4683	22	22	351859.0044	.,
33569.24855			9		9	323768.3939	22	22	363379.1453	
-5605.18609			8		80	379879.7387	48	48	385484.9248	
7630.99630:	E 20	0 0	9		9	344279.9961	21	21	336648.9998	
95605.4026	- 10		11		11	439332.7801	90	30	343727.3775	•
-6729.613097	-2	0 0	12		13	352910.3442	44 44	44	359639.9573	
-12635.85431			14		14	343297.2202	45	45	355933.0745	
25553.8928	8 1		15		15	356419.9396	22	22	330866.0468	
30133.96236	7 2		17		17	402751.7929	24 24	24	372617.8305	,
96479.88769			18		18	435340.6953	. e	30	338860.8076	, 4
-6398.957663			ឡ ខ		19	366346.9367	46	46	372745.8944	7
80836.11746			27		20	399480.52 <i>2</i> 9 425939.8118	ଳ ନ	30	334033,3579	, ,
16753.8004			72		22	360330.0163	21	21	343576.2159	
-32986.0615			23		23	332248.0998	8	30	365234.1614	., <
74144.1635	18		25		25	402819.1885	31	31	328675.025	. 7
-7496.7095			56		56	375211.3285	51	51	382708.038	,
-8886.780882			22 82		27	339587.0162	44 5	21	348473.7971	•
27646.94349			8 20		8 82	391910.797	23	23	364263.8535	,,
27761.73389			30		30	383361.1714	36	36	355599.4375	7
-8945.22762			H (31	374845.4433	44 %	44	383790.671	
50617.06384			33 33		33	387103.3257	3 8	33	336486.2619	, ,
99840.07549			34		8	446057.6271	98	30	346217.5516	•
82317.39553			32		35	397242.5112	28	28	314925.1156	7 '
1118675,997	7		% %		37	422789.6954	25	25	298337.5567	
127693.7639			38		38	449759.09	25	25	322065.326	
67896.26492			8		£ 3	393647.6476	31	31	325751.3827	., ,
123649.2298	8 8	28 88	40		40	455225.2879	4 X	25	332393.7607	- 1-
32010.091			45		42	413631.1941	24	24	381621.1031	
31739.73743			43		43	385131.9195	40	40	353392.1821	
131217.3632			4 4		4 4	445090.6797	25 25 27	25	313873.3165	
102786.8441			46		46	456813.6804	31	31	354026.8363	
138921.704			47		47	466721.2793	25	25	327799.5753	
-9533, 105196	,	0	48		48	396841.9281	25	55	406375.0333	

Reducction of Reducction of mass (%) (%) respect the initial value (%) respect to initial value (%) respect t	Boducetion of more	
		Nº of iteration
	5 25	1
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258921.704	,	/48

APPENDIX E: OUTPUT FROM GRASSHOPPER. SECTION FORCES AND VERIFICATIONS

APPENDIX E-1: MODEL A 1st Loop - MODEL A - Verifications Step 9

MODEL A - step 9 - verification of the most unfavorable element

'Most optimal cases'' Iteration Nº	Iteration Nº	25	X=4 Y	Y = 6				Results - Input	Results - Input from grasshopper		
Element_ID	Span direction	Material	Type of profile	Nº of frames	Nº of Span of frames elements [m]	Nº Elem	Deflection [m]	Max Normal Force	Max Shear force	Max Bending moment	Max. Stress
							[w] n	Nd [KN]	Vd [KN]	Md [KNm]	σ [kN/cm2]
Beams Top - X direction	X direction	5235	HEB 220	7	7.06	28	0.006059	0	41.120432	50.284216	13.673
Beams Top - Y Direction	Y direction	5235	HEB 550	2	12.10	30	0.041762	0	96.46713	195.660239	7.872
								Results - Input	Results - Input from grasshopper		
Element_ID	Span direction	Material	Type of profile	Number of frames	Length [m]	Nº Elem	Deflection [m]	Max Normal Force	Max Shear force	Max Bending moment	Max. Stress
							[w] n	Nd [KN]	Vd [KN]	Md [KNm]	σ [kN/cm2]
Columns Laterals	z direction	S235 H	НЕВ700	7	22.63	14	0.008472	-15.229993	6.567251	33.822428	0.921592044
Columns front/back	z direction	S235 H	НЕВ160	5	22.63	9	0.023459	-13.422006	16.613963	70.707307	21.71584462

Verifications								
Axial Buckling Resistance	Shear resistance	Flexural resistanc e	Flexural resistance (V+M)	Lateral Buckling Resistance	Allow Max. Stress	Max.Def		Element_ID
N _{RD} [kN]	V _{RD} [kN]	M _{C,RD} [kN]	M _{v,RD} [kN]	M _{RD,LT} [kN]	a [kN/cm2]	[m] waw [m]	[m]	
752.7319642	OK 236.3046539	OK 194.345	OK Not needed	- 60.81220998	OK 27.5	OK 0.03	0.0353125 OK Beam	OK Beams Top - X direction
3814.358746	OK 932.7815287	OK 1313.885	OK Not needed	- 746.2092612	OK 27.5	OK 0.0	0.060475 OK Direct	OK Beams Top - Y Direction

	Element_ID		0.07543 OK Columns Laterals	0.07543 OK Columns front/back
	Max.Def	[m] xem_u	0.075	0.075
	Allow Max. Stress	σ [kN/cm2]	27.5 OK	27.5 <mark>OK</mark>
	Column Buckling Resistance	M _{RD,LT} [kN]	631.8293756 OK	1.327816101 not
Verifications	Flexural resistance (V+M)	M _{v,RD} [kN]	Not needed	Not needed
	Flexural resistanc e	M _{C,RD} [kN]	yo .	YO -
	Axial Buckling Resistance, z	N _{RD,z} [kN]	OK 403.3273402 OK	OK 9.70679923 not
	Axial Buckling Resistance, y	N _{RD,y} [kN]	2853.861876	505.2937559

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First model - before Cross optimization First model - after Cross optimization Parameters considered:

1st Loop - MODEL A - Section forces (Output from GH) - Step 9 (1st Loop)

- aitei Cioss optiiiizatioi	IIZation		ţ								
considered:		Structural mater	Stee								
		Cladding materia	eria Glass								
		Load case	4								
		Iteration Nº	25								
	Beams top X direction	ction		Beams top Y direction	irection		Columns laterals			Columns Front/back	
	Normal forces	Bending	Shear Force	Normal forces	Bending S moment	Shear Force	Normal forces Be	Bending moment	Shear Force	Normal forces	Shear Force
		0 19.55473	73 15.526804		0 89.861794	43.258937	-15.229993	33.822428	6.56725	-31.297584 70.707	16.613963
		0 19.554725			950669.68 0	40.419795	-4.388199	2.339347	0.155021		12.390569
		0 17.497318			0 93.794233	45.358691	-4.388199	2.523963	0.167255		16.613961
		0 17.497319	19 14.822916		0 93.807542	45.520806	-4.388199	2.522621	0.167166	-13.422006 31.694	7.42684
		0 46.667647	47 37.710775		0 94.23069	40.397607	-4.388199	2.525563	0.167361	-12.094106 24.88	5.593712
		0 46.666551	51 37.708857		0 94.47902	45.635476	-4.388199	2.532059	0.167792	-13.422006 31.694	7.426835
		0 46.640831	33.471043		0 89.861732	43.258929	-10.827493	19.106087	3.113566		
		0 46.636256	56 33.470514		0 89.698994	40.41977	-15.229993	33.822428	6.567251		
		0 50.284216	16 40.678182		0 93.792751	45.358525	-4.388199	2.338424	0.15496		
		0 49.42504	04 41.120432		0 93.808054	45.520729	-4.388199	1.923265	0.127448		
		0 50.281543	43 36.072387		0 94.229856	40.397644	-4.388199	1.920629	0.127274		
		0 49.425848	48 35.951101		0 94.48053	45.635606	-4.388199	1.924494	0.12753		
		0 50.28374	74 40.678661		0 195.656346	93.230283	-4.388199	1.934837	0.128216		
		0 49.423872	72 41.119041		0 195.660239	90.277425	-10.827493	19.102232	3.113459		
		0 50.28373	73 36.072646		0 188.165609	96.46713					
		0 49.423881	81 35.950892		0 165.614152	74.354259					
		0 50.283756	56 40.678649		0 245.839688	104.736061					
		0 49.423888	88 41.11903		0 245.829795	114.289748					
		0 50.283651	51 36.072619		0 173.688927	82.07746					
		0 49.423826	26 35.950873		0 173.686971	76.772031					
		0 50.291197	97 40.676464		0 178.886663	86.417906					
		0 49.433909	09 41.11892		0 178.898145	86.668704					
		0 50.260496	36.068055		0 179.168526	76.354156					
		0 49.407918	18 35.947216		0 179.174524	86.685502					
		0 24.67075	75 19.626014		0 195.656391	93.23028					
		0 24.669547	47 19.624943		0 195.660283	90.277444					
		0 22.253541	41 17.920452		0 188.165084	96.46696					
		0 22.25164	54 17.92019		0 165.613614	74.354271					
					0 245.84034	104.735939					
					0 245.828097	114.289595					

APPENDIX E- 1: MODEL A

2nd Loop - MODEL A - Verifications

MODEL A - step 9 [2nd loop] (11b - after modifications) - verification of the most unfavorable element

'Most optimal cases"	Iteration Nº	25	X=4 Y=6	Y = 6			Re	sults - Input f	Results - Input from grasshopper		
Element_ID	Span direction	Material	Type of profile f	Nº of irames	Span of elements [m]	Span of Total number elements [m] of elements	Max Deflection [m] Normal Force	Max Normal Force	Max Shear force	Max Bending moment	Max. Stress
							[w] n	Nd [KN]	Vd [KN]	Md [KNm]	σ [kN/cm2]
Beams Top - X direction	X direction	5275	HEB180	7	7.06	28	0.013	0	24.440	28.994	13.623
Beams Top - Y Direction	Y direction	5275	HEB360	2	12.10	30	0.026	0	83.275	174.768	14.567

							Re	sults - Input fi	Results - Input from grasshopper	_	
Element_ID	Span direction	Material	Type of profile	Nº of frames	Length [m]	Total number of elements	Deflection [m]	Max Normal Force	Max Shear N force	Max Bending moment	Max. Stress
							[m] n	Nd [KN]	vd [kn]	Md [KNm]	o [kN/m2]
Columns Laterals	z direction	5235	HEB450	7	22.63	14	0.009389	0.009389 -22.267863	13.325585	51.226479	4.750955499
Columns front/back	z direction	5235	HEB340	5	22.63	9	0.003957	0.003957 -17.456779	18.736635	74.659846	6.924262859

				Verifications		
Element_ID	Axial Buckling Resistance	Shear resistance	Flexural resistance	Lateral Buckling Resistance		Max.Def
	N _{RD} [kN]	V _{RD} [kN]	M _{C,RD} [kN]	M _{RD,IT} [kN]	σ [kN/cm2]	u_max [m]
Beams Top - X direction	442.929 OK	202.433 OK	132.385 OK	32.678 OK	27.5 OK	0.035 OK
Beams Top - Y Direction	1054.740 OK	595.392 OK	737.825 OK	214.720 OK	27.5 OK	0.060 OK

	N _{RD} [kN]	V _{RD} [kN]	M _{C,RD} [kN]		M _{RD,LT} [kN]	9	σ [kN/cm2]	u_max [m]	
Beams Top - X direction	442.929 OK	202.433 OK	132.385 OK		32.678 OK		27.5 OK	0.035 OK	
Beams Top - Y Direction	1054.740 OK	595.392 OK	737.825 OK		214.720 OK		27.5 OK	0.060 OK	
				Veri	Verifications				
Element_ID	Axial Buckling Resistance, y	Axial Buckling Resistance,	Flexural resistance	ŭ	Column Buckling Resistance			Max.Def	
	N _{RD,y} [kN]	2 N _{RD,2} [kN]	M _{C,RD} [kN]		M _{RD,LT} [kN]	9	σ [kN/m2]	u_max [m]	
Columns Laterals	983.6066639	OK 562.92504	- YO	ŏ	229.0277405	Ä	27.5 OK	0.07543	ŏ
Columns front/back	771.093481	OK 263.86749	OK	ĕ	154.6333907	ŏ	27.5 OK	0.07543	ŏ

		×	*									
First model - before Cross optimization	timization											
First model - after Cross optimization	mization		4 6									
Parameters considered:		Structural ma	ma Steel s275									
		Cladding mat	lat Glass									
		Load case	4									
		Iteration Nº	□ 25									
	Beams	Beams top X direction	c	Be	Beams top Y direction	tion		Columns laterals			Columns Front/back	t/back
_	Normal forces	Bending	Shear Force	Normal forces	Bending moment	Shear Force	Normal forces Bending moment	ending moment	Shear Force	Normal forces	Bending moment	Shear Force
		0 6.980397	97 5.604381	0	45.705069	20.933342	-22.267863	51.226479	13.325585	-17.46	74.659846	18.736635
				0	61.819497	29.215155	-22.267863	19.508662	1.290535	-17.46	53.686577	13.657517
		0 8.828845	15 7.426909	0	65.810048	32.130934	-22.267863	24.747564	1.637008	-17.46	74.659838	18.736633
		0 8.828846	16 7.426909	0	66.220951	32.475674	-22.267863	24.746041	1.636906	-17.46	74.659846	18.736635
		•	18.0643	0	67.749792	29.187793	-22.267863	24.87474	1.645452	-17.46	53.686577	13.657517
		0 21.379605		0	73.626021	35.017341	-22.267863	25.426014	1.682058	-17.46	74.659838	18.736633
		0 21.375703	17.858869	0	45.705069	20.933341	-22.267863	33.744921	7.36045	-17.46	47.919386	10.308147
		0 21.375699	99 17.858867	0	61.819497	29.215155	-22.267863	51.226479	13.325585	-17.46	44.371715	8.812853
		0 28.992376	76 24.440022	0	65.810048	32.130934	-22.267863	19.508655	1.290535	-17.46	47.919381	10.308146
		0 28.992369	59 24.440015	0	66.220951	32.475674	-22.267863	24.747557	1.637007	-17.46	47.919386	10.308147
		0 28.991465	55 24.035365	0	67.749792	29.187793	-22.267863	24.746033	1.636906	-17.46	44.371715	8.812853
		0 28.991458	58 24.03536	0	73.626021	35.017341	-22.267863	24.874731	1.645452	-17.46	47.919381	10.308146
		0 28.991978	78 24.439964	0	101.145083	44.67901	-22.267863	25.426005	1.682058			
		0 28.991971	71 24.439957	0	145.118408	67.487665	-22.267863	33.744921	7.36045			
		0 28.991975	75 24.035474	0	145.117999	70.266135						
			58 24.03547	0	116.256209	53.053196						
		0 28.991973	73 24.439958	0	174.768395	74.388639						
		0 28.991969		0	174.763883	83.275375						
		•		0	85.422545	39.161767						
		0 28.991955	55 24.035462	0	116.334154	54.590346						
		0 28.99413	13 24.440355	0	123.874995	60.197483						
		0 28.994126	26 24.440348	0	123.8833	60.856232						
		0 28.989538	38 24.034958	0	119.479227	53.240184						
		0 28.989535	35 24.034957	0	119.481882	58.427395						
		0 15.42439	39 12.914685	0	101.145074	44.679006						
		0 15.424387	37 12.914681	0	145.1184	67.487661						
		0 15.050439	39 12.689115	0	145.117991	70.266132						
		0 15.050438	38 12.689114	0	116.256203	53.053192						
				0	174.768377	74.388631						
				0	174.763865	83.275367						

APPENDIX F: SCRIPT IN GRASSHOPPER DEFINITION

OVERVIEW OF SCRIP DEFINITION (EXAMPLE)

Scripts defined in GH:

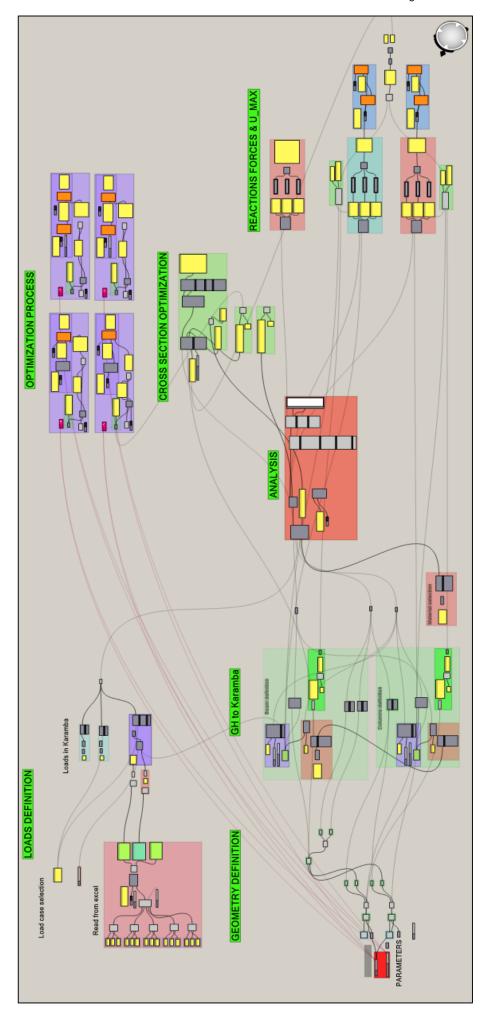
Model A

■ Model B – 1

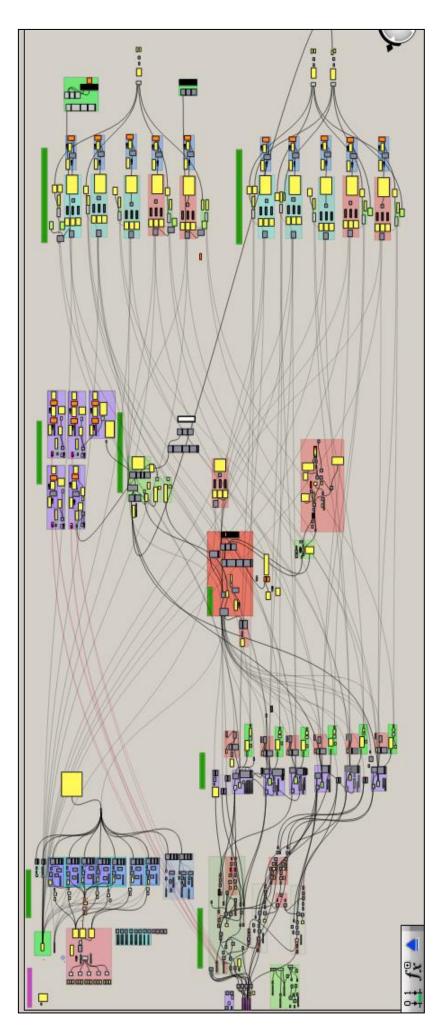
■ Model B − 2

Model B –

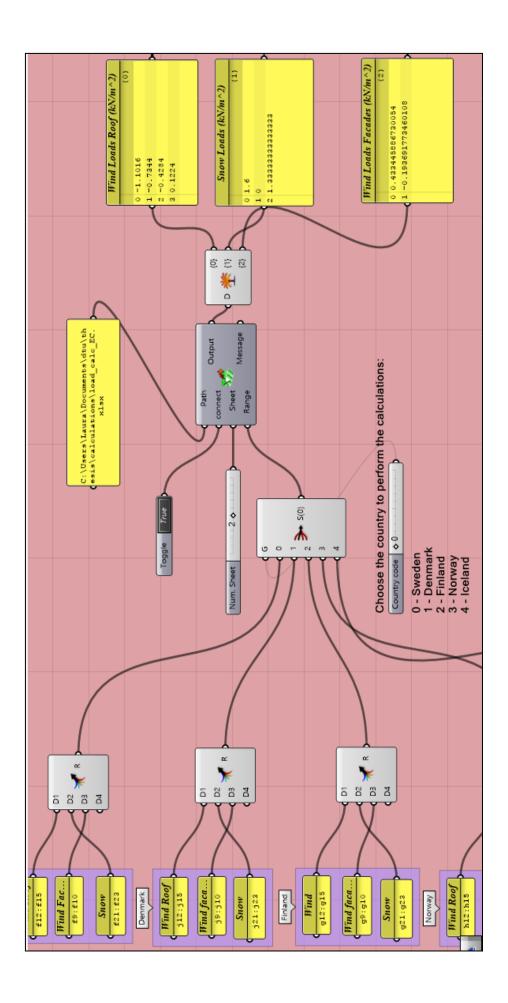
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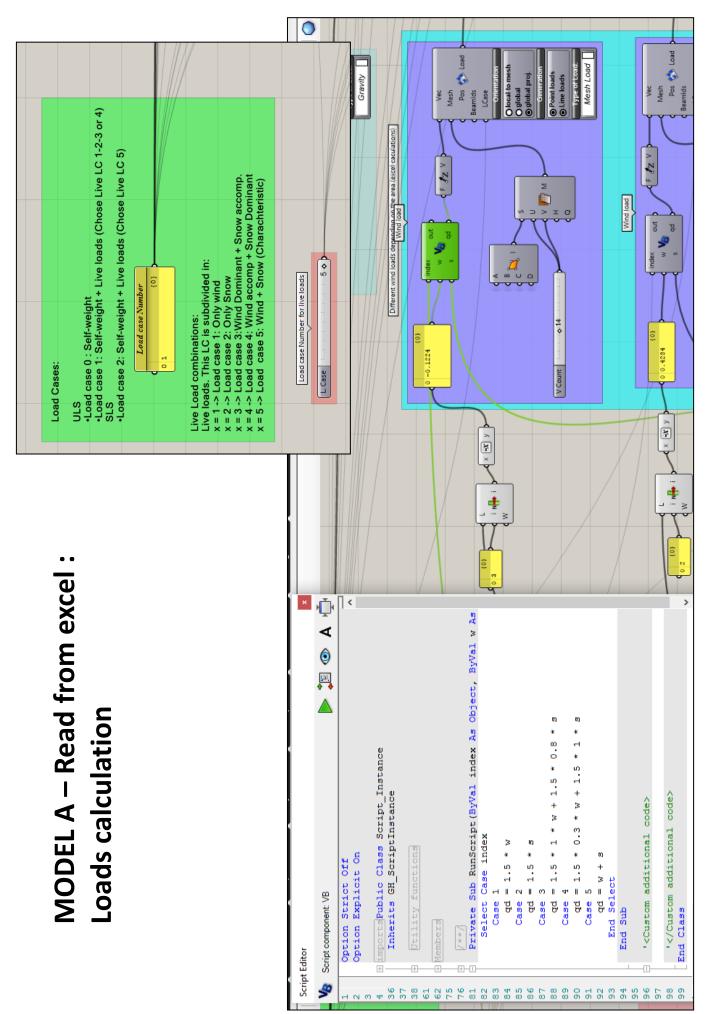


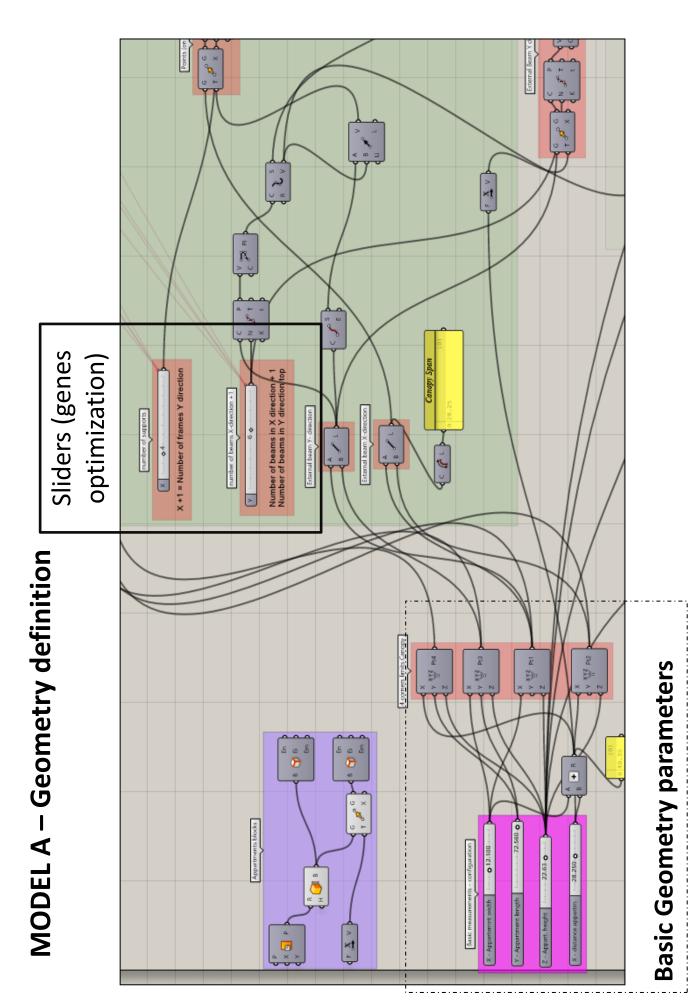
MODEL A – overview of the script



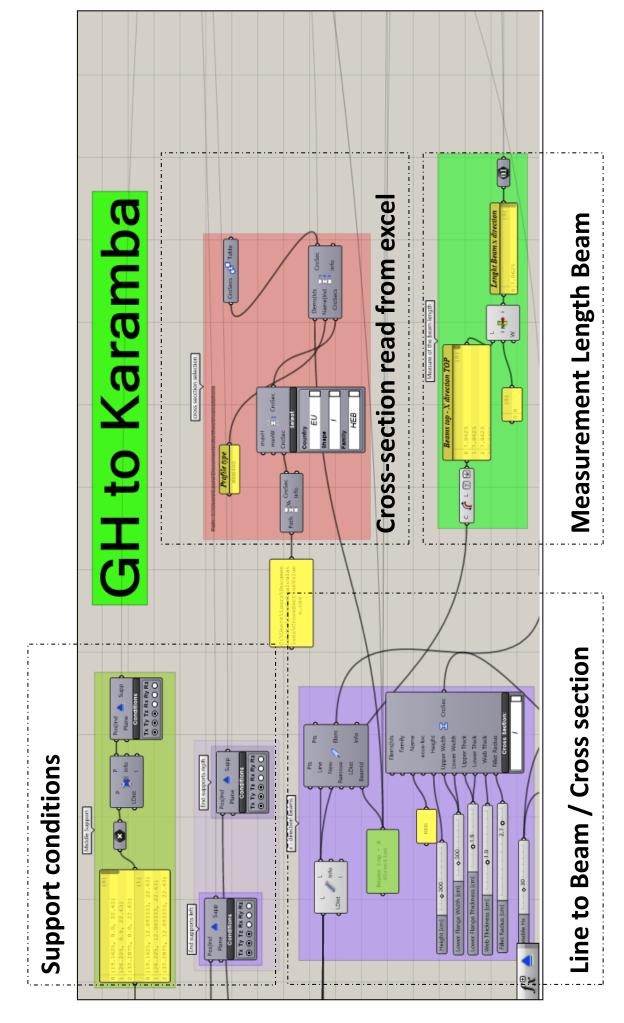
MODEL A – Read from excel: Loads definition



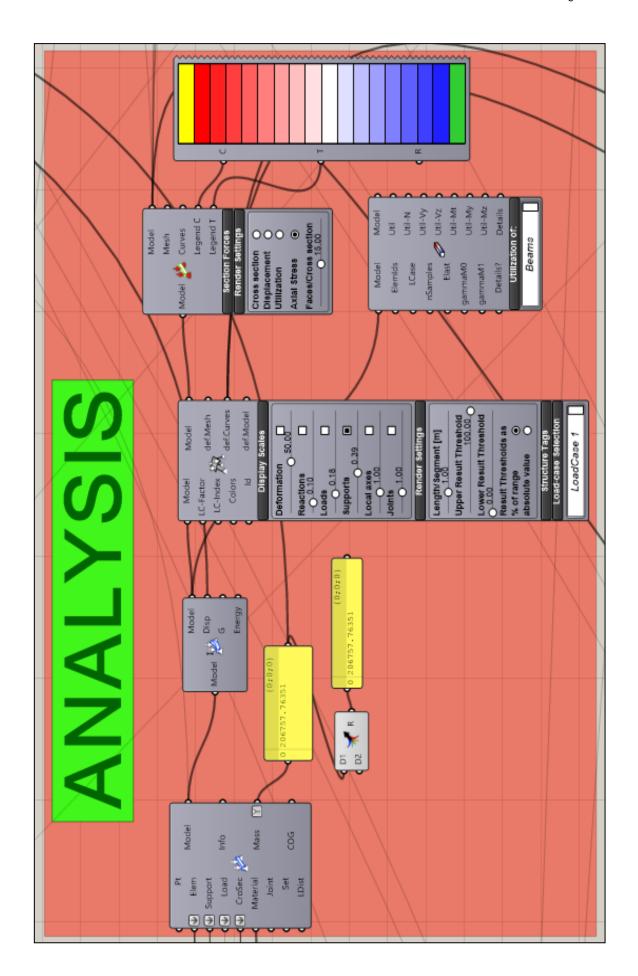




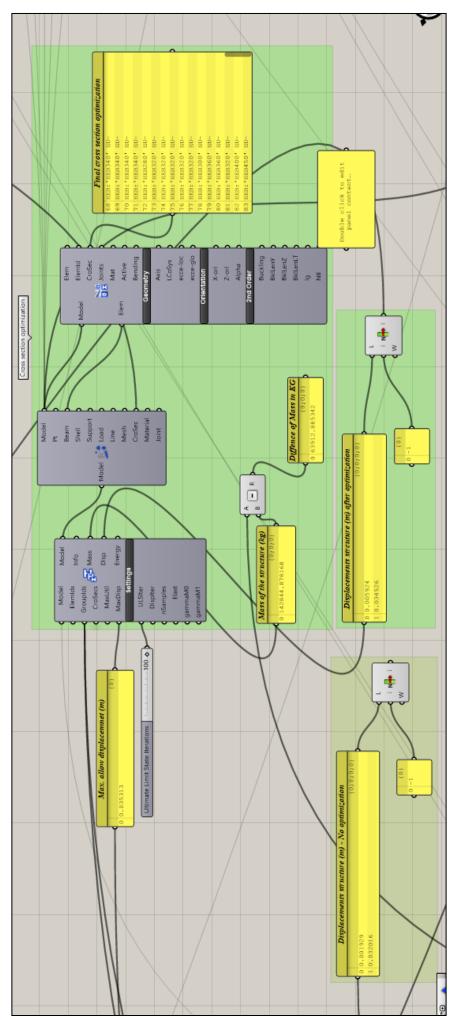
MODEL A – Grasshopper to Karamba

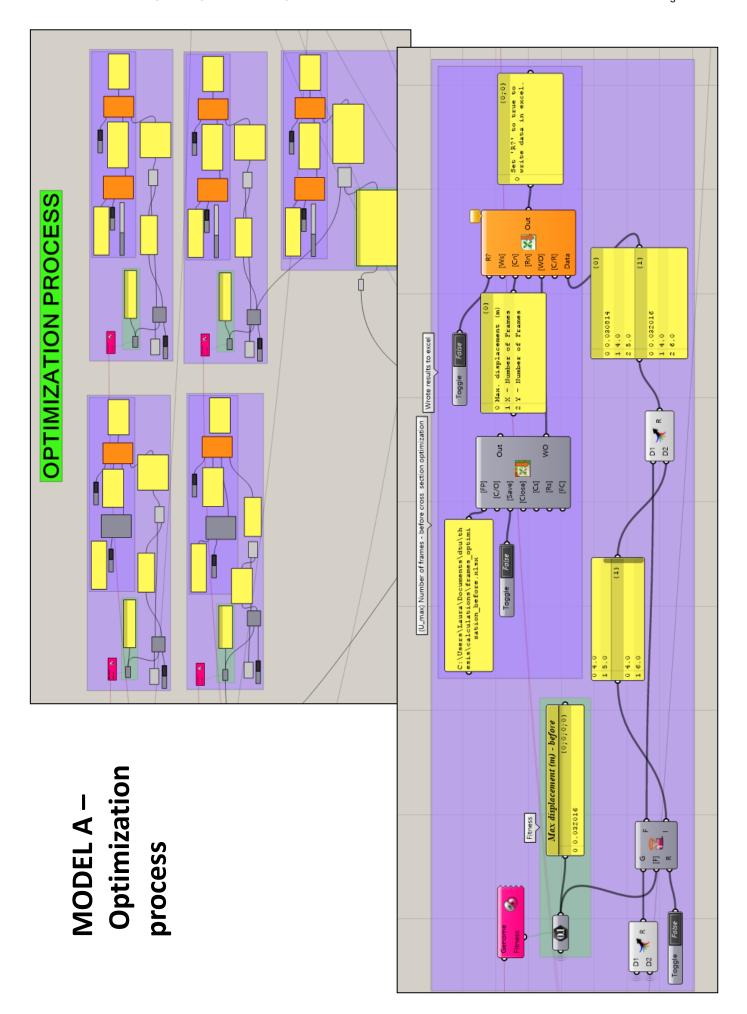


MODEL A – Analysis

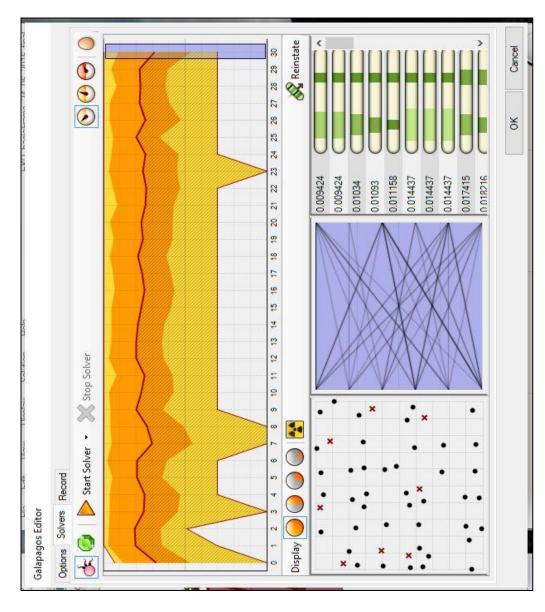


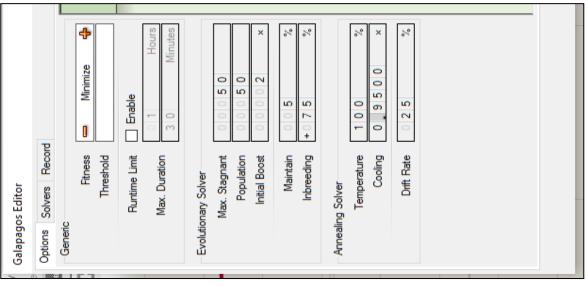
MODEL A – Cross Section optimization



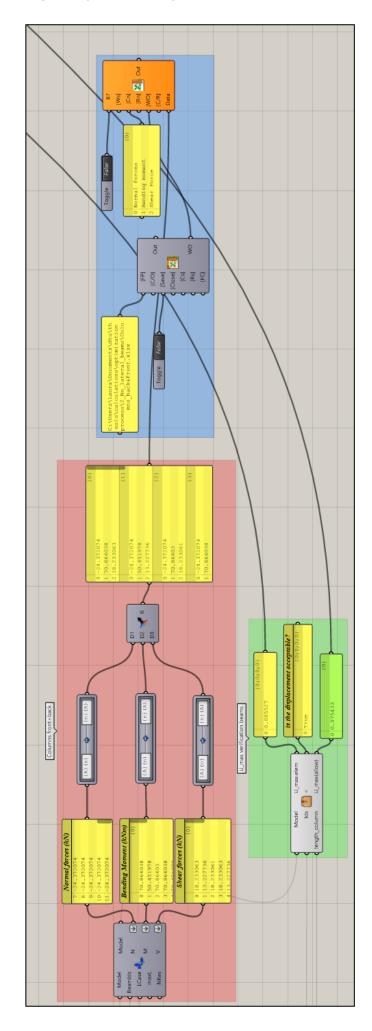


MODEL A – Optimization process input Galapagos

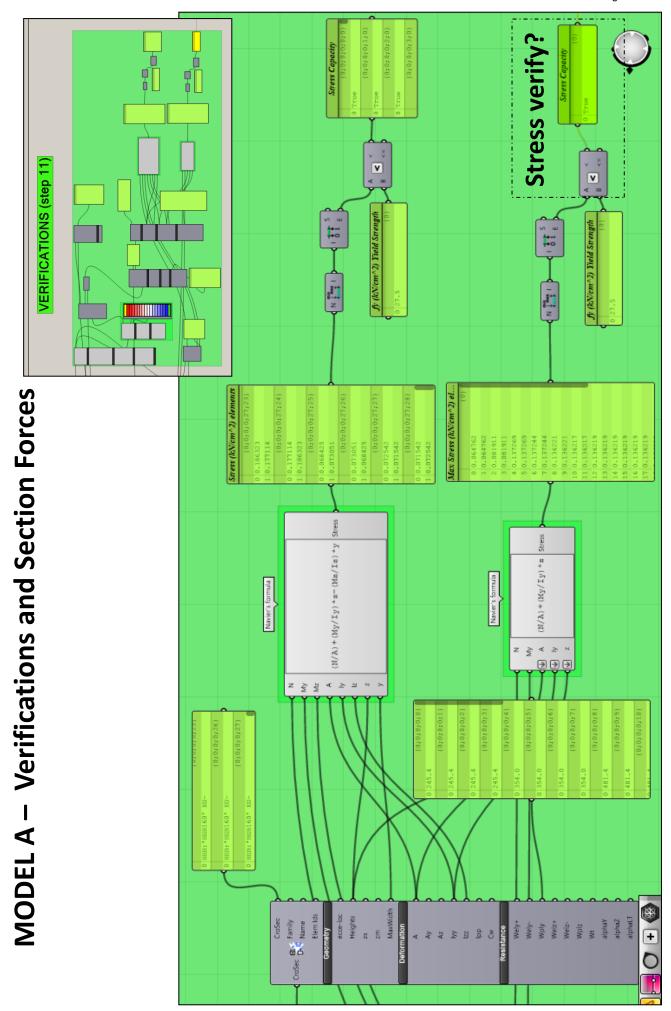


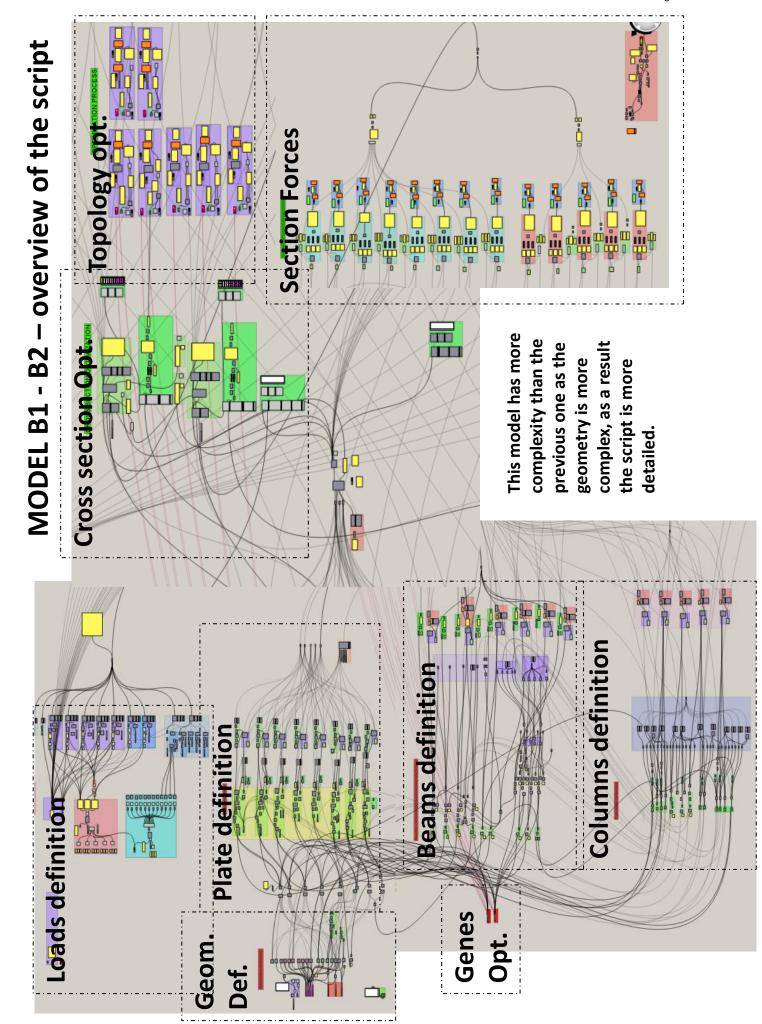


MODEL A – Verifications and Section Forces



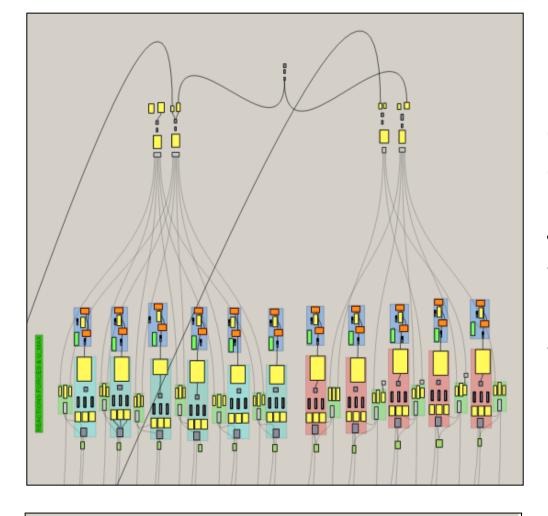
From each group of elements (Beams top X, Beams top Y, Columns Lateral and Columns Front & Back) are taken the maximum section forces: Maximum Bending Moment, Maximum Shear Force and Maximum Axial force, to carry out the corresponding verification according to Eurocode. ightarrow Export to Excel (Step 11)





MODEL B2 – Section forces

The verifications have been divided in the different elements for the optimization step 7 and optimization step 9



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Maximum section forces – Step 9

Maximum section forces – Step 7

_ • _ Analysis Increase of complexity in increase of complexity in perfectly defined for the post structural analysis element should be the script, as every the design means definition Beams support definition MODEL B3 – overview of the script **Geometry evolution Columns + Shells Surfaces** Definition Definitiön Definition Geometry Loads Start